CMC IDLING PROGRAM (P 00)

LOGIC REV 05 11/26/68

PURPOSE:

- (1) TO PROVIDE A PROGRAM TO FULFILL THE FOLLOWING PEQUIREMENTS:
 - (A) PROVIDE AN INDICATION TO THE CREW THAT THE CMC IS ENGAGED IN NO CONTROL OF COMPUTATIONAL OPERATIONS WHICH MIGHT REQUIRE CONSIDERATION FOR COORDINATION WITH OTHER CREW TASKS IN PROGRESS.
 - (B) TO MAINTAIN THE GNCS IN A CONDITION WHERE MANUAL ATTITUDE MANEUVERS CAN BE MADE BY THE CREW WITH MINIMAL CONCERN FOR THE GNCS (SEE ASSUMPTION 3).
 - (C) MAINTAIN THE CMC IN A CONDITION OF READINESS FOR ENTRY INTO OTHER PROGRAMS.

CREW

(2) TO UPDATE THE CSM AND LM STATE VECTORS EVERY FOUR TIME STEPS.

ASSUMPTIONS:

EDIT

- (1) THE IMU MAY OR MAY NOT BE ON. IF ON, THE IMU IS INERTIALLY STABILIZED BUT NOT NECESSARILY ALIGNED TO AN ORIENTATION WHICH IS KNOWN TO THE CMC.
- (2) IF NON-GNCS CONTROLLED ATTITUDE MANEUVERS ARE MADE BY THE CREW CARE MUST BE TAKEN TO AVOID IMU GIMBAL LOCK. THE IMU GIMBAL ANGLES MAY BE MONITORED BY OBSERVING THE ICDUS (V16 N20) OR BY MONITORING THE FDAI BALL.
- (3) DUPING THIS PROGRAM THE CMC ERASABLE STORAGE MAY BE INITIALIZED BY KEYING IN V36E (FRESH START). THIS WOULD BE DONE ONLY AT INITIAL CMC STARTUP OR WHEN THE CONTENT OF THE CMC ERASABLE STORAGE IS IN QUESTION. IF THIS ENTRY IS PERFORMED, THE CMC'S KNOWLEDGE OF THE PRESENT STATE VECTOR AND THE PRESENT INDIORIENTATION (PERSMAT) IS INVALIDATED.
- 14) THE PROGRAM IS MANUALLY SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PCR 507 ++ +04 AND 05 ++

EDIT

PPNG

CMC

(5) THIS PROGRAM IS AUTOMATICALLY SELECTED BY V96E, WHICH MAY BE DONE DURING ANY PROGRAM. STATE VECTOR INTEGRATION IS PERMANENTLY INHIBITED FOLLOWING V96E. NORMAL INTEGRATION FUNCTIONS WILL RESUME AFTER SELECTION OF ANY PROGRAM OR EXTENDED VERB. POO INTEGRATION WILL RESUME WHEN POO IS RESELECTED. USAGE OF V96 CAN CAUSE INCORRECT W-MATRIX AND STATE VECTOR SYNCHRONIZATION.

CONT		1210 C W
. CMC P	enc	. CPEW PROG
• SELFO	TION	• SELECTION
•		•
•		•
• • •		• • •
		•
START CMC TOLING	•	KEY IN CMC IDLING
PROGRAM OO	• • • • • • • • • • • • • • • • • • • •	PROGRAM (POO)
DISPLAY PROGRAM O	n .	M3.7F 00F

GR CUND

#10

TOTAL

TIME

POO/COLOSSUS POO/SUNDANCE POO/LUMINARY

TIME

CHECKLIST

PRO/COLOSSUS POO/SUMPANCE POOLLHMINARY #20 #30 #40 #50 #60 200700108508 POOTSTINDANCE

Ασυμημη Ινυυυ

***** TOP TIE DESHRYE DISPLAY OF PPOCRAM ON ++ IS QUIT FLAG SET? +04 RESET OUTT FLAC +prp +507 AWATT ASTPD-MAILT + 04 ACTION ++ EXTRAPOLATE DEPMAN-ENT STATE VECTORS ECEMBOD TO DEFCENT TIME CHECK 4 TIME +0= STERS. SEE SECTION 5. 519 4.12-1 Eula IS THEVE A PENGEST LUK 4 MEM BOULDAMS

37

FXIT POP AND GO TO PROGRAM SELECTED

VIV BOO

POOZEUNDANCE POOZEUMINARY

#77

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#80

CHANGE CONTROL NOTES

#90

REV 03 PCR MIT 66 REV 04 PCR 507

#100

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	PREL	AUNCH OR SER	RVICE - INITIALIZATIO	N PROGRAM (POI)	LOGIC	REV 04	07/08/68		
PURPOSE:	(1)	TO INITIALI	IZE THE PLATFORM FOR	THE PRELAUNCH PROGRAM	IS.				
	{2}	TO PROVIDE	AN INITIAL STABLE ME	MBER ORIENTATION FOR	GYRD COMP	ASSING (PO	21.		
ASSUMPTIONS:	{1}	THE PROGRAM	IS MANUALLY SELECTE	D BY DSKY ENTRY.					
	(2)	ERASABLE LO PARAMETERS)	DCATIONS HAVE BEEN PR	OPERLY INITIALIZED.(A	ZIMUTH, +:	l; EATITUD	E, +1; LAUNCHAZ,	, +1; IMU COMPENSAT	ION
PROG Cont		смс	GRCUND	CREW			CHECKLIST	TIME	TOTAL TIME
PRO	GRAM		•	CREW PROGRAM SELECTION KEY IN INITIALIZATION PROGRAM (PO1) V37E01E	- -				# 10
	IMAND	ISS ZERO	•	MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM O1	-				#20
	ROUT	•							#30
		•						PO 17 COLOSSUS	

TIRN ON "NO ATT" COMMAND COARSE ALIGN IN ISS. COARSE ALIGN TO DESIRED PLATFORM RIFHTATION TURN OFF "NO ATT" LIGHT OFF TORNORF "NO ATT" URSERVE "NO ATT" LIGHT OFF PEMOVE COARSE ALIGN TO DESIRED PLATFORM TORNORF "NO ATT" LIGHT OFF TORNORF "NO ATT" LIGHT ON ATT" LIGHT OFF TORNORF "NO ATT" LIGHT	* *			PO 1/COLOSSUS
URN ON "NO ATT" IGHT	•			
CURN ON "NO ATT" LIGHT ON LIGHT OFF	SECONDS			
IGHT ON "NO ATT" LIGHT ON TO SERVE "NO ATT" LIGHT ON TO SERVE "NO ATT" LIGHT ON TO SERVE "NO ATT" LIGHT OF TO SERVE TERMINATION OF TO SERVE TERMINATION OF POSERVE TERMINATION OF	•			
OMMAND CRARSE ALIGN IN ISS. CARSE ALIGN TO DESIRED PLATFORM IRIENTATION CHANNO RELEASE VALAFORM). CURN OFF "NO ATT" LIGHT LIGHT CHANNO RELEASE LIGHT OFF CHANNO RELEASE CHANNO RELE	. IGHT	•	OBSERVE "NO ATT" LIGHT ON	
N ISS. DARSE ALIGN TO ESIRED PLATFORM RIENTATION EMOVE COARSE ALIGN CMMAND (RELFASE LATFORM). URN OFF "NO ATT" IGHT LIGHT OFF FRMINATE PROGRAM 01 ND GC TO PRELAUNCH R SERVICE — GYRO OF PO1 AND OISPLAY OF PO2 OF PO2 OF PO2 OF PO2 OF PO2	•			
EMOVE COARSE ALIGN CMMAND (RELFASE LATFORM). URN DEF "NO ATT" IGHT UBSERVE "NO ATT" LIGHT OFF FRMINATE PROGRAM 01 ND GC TO PRELAUNCH MONITOR DSKY: MACHINATION MEDITOR DSKY: MED	N ISS. DARSE ALIGN TO ESIRED PLATFORM			
EMOVE COARSE ALIGN CMMAND (RELFASE LATFORM). URN OFF "NO ATT" IGHT URN OFF "NO ATT" LIGHT OFF FRMINATE PROGRAM 01 ND OF POLAUNCH ND GC TO PRELAUNCH ND SERVE TERMINATION OF PO1 AND DISPLAY CMPASSING PPOGRAM OF PO2 CMPASSING PPOGRAM OF PO2 CMPASSING PPOGRAM OF PO2 CMASSING PPOGRAM OF PO3	:			
TURN OFF "NO ATT" . IGHT . LIGHT OFF . LI	REMOVE COARSE ALIGN COMMAND (RELFASE	•		
LIGHT OFF PRINATE PROGRAM 01 MONITOR DSKY: IND GC TO PRELAUNCH	•			
FRMINATE PROGRAM 01 . MONITOR DSKY: IND GC TO PRELAUNCH	. IGHT	•		
FRMINATE PROGRAM 01 . MONITOR DSKY: ND GC TO PRELAUNCH DASERVE TERMINATION R SERVICE - GYRO . OF PO1 AND DISPLAY CMPASSING PPOGRAM . OF PO2	•			
· · · · · · · · · · · · · · · · · · ·	AND GC TO PRELAUNCH DR SERVICE - GYRO CCMPASSING PROGRAM		MONITOR DSKY: ORSERVE TERMINATION OF PO1 AND DISPLAY	
•••	· · ·		•	
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CHANGE CONTROL NOTES

REV 04 PCR 206

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PRELAUNCH OR SERVICE - GYRO COMPASSING PROGRAM (PO2) LCGIC REV 04 11/27/68 PURPOSE: (1) TO PROVIDE THE PROPER STABLE MEMBER ORIENTATION FOR LAUNCH. ASSUMPTIONS: (1) THIS PROGRAM MAY BE INTERRUPTED TO PERFORM THE PRELAUNCH OR SERVICE - OPTICAL VERIFICATION OF GYROCOMPASSING PROGRAM (PO3). (2) V75 WILL BE KEYED IN AND DISPLAYED DURING THIS PROGRAM TO PERMIT CREW BACKUP OF THE LIFTOFF DISCRETE. (3) THE PROGRAM IS AUTOMATICALLY SELECTED BY THE INITIALIZATION PROGRAM (POI). (4) THIS PROGRAM HAS THE CAPABILITY (VIA V78E) TO CHANGE LAUNCH AZIMUTH OF THE STABLE MEMBER WHILE GYROCOMPASSING. PROG CMC GROUND CREW CHECKLIST TIME TOTAL CONT TIME . AGC . PROG SELECTION START PRELAUNCH OR MONITOR DSKY: SERVICE-GYRO OBSERVE DISPLAY OF #10 COMPASSING PROGRAM PRDGRAM 02 { PO 2 } DISPLAY PROGRAM 02 SET TIMER #20

PO2/COLOSSUS

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•	•	•					PO2/COLOSSUS	
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•	•	•	•	•				
HOLD	ELACU MEDO NOUN TO	•	•	•				
SNAP	FLASH VERB-NOUN TO	•	•	•				
SMAP	REQUEST PROCEED AND DISPLAY STORED	••	-KEV	 IN .				#80
•	LAUNCH AZIMUTH		V7		••••			#00
•	V06 N29	•						
•	RI-XSM LAUNCH AZIMUTI	н .		•				
•	R 2-BL ANK	•	•		•			
•	R3-BLANK	•	•	•	•	. •		
•	UCM I DIMEN ATTMITI	•••••			•	. •		
•	XSM LAUNCH AZIMUTH	•		ITOR DSKY:	•	. •		
•	MEASURED CLOCKWISE FROM TRUE NORTH IN	•		ERVE VERB-NOUN SH TO REQUEST	•	. •		
•	DEGREES TO NEAREST	•		CEED AND DISPLA	v .	•		#90
	.01 DEGREES	•	OF	XSM LAUNCH	•			#70
•		•		MUTH	•			
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•	WAIT FOR KEYBOARD Entry	•••••	•	KEY IN V21E AND		. •		
•	ENTRY	• •	•	LOAD NEW LAUNCE	п.	•		
•	·	<u>.</u>	•	ALIMOID	:	•		
•		•						#110
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•	TERMINATE FLASH	•				•		
•	UPON RECEIPT OF PRO- CEED OR NEW DATA	••	KEY	IN PROCEED		•		
•	CEED OR NEW DATA					•		
	.P .NEW	• •				•		
•	.R .DATA	•				•		
•	•0	•				•		#120
-	•¢ •	•				•		
•	•E	•				•		
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•	•					•	P02/COLOSSUS	
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PC2/CCLOSSUS

LO FROM .

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SIVB .

..... START MCNITCR OF

LIFT OFF DISCRETE

PROGRAM (P11)

EXIT PO2

• LO . OR

P02/COLOSSUS

EXIT PO2

CHANGE CONTROL NOTES

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PRELAUNCH OR SERVICE - CPTICAL VERIFICATION OF GYRC COMPASSING (PO3)

LOGIC REV 08 07/12/68

PURPOSE:	(1)	TO PROVIDE AN OPTICAL	CHECK FOR	VERIFICATION :	OF ALIGNMENT (F THE	STABLE	MEMBER	OF T	HE 155	DURING	GYRO	COMPASSING
		PRIOR TO LAUNCH.											

- ASSUMPTIONS: (1) THE PROGRAM IS MANUALLY SELECTED BY DSKY ENTRY.
 - (2) THE ASTRONAUT HAS ZEROED THE OPTICS JUST PRIOR TO PROGRAM (PO3) SELECTION.
 - (3) A MINIMUM OF 45 MINUTES BETWEEN V78E AND PO3 (V65E) INSURES PROPER DAMPING OF TRANSIENTS.

++
+08 (4) IN ORDER TO PREMATURELY TERMINATE THIS PROGRAM AND RETURN TO PG2 THE ASTRONAUT MAY KEY IN V34E ON ANY FLASHING
+08 DISPLAY
++

PROG CMC GROUND CREW CHECKLIST TIME TOTAL CONT

•CREW
•PROGRAM
•SELECTION

IS PROGRAM PO2

OPERATING?

SERVICE — OPTICAL

DEPLETATION OF

PERATING? . SERVICE — OPTICAL

VERIFICATION CF

VY .N GYRO COMPASSING

PROGRAM (P03)

V65E

SET OPERATOR
ERROR LIGHT ON

BERROR LIGHT ON

#20

#10

++

+ 08 +

+08

50

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31	•					
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	•					PO3/COLOSSUS
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HOLD .	FLASH VERB-NOUN TO	•	MONITOR DSKY:		•	
• • • • • • • •	REQUEST PROCEED AND		-OBSERVE VERB-NOL	JN .	•	#90
SNAP .	DISPLAY STORED	•	FLASH TO REQUES!	r,	•	
	TARGET 1 AZIMUTH AND	•	PROCEED AND DISP	LAY .	•	
	ELEVATION:	•	OF TARGET 1 AZI		•	
	V06 N41	_	AND ELEVATION		•	
	R1-TARG AZ	•			•	
	R2-TARG ELEV	•			•	
		•	•	•	•	
	R3-TARG ICENT	•	•	•	•	
		•	•		•	
	TARG AZ-TARGET	•			•	
	AZIMUTH-ANGLE CLOCK-	•	AM I SATISFIED)	iith .	•	#100
	WISE FROM TRUE NORTH	•	THE AZIMUTH AND		•	
	TO THE TARGET. IN	_	ELEVATION OF TAR	GET	•	
	DEGREES TO NEAREST	_	1?		_	
	.01 DEGREE	•		-		
	*OI DEGREE	•	¥	AI.	•	
	T.00 5.54 T.005T	•	• Y	•N .		
	TARG ELEV-TARGET	•	•	•	•	
	ELEVATION-ANGLE FROM		•	•	•	
	THE LOCAL HORIZONTAL	•	•		•	
	(OF NAV BASE) TO THE	•	•		•	
	TARGET. IN DEGREES	•	•		•	#110
	TO NEAREST .001	•	•		•	
	DEGREE	-	_			
			_		-	
	TARG IDENT-TARGET	•	•	•		
	IDENTIFIER-IDENTIFI-	•	•	•		
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	ES AZIMUTH AND	•	•	•		
	ELEVATION FOR TARGET	•	•	• •		
	1 OR 2	•	•	•		
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	•	•	•		•	#120
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	WAIT FOR KEYBOARD		KEY IN PROCEED		•	
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		•		•	#130
	TERMINATE FLASH UPON		KEY IN V21E, V22E, OP	•	2 0
	RECEIPT OF PROCEED		V24E AND LOAD NEW	•	
	OR NEW DATA	***************************************	DATA	•	
	OK NEW DATA	• •	UATA	•	
		•		•	
	.P .NEW	•	•	•	
	.R .DATA	•	•	•	
	•0	•	•	•	
	• C •	•	*****	•••	
	•E	_			
	.E DISPLAY AND	-			#140
	D STOPE NEW	•			#140
		•			
	• DATA	•			
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	SET TARG IDENT TO				#150
	00002 IN R3 OF NOUN				
	30				
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++					
+ 08	PASTE VOSNOG AND				
+	THEN VO6N41 (DO NOT				
+08	OVERWRITE R3)				#160
++					*100
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HO! D	• • • • • • • • • • • • • • • • • • •	··· · ·	MONITOR DOVA	···· · · ·	#170
HOLD .	FLASH VERB-NOUN TO	·· · · · ·	MONITOR DSKY:	···· · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND	·· · · · · · · · · · · · · · · · · · ·	OBSERVE VERB-NOUN	··· · · · · · · · ·	#170
	REQUEST PROCEED AND DISPLAY STORED	••	OBSERVE VERB-NOUN FLASH TO REQUEST	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND		OBSERVE VERB-NOUN FLASH TO REQUEST PROCFED AND DISPLAY	··· · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND		OBSERVE VERB-NOUN FLASH TO REQUEST PROCFED AND DISPLAY	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION:		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	• • • • • • • • • • • • • • • • • • •	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41		OBSERVE VERB-NOUN FLASH TO REQUEST PROCFED AND DISPLAY	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	• • • • • • • • • • • • • • • • • • •	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	· · · · · · · · · · · · · · · · · · ·	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH		
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	PC 3/6	#170
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	PC 3/C	
SNAP.	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	PC 3/C	
SNAP.	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	PC 3/C	
• • • • • • • • •	REQUEST PROCEED AND DISPLAY STORED TARGET 2 AZIMUTH AND ELEVATION: VM6 N41 R1-TARG AZ R2-TARG ELEV		OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF TARGET 2 AZIMUTH	PC 3/C	

P03/COLOSSUS

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	•	•		•	PO3/COLOSSUS
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TARG AZ-TARGET	•			•	
AZIMUTH-ANGLE CLOCK-		AM I SATISFIED	WITH	_	
WISE FROM TRUE NORTH		THE AZIMUTH AND		•	
	•			•	
TO THE TARGET. IN	•	ELEVATION OF TA	KUET	•	
DEGREES TO NEAREST	•	2?		•	
.01 DEGREE	•			•	
	•	• Y	• N	•	
TARG ELEV-TARGET	•	•	•	•	
ELEVATION-ANGLE FROM	•	•	•	•	
THE LOCAL HERIZONTAL	•	•		•	
(OF NAV BASE) TO THE		-	•	_	
TARGET. IN DEGREES		_	-	•	
TO NEAREST .001	*	•	•	•	
	•	•	•	•	
DEGREE	•	•	•	•	
	•	•	•	•	
TARG IDENT-TAPGET	•	•	•	•	
IDENTIFIER-IDENTIFI-	•	•	•	•	
ES AZIMUTH AND	•	•	•	•	
ELEVATION FOR TARGET	•	•	_	•	
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WAIT FOR KEYBOARD		KEY IN PROCEED	•	•	
ENTRY				•	
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TERMINATE ELACU HRON	•	V5V IN H215 H22		•	
TERMINATE FLASH UPON		KEY IN V21E-V22		•	
RECEIPT OF PROCEED	*****************	OR V24E AND LOA	D NEW	•	
OR NEW DATA	• •	DATA		•	
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•	•		•	•	•	P@3/COLOSSUS	
•	•		•			FC37C0C03303	
•	•		•		•		
•	•		•		•		
REQU FORM SEQU	SH VERB-NOUN JEST PLEASE M TERMINATE UENCE 50 N25 1-00016	PER	MCNITOR DSKY: OBSERVE FLASH VERB-NOUN TO PLEASE PERFOR INATE MARK SE	ING REQUEST M TERM— QUENCE			
			.YES	•NO .	•		#290
•	•		•	•	•		
•	•		•	•	•		
:	•		•	•			
•	•		•	•	•		
. WAIT	T FOR KEYBOA RY	RD	. REJECT BUT		• • •		
•			•		•		#300
•			•				#J00
•			•	•	•		
•			•	• • • • • •	•		
•			•				
• TCOA	MINATE FLASH	LUDON	•				
. RECE	EIPT OF CEED OR REJE	ст .	KEY IN PROCEE	0			
	REJECT .	PROCEEC					#310
	•						
• •	•						
• •	•						
EDAG	SE LAST						·
	OF MARK .				,	٠	
DATA							
• •	•						
• •	•						#320
• •	•						
	•						
	•						
	COMPUTE LOS TARGET NUMB THO AND ORT OPTICS IF ALLOWED.	EP					#330
	•						
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	•				•		
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PC3/COLOSSUS

•			•	• PO3/COL OSSUS
•	•		•	•
	•		•	•
HOLD .	FLASH VERB TO REQUEST PLEASE MARK V51N BLANK R1-BLANK		MONITOR DSKY: OBSERVE VERB FLASH TO REQUEST PLEASE MARK	* #340
+67	R2-BLANK R3-BLANK		PEEASC MARK	
+07 .			•	•
++	•		•	•
•	•			•
•	:		SELECT CMC OPTICS MODE. OBSERVE SXT DRIVE (OPTIONAL)	• #350
•	•			•
•	•		•	•
	•		•	•
•	•		SELECT MANUAL OPTICS MODE	• •
•	•			*
•	•		•	• #360 •
•	•		•	•
•	WAIT FOR MAPK TERMINATE FLASH UPON RECEIPT OF MARK	. MARK	WHEN SIGHTING ON TARGET 2 IS SATIS— FACTORY PRESS MARK BUTTON	• • •
	•			•
•	•		•	*
•	•		•	• #370 •
•	CTORE MARK CATA		•	•
•	STORE MARK CATA		•	•
•	•		•	•
•	•		•	•
•	FLASH VERB-NCUN TC		MONITOR DSKY:	•
	REQUEST PLEASE PERFORM TERMINATE MARK SEQUENCE	•		• #380 •
•	V50 N25 R1-00016		TERMINATE MARK SEQUENCE	• •
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•	•		:	• PU3/COLOSSUS
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	•					•	•	PO3/COLOSSUS	
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	•			<u> </u>	AS SIGH				
	•			, m	ATISFAC	TODYS	•		#390
	•				PITSPAC	IUNTI	•		
				_,	YES	• NO	•		
	•			•	,	• 140	•		
	•			-			•		
			 REJECT 	•			:		
WAIT FOR K	E YBOA	LRD	************		. PRESS	MARK			
ENTRY			•	•	REJEC	T BUTTON			
				•					
				•		•	•		#400
				•		•	•		
				•		•	•		
				•		• •	• • • • •		
				•					
TERMINATE	ELACL	LUDCN		•					
RECEIPT OF	PROC	FER	•		EY IN P	POCEEU			
DR REJECT	, MUC	LLU		N		KUCFEU			
			•			_			
.REJECT		PROCEI	ED						#410
•	•								*410
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•	•								
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	•								
ERASE LAST	•								
SET OF MAR	к .								
DATA	•								
•	•								#420
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	-								
CALCULATE :	SM MI	S-				1			
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	•	•			•			PO3/COLOSSUS	
		•			•			L03/C0E1/2202	

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	FLASH VERB-NOUN TO REQUEST PROCEED AND DISPLAY DELTA GYPO ANGLE: VO6 N93			MONITOR DSKY OBSERVE FLAS VERB-NOUN AN DISPLAY MISA ANGLES	HING ID	1ENT	#4	40
	RI-DELTA GYRO ANGLE X R2-DELTA GYRO ANGLE Y R3-DELTA GYRO ANGLE Z		-	•			#4	50
	DFLTA GYRO ANGLE- ALIGNMENT ERROR IN DEGREES TO NEAREST .001 DEGREE	· · · · · · · · · · · · · · · · · · ·		•				
	WAIT FOR KEYBOARD			CONFER WITH			#4	60
	ENTRY		1	ARE GYRO MIS MENT ANGLES ABLE? 				
			:	SHALL I CONT UE WITH BAD DATA?	 IN-	•	#4	70
				.NO : : KEY IN TER-	•YES	•		
				V34E	•	•	#4	80
				EXIT	•	•		
					•	•		
53)				:	•	P03/C0L0SSUS	

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		'	•		•	PO3/COLOSSUS	
		'	•	•	•	503/CUE02202	
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			•				#490
			• • •	HAVE	- I		
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		1		THE			
		•		AND			
			•	GYRC)		
				ERRO	OR?		
				.N	. Y		
				•			
		,		•			#500
							*200
			KEY IN V24E A	IND	-		
		,	LOAD RI AND R	2	-		
		'	. WITH ZEROES.		•		
		•	• WITH CERUES•		. •		
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		•	• •		•		#510
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TERMI	NATE	FLASH UPON	****				
RECEI	PT O	F TERMINATE	KEY IN PROCEE	n			
OR PR	OC FEI						
DATA	5 ¢		•				
							#620
. P	· .T	.NEW					#520
R	. E						
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• D	• A	• •	•				
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•	•	•	•				#530
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•	•	STORE NEW .	•				
		DATA	•				
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•	•						#540
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•	•						
_						PO 3/COLOSSUS	
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PO3/COLOSSUS

#550

#560

. TERMINATE PO3 . AND REDISPLAY . PO2. CONTINUE . GYRO COMPAS-· SING. EXIT PO3 TORQUE Z GYRO PER ERROR TERMINATE PO3 AND REDISPLAY PO2. CONTINUE GYRO COMPASSING.

#570

EXIT PO3

LOGIC REV 06 PCR MIT 66 LOGIC REV 07 PCR 435 LOGIC REV 08 PCR 206 PCR 493

CHANGE CONTROL NOTES

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CMC POWER DOWN PROGRAM (PO6)

LCGIC REV 08 07/08/68

PURPOSE:

(1) TO TRANSFER THE CMC FROM THE OPERATE TO THE STANDBY CONDITION.

- ASSUMPTIONS: (1) WHEN THIS PROGRAM IS TURNED ON THE ASTRONAUT MUST POWER DOWN THE CMC TO STANDBY.
 - (2) THE NORMAL CONCITION OF REACINESS OF THE GNCS WEEN NOT IN USE IS STANDBY. ALL THE G/N CKT BKRS (PANEL 5) ARE CLOSED. THE IMU AND OPTICS GAN POWER SWITCHES (LEB PANEL 100) ARE OFF AND THE CMC STANDBY LIGHT (DSKY) IS ON. IN THIS CONDITION THE IMU IS IN STANDBY WITH ONLY HEATER POWER ON, OPTICS POWER IS OFF AND THE CMC IS IN STANDBY.
 - (3) A POSSIBLE CONDITION OF REACINESS OF THE GNCS WHEN NOT COMPLETELY ON IS THE SAME AS STANDBY (2) ABOVE. EXCEPT THE CMC STANDBY LIGHT ON THE MAIN AND LEB DSKYS IS OFF. IN THIS CONFIGURATION THE CMC IS RUNNING FOR COMPUTATIONAL PURPOSES THAT DO NOT REQUIRE THE IMU OR OPTICS.
 - (4) IF THE COMPUTER POWER IS SWITCHED OFF IT WILL BE NECESSARY TO PERFORM A COMPUTER FRESH START (V36)E TO INITIALIZE THE ERASABLE STORAGE. THE CMC UPDATE PROGRAM (P27) WOULD HAVE TO BE DONE TO UPDATE THE STATE VECTOR AND COMPUTER CLOCK TIME.
- (5) THE CMC IS CAPABLE OF MAINTAINING AN ACCURATE VALUE OF GROUND ELAPSEC TIME (GET) FOR ONLY 23 HRS WHEN IN THE

STANDBY MODE. IF THE CMC IS NOT BROUGHT OUT OF THE STANDBY CONDITION TO THE RUNNING CONDITION (SEE (3) ABOVE) AT LEAST ONCE WITHIN 23 HOURS THE CMC VALUE OF GET MUST BE UPDATED.

(6) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG Cont	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIM
	START CMC POWER DOWN PROGRAM (PO6) DISPLAY PROGRAM D6	•	CREW PROG SELECTION KEY IN CMC POWER DOWN PROGRAM (PO6) V37E O6E			#10
	**		MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM 06			

#20

PO6/CDL OSSUS PO6/SUNDANCE PO6/LUMINARY

	•		•		OLOSSUS
	•		•		JNDÅNCE JMINARY
	•		•		
	•		•		
	•		:		
	RESET REFSMMAT FLAG		•		
			•		
	•		•		
	•		•		
	RESET RENCEZVOUS		•		#30
	FLAG		:		
			•		
	•		:		
++	•		•		
+08	•		•		
++	•		:		
	STORE CMC CLOCK TIME.		•		
			•		#40
	•		•		
	•		•		
		••••	•		•
		•	•		
		•			
HOLD.	FLASH VERB-NOUN TO REQUEST PLEASE PER-	•	MONITOR DSKY: OBSERVE VERB-NOUN		
SNA P.	FORM CMC POWER DOWN:	•	FLASH TO REQUEST		#50
	V50 N25	•	PLEASE PERFORM CMC		
	R1-00062 R2-BLANK	•	POWER DOWN		•
	R3-BLANK	•	•		
		•	•		
	:	•	:		
	•	•	•		
	•	•	•		#40
	WAIT FOR KEYBOARD	:	:		#60
	ENTRY	•	•		
		•	•		
		•	:		
		•	•		
		••	PRESS PROCEED		
		••	BUTTON AND		
		•	HOLD DEPRESS-		#70
		•	ED UNTIL STANDBY LIGHT		
		•	COMES ON.		
		•			
		•	•	DUTAL	OL OSSUS
		•	•	P06/SI	UNDANCE
		•	•	P06/L1	UMINARY
62	`	•	•		•
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	e				-

FXIT PO6

PO6/COLOSSUS PU6/SUNDANCE PO6/LUMINARY

#80

#90

CHANGE CONTROL NOTES

TERMINATE FLASH UPON RECEIPT OF PROCEED

• ENTER

• V34E • OR • V32E

PROCEED

STANDBY

EXIT

LOGIC REV 07 PCR MIT 66 LOGIC REV 08 PCR NASA 206



EARTH ORBIT INSERTICA MONITOR PROGRAM (P11)

LCGIC REV 06 07/08/68

P11/COLOSSUS

PURPOSE:

- (1) TO INDICATE TO THE ASTRONAUT THAT THE CMC HAS RECEIVED THE LIFTOFF DISCRETE.
- (2) TO GENERATE AN ATTITUDE ERROR INDICATION ON THE FDAI ERROR NEEDLES, SCALED FOR THE 50/15 SETTING; FROM LIFTOFF TO THE BEGINNING OF PITCHOVER/ROLLOUT THE ATTITUDE ERROR IS EQUAL TO THE DIFFERENCE BETWEEN THE CURRENT VEHICLE ATTITUDE AND THE ATTITUDE STORED AT LIFTOFF. DURING PITCHOVER/ROLLOUT THE ATTITUDE ERROR IS EQUAL TO THE DIFFERENCE BETWEEN THE CURRENT VEHICLE ATTITUDE AND THE CMC NOMINAL COMPUTATION OF VEHICLE ATTITUDE BASED ON THE STORED POLYNOMIALS IN PITCH AND ROLL.
- (3) TO DISPLAY CHO COMPUTED TRAJECTORY PARAMETERS.
- ASSUMPTIONS: (1) THE PROGRAM IS NORMALLY AUTOMATICALLY SELECTED BY THE GYRO COMPASSING PROGRAM (PO2) WHEN THE CMC RECEIVES THE LIFTOFF DISCRETE FROM THE SIVB. IN THE BACKUP CASE IT WOULD HAVE BEEN SELECTED BY KEYING IN V75 ENTER AS NOTED EARLIER IN PO2.
 - (2) THE ORBIT PARAMETER DISPLAY ROUTINE IS AVAILABLE BY KEYING IN V82E.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME TOTAL Time
	• CMC PRCG • SELECTIO •				
	START ECI MCNITOR PROGRAM (P11) DISPLAY PROGRAM 11	•	MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM 11		#10
	•		•		*IV
	SEND EFFECTIVE TIME OF LIFTCFF ON DCWNLINK		VERIFY AUTOMATIC START OF DIGITAL EVENT TIMER		
	•				#20
++ +06	ZERO CMC CLOCK				
++	•				

64

HOL D

MON

#80

#90

#100

#110

#120

CALL ROUTINE TO MONITOP: LCAD TODU DACS WITH (A) FDAI ATTITUDE PITCH, ROLL, AND YAW **ERROR NEECLES AS** ATTITUDE ERRORS DE-INDICATION OF RIVED FROM PRESENT CMC COMPUTATIONS ATTITUDE AND STORED OF INSERTION. LIFTOFF ATTITUDE UNTIL PRESENT TIME BALL INDICATES EQUALS TEL (STORED INITIAL VEHICLE IN ERASABLE MEMORY) RCLLOUT AND THEN AT WHICH TIME THE GRADUAL PITCH-STORED LIFTOFF ATT-OVER. ITUDE IS REPLACED BY THE SOLUTION TO THE (B) DSKY: STORED 6TH CRDER RI-VI INCREASING BOOST POLYNEMIAL. R2-H DOT FOLLOWS NOMINAL HISTORY AT TIME TEL + TE2 R3-H INCREASING (TE2 IS STORED IN ERASABLE MEMORY) SHUT OFF BOOST POLY-NOMIAL AND HOLD ATTITUDE ERROR NEED-LES CONSTANT AT TERMINAL ERPOR. AT 163.84 SECS SHUT OFF ROUTINE TO LOAD ICDU DACS. DISPLAY ON DSKY: V06 N62 R1 - VI R2 - H DOT R3 - H VI-INERTIAL VELO-CITY MAGNITUDE. IN FPS TC NEAPEST FPS H DCT - RATE OF CHANGE OF VFHICLE ALTITUDE ABOVE LAUNCH PAC RADIUS. IN FPS TO NEAREST FPS

P11/COLOSSUS H-VEHICLE ALTITUDE #130 ABOVE THE LAUNCH PAD RADIUS. IN NAUTI-CAL MILES TO NEAREST .1 NM VERIFY SATURN SHUT DOWN #140 TERMINATE PI1 AND GO KEY IN V37EXXE TO PROGRAM SELECTED . VIA ROO #150 EXIT P11 EXIT P11

CHANGE CONTROL NOTES

LOGIC REV 5 PCR 3 LOGIC REV 6 PCR 206

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TPI SEARCH PROGRAM (P17)

CMC

LCGIC PEV 09 11/27/68

CHECKLIST

TIME

P17/COLOSSUS

TOTAL

PURPOSE:

- (1) TO ACCEPT A DESIRED TIME OF TRANSFER PHASE INITIATION (TIG(TPI)) AS A DSKY INPUT FROM THE ASTRONAUT. AND TO COMPUTE THEREFROM THE PARAMETERS ASSOCIATED WITH A MINIMUM ENERGY. SAFE PERIAPSIS TRANSFER MANEUVER AT TIG(TPI) AND THE RESULTANT RENDEZVOUS INTERCEPT.
- (2) TO PROVIDE THE ASTRONAUT WITH THE OPTION OF DEFINING TO THE CMC THE INITIAL TRANSFER TRAJECTORY SEARCH SECTOR FOR CENTRAL ANGLES EITHER GREATER THAN OR LESS THAN 180 DEGREES FROM THE POSITION OF THE ACTIVE VEHICLE (CSM) AT TIG (TPI).
- (3) TO DISPLAY TO THE ASTRONAUT THE PARAMETERS ASSOCIATED WITH THE TRANSFER (TPI AND INTERCEPT).

PROG

CONT

- ASSUMPTIONS: (1) IF P20 IS IN OPERATION WHILE THIS PROGRAM IS OPERATING THE ASTRONAUT MAY HOLD AT ANY FLASHING DISPLAY AND TURN ON THE RENDEZVOUS SIGHTING MARK ROUTINE (EITHER R21 OR R23) AND TAKE OPTICS MARKS AND/OR HE MAY ALLOW WHE RANGING MARKS TO ACCUMULATE. SEE P20 FOR DETAILED DESCRIPTION.
 - (2) THE OPERATION OF THIS PROGRAM UTILIZES THE ACTIVE VEHICLE FLAG WHICH DESIGNATES THE VEHICLE WHICH IS DOING THE RENDEZVOUS THRUSTING MANEUVERS TO THE PROGRAM WHICH CALCULATES THE MANEUVER PARAMETERS. SET AT THE START OF EACH RENDEZVOUS PRE-THRUSTING PROGRAM.
 - (3) TO EXECUTE THE TPI MANEUVER SELECT THE TRANSFER PHASE INITIATION (TPI) PROGRAM (P34).

CREW

(4) THIS PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

GROUND

			* 124 124 1	 TIME
		• CREW • PROGRAM •••SELECTION		
START TPI SEARCH PROGRAM (P17).	•	KEY IN TPI SEARCH PROGRAM (P17) V37E17E		
		:		#10
DISPLAY PROGRAM 17	•	MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM 17.		
•		•		
•		•		#20

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	•		•		,,,,
	•	•	•	• P17/COLOSS	US
	•	•	•	•	
	•	•	•	•	
		· ·		•	
WAIT FOR K	CE YBOARD	KEY	IN PROCEED .	•	
ENTRY	•		•	•	
			•	•	
		•		•	
	FLASH UPON	• •	KEY IN V25E AND LOAD THE DESIR-		#80
RECEIPT OF OR NEW DAT			ED TIG(TPI)	•	#60
		•			
• PROCEED	.NEW	•	•	•	
•	.DATA	•	•	•	
•	•	•	****	•••	
•	STORE DATA	•			
•		•			
•	•	•			#00
•	•	•			#90
•	******	•			
•					
RESET UPDA	ATE FLAG	•			
•					
•					
COMPLITE VE					#100
COMPUTE VE					
TIGITPI) A					
NOMINAL SE	EARCH OP-				
TION AS FO					
ABOVE PASS	VEHICLE IS				
	TIVE DELTA				
ALTITUDE)					
OPTION OOD					#110
	VEHICLE IS				
	SIVE VEHICLE DELTA ALT)				
SELECT OPT					
	•				
	•				
	-				
SET UPDATE					#120
	•				
	•				
	• • • • •	•		•••	

P17/COLOSSUS

	• •	•		•	•			
	• •	•		•	•			
HOLD .		•			•			
• • • • • • • •	FLASH VERB-NOUN TO	• • • • • • • • • • • • • • • • • • • •	MONITOR DSKY:		•			
SNAP .	REQUEST RESPONSE AND	•	ORSERVE VER		•			#130
	DISPLAY TPI PARA-	•	FLASH TO REQUE		•			
	METERS:	•	RESPONSE AND D	ISPLAY	•			
	V06N72	•	OF TPI PARAMET	ERS.	•			
	RI-DELT ANG(TPI)	•			•			
	R2-DELT ALT(TPI)	•	•		•			
	R3-SEARCH OPTION	•	•		•			
		•	•		•			
	DELT ANG(TPI)-CENT-	•	•		•			
	RAL ANGLE AROUND THE	•	•		•			
	EARTH OR MOON AT TIG		•		•			#140
	(TPI) BETWEEN THE	•	•		•			
	ACTIVE AND PASSIVE	•	•					
	VEHICLES. PCLARITY	•	•		•			
	INDICATES ACTIVE VE-	•	•		•			
	HICLE IS BEHIND (-)	_	_		-			
	OR AHEAC OF (+)	•	_		-			
	PASSIVE VEHICLE. IN	-	•		.			
		•	•		•			
	DEGREES TO NEAREST •01 DEGREE•	•	•		•			
	OI DEGREE	•	•		•			#150
	DC: T 4: T4TDT1 THE	•	•		•			#1.20
	DELT ALT(TPI)-THE	•	•		•			
	MAGNITUDE OF THE	•	•		•			
	ALTITUDE DIFFERENCE	•	•		•			
	BETWEEN THE POSITION	•	•		•			
	VECTORS OF THE ACT-	•	•		•			
	IVE AND PASSIVE VE-	•	•		•			
	HICLES AT TIG(TPI).	•	•		•			
	IN NAUTICAL MILES TO	•	•		•			
	THE NEAREST .1 NM.	•	•		•			
	POLARITY INCICATES	•	•		•			#160
	THE PASSIVE VEHICLE	•	•		•			
	IS ABOVE (+) OR BE-	•	•		-			
	LOW (-) THE ACTIVE	•	•		•			
	VEHICLE.	•	•		•			
		•	•		•			
	SEARCH OPTION-AN CP-	•	•		•			
	TION CODE TO CONTROL		•		•			
	CMC SEARCH FOR TERM-				•			
	INAL PHASE	•	•		-			
	DEFINITION:	-	-		•			#170
	00001-CMC SEARCHES	-			-			
	TO DEFINE A TERMINAL		-		-			
	PHASE WHERE THE REN-		<u>-</u>		•			
	DEZVOUS INTERCEPT		•		•			
	0000100 AT A 051170A1	•	•		•			
	ANCHE LECE THAN 100	•	•		•			
		•	•		•			
	DEGREES FROM THE	•	•		•			
	ACTIVE VEHICLE PCS-	•	•		•			
	ITION AT TIGETPI).	•	•		•			#100
	OR OCCUPATIONS	•	•		•			#180
	00002-SAME AS	•	•		•			
	00001 EXCEPT CENTRAL	•	•		•		D17/00/ 00000	
		•	•		•		P17/COLOSSUS	
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)
P17/COLOSSUS	
	#190
	#200
	#210
	#220
	#230

ANGLE IS GREATER THAN 180 DEGREES. WAIT FOR KEYBOARD ENTRY. DO I WISH TO CHANGE TIG(TPI)? KEY IN RECY-. CLE V32F GO TO . MAM ABOVE DO I WISH TO CHANGE THE SEARCH OPTION DISPLAYED? RECORD THESE VALUES KEY IN PROCEED

THIS PAGE INTENTIONALLY LEFT BLANK

	.NO .ALARM	.ALARM					#290
		SET UP- CATE FLAG (SEE P20)					
POSS	•	:					#300
HOLD .	TO REG SPONSE PLAY F ALARM VOS R1- R2- R3-	SNO9 - - - IED ALARM ST THIS	•••••••••••••••••••••••••••••••••••••••	MONITOR DOF: ALARM CMC CAL UTION.	R DSKY: S PROGRAM INDICATE THAT N FIND NO SOL-		#310
	•	S 00124		•	•		#320
	. UPON F	.R .E .C	•	(TP	ADJUST TIG I) AND/CR THE RCH OPTION IN RECYCLE F		#330
	: : :	•C •L •E •			GO TO "A" ABCVE		#340
	•			•		P17/COLOSSUS	

	•		•		111002000	
	•		•			
	•		•			
	•		•			
	• GO TO		•			
	• "Д"		-			
	. ABOVE		_			
	- 40046		•			
	•		•			
	•		•			
HOLD .		•				
	FLASH VERB-NOUN TO		MONITOR DSKY:			
SNAP .	REQUEST RESPONSE AND	***************************************	OBSERVE VER	R_MOUN		
SNAP .		•				4350
	DISPLAY CALCULATED		FLASH TO REQUE			#350
	TERMINAL PHASE		RESPONSE AND D	[SPLAY		
	PARAMETERS:		OF CALCULATED	TER-		
	V06N58		MINAL PHASE			
	R1-PER ALT		PARAMETERS.			
			PARAMETERS.			
	R2-DELTA V(TPI)					
	R3-DELTA V(TPF)		•			
	PER ALT-ALTITUDE OF					
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	PERIGEE ABOVE LAUNCH					
	PAC RADIUS (EARTH		DO I WISH TO A			#360
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	OF PERILUNE ABOVE		SEARCH OPTION?			
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	DELTA V(TPI)-REQUIR-			•		#370
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	ED IMPULSIVE DELTA V		•	•		
	TO ACCOMPLISH TPF		•	•		
	MANEUVER AT TIME OF		•	•		#380
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P17/COLOSSUS

CHANGE CONTROL NOTES

LOGIC REV 08 PCR MIT 66 LOGIC REV 09 EDITORIAL THIS PAGE INTENTIONALLY LEFT BLANK

PURPOSE: (1) TO CONTROL THE CSM ATTITUDE AND OPTICS TO ACQUIPE THE LM IN THE SXT FIELD AND TO POINT THE CSM TPANSPONDER AT THE LM ALONG THE +Y AXIS BASED ON THE OPTION CODE CHOSEN.

(2) TO UPDATE EITHER THE LM OR CSM STATE VECTOR (AS SPECIFIED BY THE ASTRONAUT BY DSKY ENTRY) ON THE BASIS OF OPTICAL TRACKING DATA ANC/OR WHE PANGE DATA.

ASSUMPTIONS: (1) THE IMU MUST BE ON AND ALIGNED IN THE TO PERFORM THIS PROGRAM.

(2) THE CNCS IS IN CONTROL OF THE VEHICLE IN THE AUTO MODE IN THE NOMINAL CASE. IF THE ASTPONAUT TAKES OVER CONTROL OF THE VEHICLE WITH PHO THE CSM WILL REMAIN AT THE ATTITUDE IT IS DRIVEN TO: REGARDLESS OF MODE SELECTION THE SNCS WILL CALCULATE THE PREFERRED TRACKING ATTITUDE AND THE +X-AXIS TRACKING ATTITUDE.

(3) POUTINE ROB HAS BEEN DEREORMED PRIOR TO SELECTION OF THIS PROGRAM. IN OPDER FOR THE GEOS TO PERFORM THE AUTOMATIC ATTITUDE MANELYERS THE ASTRONAUT SHOULD KEY IN VAKE AT SOME TIME PRIOR TO THE FIRST MANELYER.

(4) THE LM IS MAINTAINING A DREFERRED TRACKING ATTITUDE TO CORPECTLY PRIEMT THE OPTICAL BEACON.

(5) THE OPERATION OF THE PROGRAM INCLUDES THE FOLLOWING FLACS:

PENDEZVOUS ELAC- CONTROLS THE PERMANENT TERMINATION OF THE TOTAL PENDEZVOUS NAVIGATION PROCESS. THIS PROGRAM WILL ONLY PUN OF PESSIME RUNNING WHEN THIS FLAG IS SET. SET BY PZO SELECTION. PESSET BY SELECTION OF CMC IDLING PROGRAM(POOL), CMC POWER DOWN PROGRAM(POOL), THE OPRITAL MAVIGATION PROGRAM (PZZ), THE CISLUMAR NAVIGATION PROGRAM (PZZ), OP BY V56E. THE KEYING IN OF V56E WILL IMMEDIATELY TERMINATE PZO UNLESS A NAVIGATION MEASUREMENT IS REING PROCESSED IN WHICH CASE IT WILL HOLD UNTIL COMPLETION OF THE INCORPORATION AND THEN TERMINATE PZO.

TRACK FLAC- CONTROLS THE TEMPORARY TERMINATION OF THE TOTAL RENDEZVOUS NAVIGATION PROCESS. PESET OF THIS FLAG INTERPUTES THE AUTOMATIC ATTITUDE MANEUVER/OPTICS POINTING PROCESS, AS WELL AS THE STATE VECTOR UPDATE PROCESS. THE SET OF THIS FLAG ENABLES ALL THESE PROCESSES. SET BY 917, 20, 30, 31, 34, 35, 38, 39, 74, 75, 76, 77, 78, 79 SELECTION, PESET BY AN V37EXXE

UPDATE FLAG- CONTROLS THE TEMPORARY TERMINATION OF THE STATE VECTOR UPDATE PROCESS ONLY. SET BY P17, 20, 30, 31, 34, 35, 38, 39, 74, 75, 77, 78, 79 SELECTION. RESET BY ANY VATEXXE AND IT IS ALSO RESET AND SET DUPING THE PRETHOUST COMPUTATIONS TO PROTECT ERASABLE MEMORY.

PREESPEN ATTITUDE ELAG-DEFINES WHICH ATTITUDE THE SPACECRAFT IS TO BE ALIGNED TO BY R61. SET BY TURN ON THE P20. OF V765. RESET BY V775.

STICK FLAG - RESET BY FYECUTION OF A PROGRAM CHANGE VIA ROO AND BY VERE, SET BY TAKING RHC OUT OF DETENT WHEN THE SC CONTROL SWITCH IS UMO AND WHEN THE THO IS NOT CLOCKWISE. COURATE DRIVE IS NOT PERFORMED IF THE STICK FLAG IS SET.

STATE VECTOR FLAS - REFINES WHICH STATE VECTOR WILL BE HARDSTED BY SIGHTING MARKS AND VHE RANGING. SET TO LM BY ARREST TO LM BY ARREST TO CSM BY VAIE.

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VHE PANCE FLAG - SET BY VAZE, DESET BY VARE, AND BY ANY EXECUTION DE A PROCEAM CHANGE VIA ROO. ALLOWS AUTOMATIC WHE PANCE DATA TO BE USED BY THE PENDEZVOUS DATA PROCESSING ROUTINE (P22).

(6) THIS PROGRAM IS SELECTED BY THE ASTROMAUT BY DSKY ENTRY. IT MAY BE TERMINATED BY THE SELECTION OF THE CMC IDLING PROGRAM (P00), CMC POWER ORWN PROGRAM (P06), THE ORBITAL NAVIGATION PROGRAM (P22), THE CISLUMAR NAVIGATION PROGRAM (P23) OR BY V56F. POO SELECTION WILL TERMINATE P20 AND ANY OTHER PROGRAM IN PROCESS AND ESTABLISH P00. ALL OTHER PROGRAMS SELECTED WILL CHANGE THE PROGRAM NUMBER DISPLAYED ON THE DSKY BUT WILL NOT TERMINATE P20. THIS PROGRAM IS DESIGNED TO OPERATE AUTOMATICALLY AND SIMULTANEOUSLY WITH ANOTHER PROGRAM WITHOUT REQUIPING USE OF THE DSKY UNLESS NON-NOMINAL CIRCUMSTANCES REQUIRE CMC COMMUNICATION WITH THE ASTROMAUT. IF V56F IS KEYED INTO THE DSKY AND P20 IS THE ONLY PROGRAM RUNNING, POUTINE ROO WILL BE INITIATED.

(7) W MATRIX INITIALIZATION FOR RENDEZVOUS MAY BE ENABLED (PESET PENDWELAG) IN ANY OF THE FOLLOWING WAYS:

- (A) KEYING IN DE VERR 93E
- (B) COMPUTER ERESH START (KEYING IN OF VERR 36E)
- (C) STATE VECTOR UPDATE FROM THE GROUND
- (D) STATE VECTOR CALCULATION MADE DURING ENTRANCE TO P22, OR P23.
- (F) CHANGES TO INITIALIZATION PARAMETERS IN V67

(8) THERE IS A PENDEZVOUS OPTICS MARK COUNTER USED IN THE CMC TO COUNT THE NUMBER OF OPTICS MARKS USED TO CHANGE EITHER STATE VECTOR AND THERE IS A RENDEZVOUS VHE PANGING MAPK COUNTER USED IN THE CMC TO COUNT THE NUMBER OF VHE RANGING MAPKS USED TO CHANGE FITHER STATE VECTOR.

THESE COUNTERS ARE TERPED BY SEVERAL DISTINCT EVENTS, THEY ARE:

- (A) KEYING IN OF V37E20E (INITIATION OF (P201).
- (B) COMPLETION OF THE TARGET DELTA V PROGRAM (P76).
- (C) SELECTION OF A NEW PROGRAM FROM A PROGRAM WHICH HAD TUPNED ON AVERAGE G (SEE ROO LOGIC).
- (D) INITIALIZATION OF THE W MATRIX FOR PENDEZVOUS (FOR ANY REASON, SEE ASSUMPTION 7).
- (E) SELECTION OF RETURN TO EARTH PROGRAM (P37)
- SUMMARY OF EXTENDED VERRS ASSOCIATED WITH THE PROGRAM:

 V44E SET SURFACE FLAG-CAUSES TRACKING OF THE LANDING SITE BASED ON THE STORED LANDING SITE

 V45E RESET SURFACE FLAG-CAUSES TRACKING OF THE LANDING SITE BASED ON THE LANDING SITE

 V54E DO R23 ALLOWS BACK-UP MARKING ON THE LANDING P20.

 V56E RESET THE RENDEZVOUS FLAG. CAUSES TERMINATION OF P20.

 V57E DO R21 ALLOWS OPTICS MARKING ON THE LANDING P20.

 V58E PESET STICK FLAG ALLOW AUTO MANEUVERS

 V67E W-MATRIX RMS EPROR DISPLAY

 V76F SET PREFERRED ATTITUDE FLAG DRIVE TO PREFERRED ATTITUDE

 V77F PESET PREFERRED ATTITUDE FLAG DRIVE TO +X-AXIS ATTITUDE

 V90F SET STATE VECTOR FLAG TO LANDING WILL UPDATE LANDING VECTOR

 V91F SET STATE VECTOR FLAG TO CSM. DATA WILL UPDATE CSM STATE VECTOR

 V97F SET STATE VECTOR FLAG TO CSM. DATA WILL UPDATE CSM STATE VECTOR

 V97F SET VHE PANGE FLAG ALLOWS R22 TO ACCEPT RANGE DATA.

 V97F PESET VHE PANGE FLAG STORS ACCEPTANCE OF PANGE DATA.

 V97F PESET VHE PANGE FLAG STORS ACCEPTANCE OF PANGE DATA.

 V97F PESET PENDWELAG CAUSE INITIALIZATION OF W MATRIX FOR RENDEZVOUS AT MEXT DATA INCORPORATION

P20/COLOSSUS

PROG

CONT

START PEMPEZVOUS

(201.

DISPLAY P20

MAYTCATION PPOGPAM

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INCOMPLETE SUMMARY OF ADDITIONAL DATA OR JOBS AVAILABLE WHILE THIS PROGRAM IS OPERATING:
    PIT - AVAILABLE BY KEYING IN VSTELTE
    P30 - AVAILABLE BY KEYING IN V37E30E
    P31 - AVAILABLE BY KEYING IN V37E31E
    P34 - AVAILABLE BY KEYING IN V37E34E
    P35 - AVAILABLE BY KEYING IN V37E35E
    P38 - AVAILABLE BY KEYING IN V37E38E
    P39 - AVAILABLE BY KEYING IN V37E39E
    P74 - AVAILABLE BY KEYING IN V37E74E
    P75 - AVAILABLE BY KEYING IN V37E75E
    P76 - AVAILABLE BY KEYING IN V37E76E
    P77 - AVAILABLE BY KEYING IN V37E77E
    P78 - AVAILABLE BY KEYING IN V37E7BE
    P79 - AVAILABLE BY KEYING IN V37E79E
    RO3 - AVAILABLE BY KEYING IN VARE
    R30 - AVAILABLE BY KEYING IN V825
    P31 - AVAILABLE BY KEYING IN VR3E
    P34 - AVAILABLE BY KEYING IN V85E
    NO9 - AVAILABLE BY KEYING IN VISNOSE
    N20 - AVAILABLE BY KEYING VIENZOE
    N22 - AVAILABLE BY KEYING V16N22E
    N36 - AVAILABLE BY KEYING V16H36F
    NOT - AVATLABLE BY KEYING VIGNOTE
    N92 - AVAILABLE BY KEYING V16N92E
    Mg5 - AVATLABLE BY KEYING VIANGE
    N96 - AVAILABLE BY KEYING VI6N96F
    V46E- AVAILABLE BY KEYING IN V46E
    V60 - AVAILABLE BY KEYING V60F
    V61 - AVAILABLE BY KEYING V61E
    V62 - AVAILABLE BY KEYING V62F
    V63 - AVAILABLE BY KEYING V63E
    V96F- AVATLABLE BY KEYING V96E
                                                                                                  TIME
                                                                                                             TOTAL
CMC
                  CECUMO
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                                                                                                             TIME
                                         .CPEW
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```

KEY IN RENDEZVOUS

****** PROGRAM

(P20)

V37F20F

1A

#10

P20/CPLPSSUS

1A

	•	MONTTOR DSKY: OBSERVE DISPLAY OF P20		#20
· ·				
DO IMU STATUS CHECK ROUTINE (ROZ)	•	DO IMU STATUS CHECK POUTENE (ROZ)		
• •				#30
SET THE PREEFPRED ATTITUDE FLAG				
• •				
SET THE STATE VECTOR				#40
• •			٠	
SET TPACK FLAG				
•				#50
SET HERBATE FLAG				
: :				
SET PENDEZVOUS FLAG				#60
? FRO THE RENDEZ VOUS				
JOSTIC MARK CUMALES.				
•			PZO/COLOSSUS	#70

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#80

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#100

#110

#120

P20/CPLPSSIS

ZERO THE RENDEZVOUS VHE PANGING MARK COUNTER. EXTRAPOLATE PERMA-NENT STATE VECTORS (LM AND CSM) TO THE PRESENT TIME USING PRECISION INTEGRAsEbu IS TRACK FLAG SET? FXIT TS THE REFSMMAT FLAC SET?

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FXIT

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#130

#140

#150

#160

P20/CCLOSSUS

	:		
	CALL THE RENDEZVOUS DATA PROCESSING ROUTINE (R22)		THE RENDEZVOUS DATA PROCESSING POUTINE IS NOW AUTOMATIC
++ +2? ++ PCR 606	: : :		
5 0	SET R61 COUNTER EQUAL TC ZERO.		
•	SET LM TARGET FLAG FOR USE BY AUTO OPTICS POSITIONING POUTINE (R52).		
	•		
	DO TPACKING ATTITUDE POUTINE (P61) WHICH WILL CALL ATTITUDE MANEUVER POUTINE (P60)	•	DO TRACKING ATTITUDE ROUTINE (R61) WHICH WILL CALL ATTITUDE MANEUVER ROUTINE (R60)
	: :		
++ +2? ++ PC R 606	: : :		
	•		*******
	:		

CALL THE AUTO DPTICS POSITIONING POUTINE (P52)		THE AUTO OPTIC ITIONING POUTI (P52) AND THE ING ATTITUDE R	NE TPACK-
:		QR61) APE NOW MATIC.	
• • • • •		Y .	N
•		DO I DESIRE TO SELECT CMC IDL PROGRAM(POO)?	
WAIT FOR KEYBOARD		SFLECT CMC IDLING PROGRAM	• N
	•	(POO). KEY IN V3 ⁷ FOOF	•
PESPONSE TO VS6F OR V37F00E IMMEDIATELY.		 кғү V56	TN F
•V •V •5 •3 •6 •7 •F			•
• • • • • • • • • • • • • • • • • • •		Fx	IT ₽20

P20/COLOSSUS

#170

#180

#190

#200

#210

#220

PROJECT USSUS

1A

GO TO CMC DBSERVE IDLING PROGRAM DISPLAY OF POO. (POO) VIA ROO (NOTE: POO WILL TERMINATE THE PENDETVOUS, TRACK+ AND UP-FXIT P20 DATE FLAGS AND ALSO THE AUTO OPTICS POST-TIONING ROU-TINE (P52)) AND THE PENCY-VOUS DATA PRC-CESSING ROUT-INF (R22)

#240

#230

TERMINATE THE REN-DEZVOUS, TRACK, AND, UP-DATE FLAGS AND THE AUTO OPTICS POSTTION-ING ROU-TINE [352] AND THE PENDEZ-VIUS DATA

PROCESS-

THE (R22)

#250

4260

#270

IS P20 THE ONLY PRO-ERAM RUN-NING?

P20/CDLOSSUS

הפטערנין טככווכ

FX11 5V11 P20 920 OD ROUTINF ROO FX11 P20 FX11 P20 P20	• •	•	
P20 P20 PD POINTINE POO		•	
P20 P20 PD POINTINE POO	• •	•	
P20 P20 PD POINTINE POO		•	
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#290

CHANGE CONTROL MOTES

1001C REV 18 PCR NASA 147
LDGTC REV 19 PCR MIT 66
LDGTC REV 20 PCR 468
PCR 206
LDGIC REV 21 PCN 579
LDGTC REV 22 PCR 606

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PROG

GROUND TRACK DETERMINATION PROGRAM (P21)

LCGIC REV 66 07/08/68

PURPOSE: (1) TO PROVIDE THE ASTRONAUT DETAILS OF HIS CROWND TRACK WITHOUT THE NEED FOR GROWND COMMUNICATION.

ASSUMPTIONS: (1) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

(2) THIS PROGRAM MAY BE SELECTED WHILE THE CSM IS IN FITHER EARTH OR LUNAR ORBIT TO DEFINE THE GROUND TRACK OF FITHER THE LM CR CSM.

(3) THIS PROGRAM ASSUMES THE VEHICLE WHOSE GROUND TRACK PARAMETERS ARE CALCULATED TO REMAIN IN FREE FALL FROM THE PRESENT TIME UNTIL T LAT LONG.

CMC		GRCUND	CREW		CHECKEEST	TIME	TOTAL TIME
			•CREW •PRCGRA* •SELECT				
START GROUNT TO DETERMINATION PROGRAM (P21). DISPLAY PROGRA			KEY IN GROUND TO DETERMINATION PROGRAM (P21) V37E21E	RACK			#10
			•				
			MONITOR DSKY: OBSERVE DISPLAY PROGRAM 21	n F			#20
•			•				<i>**</i> • • • • • • • • • • • • • • • • • •
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•	•	•	•	•		P21/LUMINARY	

#140

#150

#160

#170

#180

CALCULATE LATITUDE, LONGITUDE AND ALTI-TUDE OF VEHICLE AT T LAT LONG MONITOR DSKY: HOLD .. FLASH VERB-NOUN TO REQUEST RESPONSE AND OBSERVE VERB-NOUN FLASH TO REQUEST DISPLAY LATITUDE, SNAP .. LONGITUDE AND RESPONSE AND DISPLAY OF LATITUDE, LONGI-ALTITUDE: V06 N43 TUDE AND ALTITUDE R1-LAT R2-LONG R3-ALT LAT-LATITUDE OF VEHICLE. + IS NORTH. IN DEGREES TO NEAR-EST .O1 DEGREE. LONG-LONGITUDE OF VEHICLE. + IS EAST. IN DEGREES TO NEAR-EST .01 DEGREE. ALT-ALTITUDE OF VEHICLE ABOVE THE LAUNCH PAD RADIUS (EARTH ORBIT) OR THE LUNAR RADIUS AT THE MOST RECENTLY DEFINED (SEE SECTION 5 OF R5771 LANDING SITE (LUNAR ORBITI. IN NAUTICAL MILES TO NEAREST .1 P21/COLOSSUS P21/LUMINARY LOGIC REV 6 PCR 206

ORBITAL NAVIGATION PROGRAM (P22)

LOGIC REV 11 11/26/68

PUR PO SE:

- (1) TO LOCATE AND TRACK A LANDMARK SUITABLE FOR NAVIGATION PURPOSES.
- 12) TO OBTAIN SIGHTING MARKS ON THE CHOSEN LANDMARK.
- (3) TO CALCULATE THE ORBITAL PARAMETER CHANGES GENERATED BY THE LANDMARK SIGHTING MARKS.
- (4) TO DISPLAY THE CRBITAL PARAMETER CHANGES GENERATED BY THE FIRST SIGHTING MARK ON A LANDMARK, FOR DECISION BY THE NAVIGATOR/GROUND AS TO THE VALIDITY OF THE LANDMARK AND NAVIGATION PROCESS PRIOR TO INCORPORATION OF STATE VECTOR CHANGES AS A RESULT OF THE SIGHTING MARKS.
- (5) TO PROVIDE UPCATED COORDINATES OF THE KNOWN LANDMARKS.
- (6) TO PROVIDE COORDINATES OF UNKNOWN LANDMARKS.
- (7) TO TRACK A PRELOADED LANDING SITE.
- (8) TO PROVIDE COORDINATES OF A NEW LANDING SITE(TREATED AS AN UNKNOWN LANDMARK)
- (9) TO PROVIDE CORCINATES OF AN OFFSET LANDING SITE.
- (10) TO POINT THE OPTICS ALONG AN ADVANCED ORBIT GROUND TRACK FOR THE PURPOSE OF TRACKING AND MAPPING A NEW LANDING SITE.

- ASSUMPTIONS: (1) THERE ARE 25 KNOWN LUNAR LANCMARKS STORED IN FIXED MEMORY IN THE CMC AND ONE LANDMARK MAY BE STORED IN ERASABLE MEMORY IN THE CMC. THE LANDMARK STORED IN ERASABLE MEMORY IS REFERRED TO AS THE LANDING SITE AND IS DESIGNATED BY LANDMARK CODE NUMBER 01, THIS CODE IS AVAILABLE FOR LUNAR LANDMARK STORAGE ONLY.
 - (2) THERE ARE TWO TYPES OF LANDMARK TRACKING METHODS:
 - (A) "KNCHA" LANDMARK TRACKING THE TRACKING OF AN EARTH LANDMARK MADE KNOWN TO THE CMC BY LATITUDE, LONGITUDE OVER 2, AND ALTITUDE AND THE TRACKING OF A LUNAR LANDMARK MADE KNOWN TO THE CMC BY ITS LANDMARK CODE NUMBER OR BY LATITUDE. LONGITUDE OVER 2. AND ALTITUDE.
 - (B) "(UNKNOWN)" LANDMARK TRACKING THE TRACKING OF A LANDMARK OR SURFACE FEATURE IDENTIFIED TO THE CMC AS AN UNKNOWN LANDMARK, ONE WHOSE COORDINATES ARE NOT KNOWN.
 - (3) THERE ARE TWO TYPES OF LANDING SITE MAPPING METHODS, IN EITHER CASE THE LANDING SITE COORDINATES MAY BE STORED IN CMC MEMORY (REFER TO ASSUMPTION 1), THEY ARE:
 - (A) LANDING SITE DESIGNATION TRACK AND MARK ON AN UNKNOWN LANDMARK. STORE THE RESULTING COORDINATES IN LANDMARK CODE G1. IF MAPPING ONLY IS DESIRED, I.E. NO STATE VECTOR CALCULATION OR CORRECTIONS, THE ASTRONAUT NEED TAKE CALY ONE MARK.

- (B) LANDING SITE OFFSET WHILE TRACKING AND MARKING ON A PRIMARY LANDMARK (KNOWN OR UNKNOWN), POINT THE OPTICS SLOS AT THE CHOSEN LANDING SITE AND MARK IT ONCE, (AT LEAST ONE MARK ON THE PRIMARY LANDMARK MUST HAVE BEEN MADE PRIGR TO THIS), THEN CONTINUE MARKING ON THE PRIMARY LANDMARK. STORE THE RESULTING COORDINATES OF THE OFFSET LANDING SITE IN LANDMARK CODE 01. THE ASTRONAUT HAS TWO WAYS OF DEFINING TO THE CMC WHICH MARK WAS MADE ON THE OFFSET LANDING SITE. THEY ARE:
 - 1. KEY IN V52E AFTER MARKING ON THE OFFSET LANDING SITE. THIS WILL SET THE INDEX OF OFFSET DESIGNATOR EQUAL TO THE VALUE OF THE MARK COUNTER.
 - 2. SIMPLY MARK ON THE OFFSET LANDING SITE BUT MAKE A MENTAL NOTE OF WHICH MARK IN THE SEQUENCE IT WAS AND THEN SET THE INDEX OF OFFSET DESIGNATOR TO THAT VALUE WHEN IT IS DISPLAYED FOLLOWING THE SIGHTING MARK ROUTINE.
- (4) ACQUISITION OF A LANDMARK MAY BE AIDED BY THE CMC BY USE OF THE AUTOMATIC OPTICS POSITIONING ROUTINE (R52). CARE SHOULD BE EXERCISED WHEN AN UNKNOWN LANDMARK IS CHOSEN TO KEEP THE OPTICS OUT OF THE CMC MODE TO AVOID POSSIBLE PROGRAM ALARMS.
- (5) ACQUISITION OF A PRELCADED LANDING SITE MAY BE ALDED BY KEYING LANDMARK CODE OF INTO THE VO5N70 DISPLAY FOR USE BY THE AUTOMATIC OPTICS POSITIONING ROUTINE (R52). TO IMPROVE THESE COORDINATES REFER TO ASSUMPTION 3.
- (6) DURING LUNAR ORBIT WHILE IN THE CMC IDLING PROGRAM (POO) THE LUNAR LANDMARK SELECTION ROUTINE (R35) IS AVAILABLE TO AID THE CREW IN THE SELECTION OF APPROPRIATE LANDMARKS PRIOR TO THE SELECTION OF THIS PROGRAM.
- (7) THE GROUND TRACK DETERMINATION PROGRAM (P21) IS AVAILABLE TO AID THE CREW IN CHOOSING APPROPRIATE LANDMARKS PRIOR TO SELECTION OF THIS PROGRAM.
- (8) THE GROUND TRACK DETERMINATION PROGRAM (P21) IS AVAILABLE TO THE CREW FOLLOWING THIS PROGRAM TO PROVIDE UPDATED GROUND TRACK INFORMATION.
- (9) POSSIBLE ATTITUDE CONTROL METHODS MIGHT BE AS FOLLOWS (IN ALL CASES CARE MUST BE TAKEN TO MONITOR POSSIBLE IMPENDING IMU GIMBAL LOCK).
 - (A) MANUAL CONTROL BY THE PILOT OR NAVIGATOR WITH THE ROTATIONAL HAND CONTROLLER.
 - (B) MANUAL RATE CONTROL BY THE NAVIGATOR WITH THE MINIMUM IMPULSE CONTROL IN THE GNC FREE MODE.
- (10) THE PROGRAM MAY BE PERFORMED WITH SIVB ATTACHED IF THE LAUNCH VEHICLE GUIDANCE SWITCH IS PLACED IN THE CMC POSITION THERBY PERMITING SIVB ATTITUDE CONTROL WITH THE ROTATIONAL HAND CONTROLLER. GNC A/P CONTROL IS REQUIRED IN THIS CASE.
- (11) THE IMU MUST BE CN AND ALIGNED IN ORDER TO COMPLETE THIS PROGRAM.
- (12) SELECTION OF THIS PROGRAM WILL TERMINATE THE RENDEZVOUS NAVIGATION PROGRAM (P20).
- (13) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG CONT

CMC	GROUND	CREW	CHECKLIST	TIME	TOTA TIM
		. CREW PROG SELECTION			
START ORBITAL NAVIGATION PROGRAM (P22) DISPLAY PROGRAM 22	•	KEY IN ORBITAL NAVIGATION PROGRAM (P22) V37E22E			#10
		•			#10
	•	MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM 22			
•					#20
DO IMU STATUS CHECK RCUTINE (RO2)	•	DO IMU STATUS CHECK ROUTINE (RO2)			
•	•				#30
RESET THE RENDEZVOUS FLAG					
•.					
COMPUTE ANGLE BETTWEEN Y AND V X R.					#40
THIS ANGLE IS THE MAGNITUDE OF THE MAXIMUM POSSIBLE MIDDLE GIMBAL ANGLE ASSUMING THE SC X AXIS IS KEPT IN THE ORBIT PLANE.					
•					#50

P22/COLOSSUS

	•			P22/COLOSSUS
	•			
	•			
++ +11 ++	RESET RENCWFLAG (SEE P20)			
PCN 580	•			
	•			
	5. LOW DEED HOUR TO		HOUTTON DOWN.	#60
HOLD .	FLASH VERB-NOUN TO REQUEST RESPONSE AND DISPLAY MAGNITUDE OF MAX POSSIBLE MIDDLE GIMBAL ANGLE IF THE X AXIS IS HELD IN THE ORBIT PLANE. VO6 N45 R1:BLANK	•	MONITOR DSKY: OBSERVE VERB-NOUN FLASH TO REQUEST RE- SPONSE AND DISPLAY OF MAGNITUDE CF MAX POSSIBLE MIDDLE GIMBAL ANGLE IF THE SC X AXIS IS KEPT IN THE ORBIT PLANE	
	R2:BLANK			#70
	R3:MGA IN DEG. TO NEAR-		•	
	EST .01 DEG.			
	•		IS MGA GREATER THAN	
	•		60 DEG? IF SO, IT IS ADVISABLE TO REALIGN	
	•		THE IMU IF TIME PER- MITS.	
	•			#80
	•		•Y •N	
	•			
	WAIT FOR KEYBOARD		. HAS THE ISS BEEN	•
	ENTRY:		• ALIGNED IN THE • PAST 3 HOURS?	
			• • • • • • • • • • • • • • • • • • •	
			• • •	#90
		*******	• • •	
		ADVISE CREW	SHALL I BYPASS . IMU ALIGNMENT? .	
		PATED G+N .	•	
		ERRORS AND Subsequent	•N •Y •	
		PROCEDURES	• •	
			• • • • •	#100
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		SELECT IMU	•
		REALIGN (P52).	
		MUEN COMPLETE	•
		WHEN COMPLETE.	•
		RESELECT THIS	•
		PROGRAM.	•
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		EXIT P22	•
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ERMINATE FLASH UPON			•
ECEIPT OF PROCEEC	•		
R NEW PROGRAM	• • • • • • • • • • • • • • • • • • • •	KEY IN PROCEED	
	•		*
.P .NEW		•	
.R .PRCG		_	
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•E			
•E EXIT		CONTROL ATTITUDE	
•D P22		FOR LANDMARK	
		ACQUISITION	
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• "A"			
• SEE BELOW			
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• •			
ET THE INDEX OF			
FFSET DESIGNATOR			
QUAL TO ZERO.			
A04F IA 55 MA			
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ESET MARK CATA			
ALID BIT FOR INFOR-			
MEID DIE ECK INCCK-			
ATION TO DOWNLINK.			
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•		#A #	
		SEE BELOW	
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	IS THE SPACE IN LUNAR ORE			IS THE SPACEC LUNAR ORBIT?	RAFT IN	•	#160°
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	•	•		•	•		
	GO.TO	•		•	60.10		
	#B# BELOW	•		•	"B" BELOW		#170
	BELOW	•		•	DECOM	· ·	#170
		•		•			
		•		A LIMAR CREAT			
		•		LUNAR ORBIT			
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		•		•			
		•		•		•	# 180
		•		•	•	·	****
POSS							
HOLD .	FLASH VERB REQUEST RES!			MONITOR DSKY: OBSERVE VERB			
SNAP .	. DISPLAY LMK		•	FLASH TO REQU			
	 V05N70 		•	RESPONSE AND			
	. RI-BLANK			LMK CODE.	-		•
	R2 ABCDE			TO HAVE THE A			
	•			ING ROUTINE (1	#190
	. A=1 IF KNOW!			POINT ALONG T			
	. A=2 IF UNKNO . B= INDEX OF			GROUND TRACK ADVANCED ORBT			
	. DESIGNATO			60 DEGREES AH			
	• C= NOT USED			THE SPACECRAF			
	• PROGRAM	7 61		CHANGE THE LA			
++ +11	. DE-LMK ID NO	N		CODE TO PX (W >3 AND X DEFI			
++	. NOTE: THE PE	ROBABLE		MANY DRBITS A			
PCN		IS ARE:		ARE DESIRED).		;	#200
5 94	. A=1. DE=	=C'O+ NDMARK		CODE IS USED THE AUTOMATIC			
		EWN BUT		POSITIONING R			
		TSTORED		(R52) AT THIS	TIME		
		CMC HDRY					
	. A=1. DEX			•	•		
	. LAN	IDMARK		•	•		
		IWN AND IRFD		•	•		
	. 311	. 下てひ		•	•	P22/COLOSSUS	# 210
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102	•			•	•		
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103				,)
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++ +11 ++ FDIT	A=2. DE NOT MEANINGFUL: UNKNOWN LANDMARK.			•	• • • • •	
			GATE ON	H TO NAVI-	•	#220
			.Y		• • •	
			. USE (. MATIC . POSI . ROUT	OF THE AUTO- C OPTICS TIONING INE (R52) ID IN ACQUI-		#230
			• • N:	.Y O I WISH TO AVIGATE ON HIS LAND— ARK?	• • • • • • • • • • • • • • • • • • •	#240
			SET OP- TICS MODE SWITCH TO MAN- UAL.	•	• • • • • • • • • • • • • • • • • • •	∉ 250
	WAIT FOR KEYBOARD ENTRY	•	•	KEY IN V22F AND LOAC NEW LMK CODE	• • • • • • • • • • • • • • • • • • •	#260
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TERMINATE	FLASH UPON	• • • • • • • • •	PROCEE	0		
RECEIPT O	F PROCEEC	-				
OR NEW DA	.TA					
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NEW DATA	•					
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	<u> </u>	•	•	
	IS DE > 0?	•	•	
	STORED LANDMARK) <u>.</u>	•	
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	.N .Y	•	•	#320
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		•	• •	#330
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POS S	. FLASH VERB-NOUN	•	MONITOR DSKY: .	
HOLD .	. TO REQUEST RESP-		OBSERVE VERB-NOUN .	
******	. ONSE AND DISPLAY		FLASH TO REQUEST .	
	. LMK COORDINATES:	•	RESPONSE AND DISPLAY .	
	. V06 N89 . R1—LAT	•	LMK COORDINATES FOR . USE ONLY BY THE AUT	
	DO 1 ONC /2	•	OMATIC OPTICS POSIT	
	R3-ALT	•	IONING ROUTINE (R52) .	
	• 63 821	:	AT THIS TIME.	#340
	. R1-LAT IS LATI-	•		
	. TUDE OF L#K IN	•	•	
	. DEG TO NEAREST	•	•	
	001 DEGREE, +	•	•	
	. IS NORTH	•		
	•	•	DO I WISH TO NAVI	
	• • • • • • • • • • • • • • • • • • •	•	GATE ON AN UNKNOWN .	
	 R2-LONG/2 IS LONGITUDE OF 	•	LANDMARK?	
	. LMK DIVIDED BY	•	YN	#350
	. 2. IN DEGREES	•	• • • • • • • • • • • • • • • • • • • •	
	. TO NEAREST .001	•		
	. DEGREE, + IS	•		
	. EAST	•	. DO I DESTRE THE .	
	•	•	. USE OF THE AUTO	
	. R3-ALT IS ALTI-	•	. MATIC OPTICS .	
,	. TUDE OF LMK	•	. POSITIONING ROU	
	 ABOVE THE FISCH- ER ELLIPSOID 	•	. TINE (R52) TO AID IN ACQUISITION? .	
	. IN EARTH ORBIT	•	• 14 ACQUISTITUAL •	#360
	. ABOVE THE MEAN	•	N .Y .	#300
	. LUNAR RADIUS FOR	•		
,	. LUNAR ORBIT	•	• •	
	. IN NAUTICAL	•		
	. MILES TO THE	•	SET OPTICS	
	. NEAREST .O1 NM.	•	MODE SWITCH	
ı	•	•	TO MANUAL	
	• •	•		
	•	•	• •	4776
		•	• •	#370 P22/COLOSSUS
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. WAIT FOR KEY-AM I SATIS-. BOARD ENTRY FIED WITH THESE VALUES? . • Y #380 KEY IN PROCEED . KEY IN V25E AND LOAD LMK PARAMETERS: . . TERMINATE FLASH #390 UPON RECEIPT OF PROCEED OR . NEW DATA ۰E •R #400 .STORE . NE W .DATA **#410** SET SIGHTING FLAG TO LMK FOR USE BY SIGHTING MARK AND AUTO OPTICS POSITION-ING ROUTINES. #420 P22/COLOSSUS

106

P22/COLOSSUS #430 #440 #450 #460 #470

SET MARK INDEX TO 5 FOR USE BY SIGHTING MARK ROUTINE (R53) DO AUTO DPTICS DO AUTO OPTICS POSITIONING **POSITIONING** ROUTINE (R52). ROUTINE (R52). INCLUDES SIGHT-INCLUDES SIGH-ING MARK ROUTINE TING MARK ROU-(R53). TINE (R53). FLASH VERB-NOUN MONITOR DSKY: TO REQUEST RESPONSE OBSERVE FLASHING AND DISPLAY LMK DATA VERB-NOUN TO REQUEST SNAP . . V05 N71 RESPONSE AND CISPLAY . R1-BL ANK LMK DATA R2-ABCDE R3-BLANK A=1 IF KNOWN LMK A=2 IF UNKNOWN LMK. B= INDEX OF OFFSET DID I MARK AN OFFSET DESIGNATOR LANDING SITE AND SO C = NOT USED IN THIS INDICATE BY KEYING PROGRAM. IN V52E FOLLOWING DE=LMK ID NO N THE MARK? NOTE: THE PROBABLE **CPTIONS ARE:** A=1, DE=00-LANDMARK KNOWN BUT NCT STORED IN CMC MEMORY A=1. DE>00-LANDMARK KNOWN AND STORED MAKE SURE B COR-A=2, DE NOT RESPONDS TO THE **MEANINGFUL:** MARK MADE ON THE UNKNOWN OFFSET LANDING LANDMARK. SITE OR IS SET TO

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ZERO.
                                           MAKE SURE B RE-
                                           MAINS UNCHANGED
                                            IF A OR DE ARE
                                           CHANGED.
                                           IS THE DATA IN R2
                                           CORRECT FOR THIS
                                           SIGHTING?
                                           NOTE: IN EARTH ORBIT .
                                           DE MUST BE SET EQUAL .
                                           TO OO BECAUSE LAND-
                                           MARK CODE 01 IS
                                           AVAILABLE FOR LUNAR
                                           LANDMARKS ONLY.
WAIT FOR KEYBOARD
                                                 KEY IN V22E AND .
ENTRY
                                                 LOAD CORRECT
                                                 DATA.
TERMINATE FLASH UPON
RECEIPT OF PROCEED
                                           KEY IN PROCEED
OR NEW DATA
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POSS .				VERB-NOUN	•	MONITOR D			•	
HOLD				UEST RESP-		OBSERVE V			•	
• • • • • • •				ND DISPLAY		FLASH TO			•	
SNAP	•	FWK	CO	ORDINATES	•	RESPONSE			•	
•	•	V 06	N8	9	•	LMK COORD	INATE:	S	•	
•	•		1-L		•				•	
•	•			ONG/2	•	•			•	
•	•	R:	3 A	LT	•	•			•	#590
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•				IS LATI-	•	•			•	
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•	•	LON	GIT	UDE OF LMK	•	•			•	
•				D BY 2.	•	•			•	#600
•		IN I	DEG	REES TO	•	•			•	
•	•	NEA	RES	T .001	•	•			•	
•		DEGI	REE	+ IS EAST	•	•			•	
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•	•	R 3=	ALT	IS ALTI-	•	•			•	
		TUDI	E 0	F LMK	•	•			•	
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		FOR	EA	RTH AND	•		,		•	
•		MEA!	N L	UNAR RADI-	•	•			•	#610
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•	TERMINATE E			VEN IN MASS AND	- •			
•	 TERMINATE F UPON RECEIP 	T OF .	•	KEY IN V25E AND LOAD LMK PAPAMETERS:	• • •			
•	• • PROCEED OR ! • • DATA	NEW .	•		- •			
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	W	·Ĉ.		•	•			#640
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•	. STORE NEW							
	. DATA	• •						
•	•							#650
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	•							# 660
								9000
	SET THE DATA BIT FOR INFO TO DOWNLINK	ORMATION						
	•							
	•							
	SET K = 0							#670
	•							
	•							
	SET J = 0							
	•							
	•							#680
++	•							+000
+11	•							
PCN 580	•							
J 00	•							

• SET J = J+1. IS J GREATER THAN THE MARK COUNTER? . GET JTH MARK DATA . FROM CMC STORAGE IS J = INDEX OF OFF-SET DESIGNATOR? STORE MARK IN OFFSET . DESIGNATOR . LCCATION IS THIS THE FIRST MARK?

#730

#720

#690

#700

#710

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+ 113 .	•		•				
• •	•	• •	•			P22/COLOSSUS	
•	•	• •	•			F22/C0E03303	
*		• •	•				
•			•				
+			•				
+ .	IS THE KNOWN I	FLAG .	•				#740
+11 .	RESET?	•	•				
++ .	• Y		•				
EDIT .	• T	.N .	•				
	•		•				
•			•				
•	COMPUTE AND		•				
•	STORE LMK	• •	•				
•	LOCATION FROM	• •	•				#750
•	. FIRST MARK . Data	•	•				**>0
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•	•		•				
** .	•		•				
+11 .	•	• •	•				
++ .	•	• •	•				
EDIT .	•	• •	•				
•	COMPUTE ORBIT	PARA-	•				
	METERS AND LA		•				#760
	COORDINATE CH		•				
•	AS A RESULT O		•				
•	SIGHTING MARK		•				
•			•				
•	•		•				
			•			•	
			•				
•	IS K = 0?		•				
•			•				#770
•	• • •	• N	•				
•	•	•	•				
POSS	· •	:	•				
HOLD .	FLASH VERB NO	UN .	•	MONITOR DS	Y: 08-		
• • • • • • • •	TO REQUEST RE			SERVE VERB	NOUN		
SNAP			•	FLASH TO RE			
•	PLAY ORBIT PA		•	RESPONSE AN			
•		•	•	OF ORBIT PA	KAMETER		#780
•	VALUAGO	•	•	CHANGES			#.JU
	OI DELTA D	•	•				
	R2-DELTA V			•			
•	R3-BLANK	•	•	•			
•	1	•	•				
	DELTA R-MAGNI	- •	•	ARE ORBIT			
	TUDE OF THE	-	•	CHANGES ACC FOR INSERT			
	, DIFFERENCE BE- . TWEEN THE POS		•	CMC CALCULA			
	TION VECTOR	- •	•	POSITION AN			#790
	BEFORE AND AF	TER .	•	CITY?			-
	INCORPORATION		•				
•	THE LANDMARK	•	•	• Y	. N	0001001 00010	
•	•	•	•	•	•	P22/COLOSSUS	
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	•	•	•	•	•		
•	SIGHTING DATA IN	•	•	•	•		
	N.M. TO NEAREST	•	•	•	•		
•	0.1 N.M.	•	•	•	•		
•	DELTA V-MAGNI-	•	•	•	•		
	TUDE OF THE	•	•	•	•	_	
	DIFFERENCE BE- TWEEN THE VELO-	•	•	•	•	#	800
	CITY VECTOR BE-	•	•	•	•		
	FORE AND AFTER	•	•	•	•		
	INCORPORATION OF THE LANCMARK	•	•	•	•		
	SIGHTING DATA IN	•	•	•	•		
	. FPS TO THE NEAR- . EST .1 FPS.	•	•	•	•		
	5 531 41 FF34	•	•	•	•		
	•	•	•	•	•	#	810
•	•	•	•	•	•		
		•	•		•		
	WAIT FOR KEY-	•	• • • • • • • • • • • • • • • • • • • •	KEY IN PROCEED	•		
	BOARD ENTRY	•	• •		•		
,	•	•	•		•		
	•	•	•		•		
	TERMINATE FLASH	•	• • • • • • • • • • • • • • • • • • • •	KEY IN RI	ECYCLE		820
	. UPON RECEIPT OF	•	• •	V32E			
	PROCEED OR RE-	•	•				
		•	•		•		
	R .P E .R	•	•		•		•
		•	•	•	•		
	Y .C	•	•		50 TO		
•	C .E L .E	•	•		MAM ABOVE	*	830
	. E .C	•	•			,	
	• •	•	•				
	• • •	•	•				
	. 60 TO .	•	•				
	. MAT . ABOVE .	•	•				
,	•	•	•				
	•	•	•				840
•	SET K = 1	•	•			•	1070
		•	•				
•	•	•	•				
		•	•				
	•	•	•				
•	•	•	•				
	•	•	•			P22/COLOSSUS	
	•	•	•				
•	•	•	•				
	1			-	`	\	

UPDATE CMC STATE
 VECTOR AND COMPUTE
 REVISED LANDMARK

IS THERE DATA IN THE OFFSET DESIGNATOR

MAP THE OFFSET

LANDING SITE

COORDINATES

LECATION?

P22/COLOSSUS #850 #860 #870 #880 #890

MONITOR DSKY:
OBSERVE VERB-

DID I DESIGNATE AN OFFSET LANDING SITE FROM A PRIMARY LAND-MARK BY USE OF V52E

FOLLOWING A MARK?

OBSERVE VERB-NOUN FLASH TO REQUEST RES-PONSE AND DIS-

PLAY OF THE LANDMARK COOR-DINATES ON

WHICH I WAS MARKING.

· MARKIN

POSS HOLD .

VO6 N89 R1-LAT R2-LCNG/2 R3-ALT

FLASH VERB-NOUN TO

DISPLAY POTENTIAL

NATES

REQUEST RESPONSE AND

LANDING SITE COORDI-

R1=LAT IS LATITUDE OF LMK IN DEG TO NEAREST .001 CEGREE + IS NORTH

#900

n	2	2	"	- 1	31	1	_	c	•	11	ıs
 _	•	•	,	- 1	.31	L. 3	u	. 3	-3	ы	

R2=LONG/2 IS LONGI- TUDE OF LMK DIVIDED BY 2. IN DEGREES TO NEAREST .001 DEGREE + IS EAST	•	MONITOR DSKY: OBSERVE VERB— NOUN FLASH TO REQUEST RESPONSE AND DISPLAY OF THE OFFSET LAND— ING SITE COORDI— NATES.
R3=ALT IS ALTITUDE OF LMK ABOVE THE FISCHER ELLIPSOID FOR EARTH AND MEAN LUNAR RADIUS FOR MOON. IN NAUTICAL MILES TO THE NEAREST .01 NM. WAIT FOR KEY- BOARD ENTRY		
		AVAILABLE FOR LUNAR LANDMARKS. FOR THIS REASON THE ASTRONAUT

#910

#920

#930

#940

#950

		EITHER ON THI: WHEN I	ALWAYS KEY V32E OR V34E S DISPLAY N EARTH ORBIT.		,
		.YES			,
	•	KEY IN	PROCEED .		
	•	•	:		
TERMINATE FLASH UPON RECEIPT OF PROCEED OR RECYCLE	•	. V	EY IN RECYCLE 32E		1
•R •P		•	•		
•E •R •C •C •Y •C		•	•		
.C .E		GO TO	GO TO		1
.E .D		ABOVE	ABOVE		
GO TO					
A					
ABOVE .					4
STORE THESE COORDIN- ATES AS UPDATED ANDING SITE PARAMETERS.					
•					
•					#:
GC TO "A" ABOVE					-

V34E

#1010

THE PROGRAM WILL
COME TO THIS POINT
FROM A V34E ON ANY
FLASHING DISPLAY.

DO ROUTINE (ROO)

DO ROUTINE (ROO)

EXIT

P22

#1020

#1030

;

CHANGE CENTROL NOTES

EXIT

P22

LOGIC REV 07 PCR MIT 66 LOGIC REV 08 PCR MIT 116 PCR MIT 83 LOGIC REV 09 PCR 206 LOGIC REV 10 PCN 552 LOGIC REV 11 PCN 580 PCN 594

'18

PURPOSE:

++

EDIT

CONT

CISLUNAR MIDCOURSE NAVIGATION PROGRAM (P23)

LCGIC REV 14 11/27/68

(1) TO DO MIDCOURSE NAVIGATION BY INCORPORATION OF STAR/EARTH AND STAR/MOON OPTICAL MEASUREMENTS.

ASSUMPTIONS: (1) THIS PROGRAM DOES NOT REQUIRE THAT THE IMU BE ON.

(2) IF THE IMU IS NOT ALIGNED THE ASTRONAUT MUST ACQUIRE THE STAR/LMK OR STAR/HOR MANUALLY.

(3) (A) IF THE IMU IS ALIGNED THE ASTRONAUT MAY ACQUIRE THE LMK/HOR AUTOMATICALLY.

(B) IF THE IMU IS ALIGNED THE ASTRONAUT MAY ACQUIRE THE STAR AUTOMATICALLY.

(C) IF THE IMU IS ON THE ASTRONAUT MUST TAKE APPROPRIATE PRECAUTIONS TO PREVENT POSSIBLE IMU GIMBAL LOCK.

(4) PRIOR TO EACH MARK THE PROGRAM WILL CALL FOR AN OPTICS CALIBRATION WHICH MAY BE DONE OR BYPASSED DEPENDENT UPON THE STABILITY HISTORY OF THE CALIBRATION (SEE PURPOSES AND ASSUMPTIONS OF R57).

(5) TO PERFORM THE MARK THE ASTRONAUT SHOULD FINALLY SELECT MINIMUM IMPULSE CONTROL (EITHER GNCS OR SCS) AND THE OPTICS SHOULD BE IN MANUAL IN ORDER TO MAINTAIN THE FIX.

(6) THE OPTICS SHOULD BE ON FOR 15 MINUTES PRIOR TO MARKING.

(7) THE CMC DOES NOT CHECK FOR MODN/EARTH OCCULTATION OR SUN BRIGHTNESS IN THIS PROGRAM

(8) THIS PROGRAM IS DESIGNED FOR ONE MAN OPERATION WITHIN THE CONSTRAINTS OF MODE SWITCHING WHILE IN THE LEB.

(9) THE SIGHTING IS ON THE BODY FOR WHICH THE STATE VECTOR IS DEFINED.

+14 ++ (10) A STAR SERIAL O WILL GIVE PARITY FAILURE, AND LANDMARK SERIAL O1 WILL NOT WORK.

(11) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG CMC GROUND

CREW

CHECKLIST

TIME

TOTAL TIME

.CREW PROG. SELECTION

```
START CISLUNAR MID-
                                          KEY IN CISLUNAR MID-
COURSE NAVIGATION
                                         COURSE NAVIGATION
PROGRAM (P23)
                                          PROGRAM (P23)
                                                                                                                    #10
DISPLAY P23
                                               V37E 23E
                                          MONITOR DSKY:
                                          OBSERVE DISPLAY OF
                                          PROGRAM 23
                                                                                                                    #20
SET MARK INDEX TO 1
FOR USE BY THE
SIGHTING MARK
ROUTINE R53
                                                                                                                    #30
RESET RENDEZVOUS FLAG
SET TARGET FLAG TO
                                                                                                                    #40
STAR FOR USE BY THE
SIGHTING MARK ROU-
TIME (R53) AND BY
THE AUTO OPTICS
POSITIONING ROUTINE
(R52)
                                                                                                                    #50
RESET V94 FLAG
                                                                                                     P23/COLOSSUS
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++ +13 ++

EDIT PCR 206

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	•					
•	•				P23/COLOSSUS	
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•		- -	н Ви			#60
•	RESET R57		•			•
•	FLAG		•			
•			•			
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	•		• • •			
•	•		•			
		- -				
IS RE	FSMMAT FLAG		IS THE IMU ON AN	D		
SET?			AL I GNED?			#70
• Y	• N		.N	. Y		
•	•		•	•		
•	•		•			
•	•		•	•		
•	•		•			
•	•			•		
•	•		MANUALLY MAN-			
•	•		EUVER THE	•		
•	•		VEHICLE UNTIL A			#80
•	•		SUITABLE LMK/			
•	•		HOR IS IN THE			
•	•		SXT FIELD OF	·		
	•		VIEW FOR USE BY	-		
	•		THE OPTICS CAL-	•		
_	•		IBRATION ROUT-	-		
_	•		INE, R57	-		
-	_		NOTE: IN GROER	•		
_	· ·		TO CONSERVE	•		
-	_		FUEL THIS ATT-	•		#90
_			ITUDE SHOULD BE	•		# 7U
_	× .		CLOSE TO THE	•		
-	•		NAVIGATION LMK/	•		
-	•		HOR	•		
-	•			•	•	
-	•		_	•		
-	4		-	•		
	•			•		
				•		
. DO	OPTICS		DO OPTICS	•		100
- CA	LIBRATION		CALIBRATION	-	•	100
	UTINE, R57		ROUTINE, RST	•		
		•		•		
-			_	•		
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	•		-	-	P23/COLOSSUS	
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	• •		•)	P23/COLOSSU\$	
	• •		•	ı		
	•		•	•		
				•		
	. DO SIGHTING	• • • • • • • • • • • • • • • • • • • •	DO THE SIGHTING .	•		
	. MARK ROUTINE,	•	MARK ROUTINE,	•	#1	10
	• (R53) •		R53. MANUALLY MAN-		#1	. 10
	•		EUVER THE			
	• •		VEHICLE UNTIL A	•		
	• •		SUITABLE LMK/	•		
	• •••		HOR IS IN THE SXT FIELD OF	•		
	. GO TO		VIEW. THEN,			
	. ндн		BY USE OF MIN-	,		
	. BELOW		IMUM IMPULSE	•		
	•		CONTROL (GNCS or SCS) AC-	.	71	20
	•		QUIRE THE LMK			
	•		AND BY USE OF	•		
	•		OPTICS CONTROL-	•		
	•		LER SUPERIMPOSE	•		
	•		THE NAVIGATION	•		
	:		LIST.	•		
	•			•		
	•		•	•	#1	30
	•		•			
	•		• • •	•		
	•		•	•		
	•		GO TO	•		
	•		#A" BELOW	•		
	•		DELUM	•		
	•			•		
	•			•	#1	40
	•		•	•		
	•	•		, 		
	•	•				
	•	•	•			
	•	•	•	• •		
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POSS		•		· · ·		
HOLD .	FLASH VERB-NOUN TO	•	MONITOR DSKY:	•	#1	l 50
	REQUEST RESPONSE AND		CBSERVE VERB-NOU	٠.		
SNAP .	DISPLAY MEASUREMENT IDENTIFICATION	•	RESPONSE AND DISPI	I AY		
	V05 N70		OF MEASUREMENT	•		
	R1-000DE (STAR ID)		IDENT IF ICATION	•		
	R2-ABCDE (LMK ID)		***********	•		
	R3-00CD0 (HOR ID)	•	•	•		
		•	•	•		
		•	•	•		
		•	•	•	B33 feor beene	
		•	•	•	P23/COLOSSUS	
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•	•	P23/COLOSSUS
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	<u> </u>	
•	I I	
STAR IDENTIFICATION	•	
	•	***
R1=000XX FOR ID .		#160
NUM. OF STAR .	DO I INTEND TO DO .	
USED .	THE SIGHTING .	
•	MANUALLY? .	
LANDMARK IDENTIFI	THIS DATA WILL BE .	
CATION .	USED AT THIS TIME .	
R2=ABCDE WHERE	FOR THE AUTOMATIC .	
RZ-ADCDE WHERE		
•	TRACKING MANEUVER .	
A AND B-NOT .	AND BY THE AUTO	
USED .	MATIC OPTICS .	
•	POSITIONING ROUTINE .	#170
C=1 FOR EARTH .	(R52) ONLY. IF IT IS .	
LMK .	INTENDED TO DO MAN-	
21.11	UAL ACQUISITION THIS .	
6-2 FOR MORN		
C=2 FOR MOON .	DISPLAY NEED NOT BE .	
ŁMK .	REVIEWED. IT WILL BE .	
•	REDISPLAYED PRIOR TO .	
DE-LANDMARK ID .	STATE VECTOR .	
NUMBER .	CALCULATION	·
113-11-21		
R2=00000 FOR .	. Y . N	2100
	•Y •N •	#180
HOR. MEAS	• •	
•	• •	
HORIZON IDENTIFI	•	
CATION .	SET OPTICS MODE	
R3≈00CDO WHERE .	SWITCH TO MANUAL	
•		
C=1 FOR EARTH .		
HOR		
C=2 FOR MOON .	• • •	#190
HOR.	• •	#170
nur• •	• •	
•	• •	
D=1 FOR NEAR .	•	
HORIZON .	.AM I SATISFIED .	
D=2 FOR FAR .	.WITH THIS DATA? .	
HORIZON .		
	• •Y •N •	
R3=00000 FOR .		
I MV MEAC	• • •	
FWW WEAP.	• • • •	
	•	#200
•	.RECORD MEAS	
•	.UREMENT IDEN	
•	.TIFICATION	
•	.DATA IN	
	•FLIGHT DATA	
	.BOOK IF	
	.DESIRED	
•	• • •	
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•	• • •	#210
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	•	·	: :				#27	70
POSS							, 	•
HOLD .		LASH VERB-N		•	MONITOR DSK		•	
• • • • • •		O REQUEST P		• • • • • • • • • • • • • • • • • • • •	OBSERVE		•	
SNAP .		SPENSE AND E		•	NOUN FLASH			
	• •	PLAY LMK DAT VO6 N89	Α .		REQUEST RES			
	•	100 1109	:		LMK DATA IF			
	•	R1: LAT	•		NON-IDENTIF		•	
	•	R2: LONG/	2 .		LANDMARK WA		•	
	•	R3: ALT	•		CHOSEN		#28	10
	• .		•					
	٠ ١	.AT:-LATITUD DF L⊮K IN DE	e e		•			
		O NEAREST			•			
		DEG						
	•	+ IS NORT	Ή .		AM I SATISF	IEO		
	•		•		WITH THIS D	YA7		
		.ONG/2-L DNG 1						
		UDE OF LMK			•Y	•N		
		IDED BY 2 I			•	•	#29	0
	• L	DEG TO NEARE .001 Deg	:51 •		•	•		
	• •	+ IS EAST	•		•	•		
	•	. 15 1.5	:		•			
		LT-ALT OF L	MK .		•	•		
		N NAUTICAL	•		•	•		
		ILES TO NEA			•	•		
		ST -01 N.M.	•		•	•		
		FOR EARTH BOVE FISCHE	D		•	•	#30	
		LLIPSOID) t			•	•	#3U	v
		IOON ABOVE M			•	•		
		UNAR RADIUS			•	•		
	• -				•	•		
	•	•	•		•	• •		
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	: -	·	:			- :		
	. H	AIT FER KEY	-	•	KEY IN	•		
	. 8	DARD ENTRY		• • • • • • • • • • • • • • • • • • • •	PROCEED	•	#31	.0
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	•		•				P23/COLOSSUS	

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	TERMINATE FLASH . UPON RECEIPT OF . PROCEED OR NEW .	 KEY IN . V25E . AND .	
	. DATA	LOAD .	#320
	• R •DATA •	DATA .	,
	• • • • • • • • • • • • • • • • • • •	• • • • • • • • • • • • • • • • • • •	
	• • E DATA • • • • • • • • • • • • • • • • • •	•••••	
	• • • •		
	• •		#330
	• •		
POSS HOLD . SNAP .	FLASH VERB-NOUN TO REQUEST PLEASE PERFORM GNCS AUTO- MATIC MANEUVER		
	V50 N25 R1-00202	TATIC MANEUVER	#340
	R 2-BLANK R 3-BLANK	<u>:</u>	
	*	•	•
	•	• •	•
	•	•	****
	•	DO I WISH TO HAVE THE GNCS COMPUTE THE	#350
	:	SPACECRAFT ATTITUDE REQUIRED TO POINT	
	•	THE LLOS AT THE CHOSEN LMK/HOR AND	
	•	PERFORM THE ATTITUDE MANEUVER ROUTINE	
	•	(R60)? THIS ATTITUDE WILL	#360
	•	BE COMPUTED (VEC- POINT) TO POINT THE LLOS AXIS AT THE	
	•	LMK/HOR BUT WILL NOT CONSTRAIN THE ORIEN-	
	•	TATION ABOUT THAT VECTOR (THE DRIFN-	
	•	TATION ABOUT THAT VECTOR COULD RESULT	
	•		P23/COLOSSUS
126	•		
	`	`	

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+13 ++ 520+ EDIT

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•			IN IMU G	IMBAL LOCK).
			.N	.Y
WAIT FOR KEYBOAR	D		•	•
ENTRY			•	•
				•
		•	KEY IN	•
•			ENTER	•
		•	•	•
			•	•
			•	•
TERMINATE FLASH		•	• -	
RECEIPT OF PROCE OR ENTER	ΕU	•••••		EY IN ROCEED
)K ENIER		•		KUCEEU
•E	. P			
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. INTEGRATE CS	M		MANUALLY	•
. STATE VECTOR	TO		MANEUVER	THE .
. THE PRESENT			VEHICLE	•
. TIME.			UNTIL A Suitable	1 44
•			/HOR IS	
•	•		THE SXT	•
•	•		FIELD OF	•
CALCULATE TO			VIEW FOR	USE .
CALCULATE TH ATTITUDE TO	Ė		BY THE OPTICS C	A1 -
. POINT THE			IBRATION	ML- •
. LLOS AXIS AT	THE		ROUTINE,	R57 .
. LMK OR HORTZI	DN		NOTE: IN	•
. USING THE PO			ORDER TO	•
 ING VECTOR RI TINE, REFER 			CONSERVE	•
• SECTION 5.2.			FUFL THIS ATTITUDE	•
• DE R577 •			SHOULD BI	•
			CLOSE TO	
•			NAVIGATI	
•			LMK/HOR	
• •				
			•	

• •		• • •	P23/COLOSSUS	
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•		• • •	44	430
•		• • •	*	730
•		• • •		
•		• • •		
•				
 DO THE ATTITUDE MANEUVER ROUTINE 		. DO THE . ATTITUDE		
. (R60)	********	 MANEUVER 		
•	• •	. ROUTINE (R60)	#4	440
: :		• • • •		
•		•		
•		•		
IS R57 FLAG SET?		• •		
•		•		
• •N •Y		• •		
		•	#4	450
DC R57 OPTICS . CALIBRATION .	• •	DO R57 OPTICS CALI- BRATION ROUTINE		
ROUTINE	• •	(WHEN LMK ACQUIRED)		
•		•		•
•		•		
		•		
SET V94 FLAG		•	#4	460
		•		
•		•		
•		•		
SET R57 FLAG		•		
		•		
•		•		
DO THE AUTOMATIC OPTICS POSITIONING		DO R52 AUTOMATIC OPTICS POSITIONING		470
ROUTINE (R52) (CALLS R53)	• •	ROUTINE. THE CPTICS POINTING PROCESS IS		
.NORMAL .PREMATURE		NOW AUTOMATIC AND WILL POINT THE SXT		
.EXIT .EXIT VIA		STAR LOS AT THE		
• • • • • • • • • • • • • • • • • • •		CHOSEN STAR UNTIL		
. DO V94 LOGIC (AS		THE SIGHTING MARK ROUTINE (R53) IS		
•			P23/COLOSSUS	
•		e e		

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))	<i>)</i>	
119	+	•				
•	+	•				
	+	•			P23/COLOSSUS	
	*	•				
	*	•				
	++	• DEFINED IN EXTEN-	CALLED	AUTCHATICALLY	,,,,	_
	EDIT	• DED VERB SECTION		AUTEMATICALLY CHING OPTICS	#480	J
	PCR	• OF THIS DOCUMENT)		MANUAL.		
	206	•		ASTRONAUT DE-		
		•		O HAVE THE		
		• •		IN REACQUIR-		
		• •	ING THE	LANDMARK/		
		• •••		I HE MAY KEY		
		• •		UNTIL HE HAS		
				ED THE SIGHT-		_
		• 60 TO		RK ROUTINE	#490)
		• ABOVE	(R53).	TO ACTUAL MARK		
				RONAUT SHOULD		
		•		A MINIMUM		
				MODE (EITHER		
		RESET V94 FLAG		SCS) IN	·	
				O MAINTAIN		
		•		CHOR AND STAR		
		•		N THE SXT		
		•	FIELD O	OF VIEW.	#500)
		RESET R57 FLAG				
				•		
		•		•		
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		ндн .	MAM	•		
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		• • • •	•	• •	•	
	HOLD .	FLASH VERB-NOUN TO .	MONITOR	UCAA.	450	
	******	REQUEST RESPONSE AND .		E VERB-NOUN	* #520	,
	SNAP .	DISPLAY MEASUREMENT .		O REQUEST	•	
		IDENTIFICATION		E AND DISPLAY	•	
		V05 N71 .	OF MEAS		•	
		RI-OGODE (STAR IC) .	IDENT LF	ICATION	•	
		R2-ABCCE (LMK ID) .			•	
		R3-00CD0 (HOR ID) .		•	•	
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				P23/COLOSSUS	
	•	•		1 23700003300	
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	•	•		•	
STAR IDENTIFICATION-	•	•		•	
R1=000XX FOR ID	•	•		•	#530
NUM. OF STAR	•	•		•	
USED	•			•	
LANDMARK IDENTIFI-	•	IS THIS DATA COR FOR THE MARK I M			
CATION R2=ABCDE WHERE	•	.Y	•N	•	
	•	•	•	•	
A AND B-NOT	•	•	•		
USED	•	•	•		F540
	•	•	•	• -	1940
C=1 FOR EARTH	•	•	•		
LMK	_	-			
C=2 FOR MOON	•	•	•		
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D=2 FOR FAR	•	DECODO NE ACUAL	•	•	
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HOLD. SNAP.

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			#690
SUBTRACT CALIBRATION ANGLE (STORED BY R57) FROM MEASURED TRUNNION ANGLE.			
RESET RENDWFLAG SEE (P20).			#700
COMPUTE STATE VECTOR CHANGE DUE TO THE MEASUREMENT			#7 10
FLASH VERB-NOUN TO REQUEST RESPONSE AND DISPLAY ORBITAL PARAMETER CHANGES VO6 N49 R1-DELTA R R2-DELTA V R3-BLANK	•	MONITOR DSKY: OBSERVE VERB-NOUN FLASH TO REQUEST RESPONSE AND CISPLAY OF ORBITAL PARAMETER CHANGES	#720
DELTA R-MAGNITUDE OF THE DIFFERENCE BETWEEN THE POSITION VECTOR BEFORE AND AFTER INCCRPORATION OF STAR MEASUREMENT IN N. M. TO NEAREST O.1 N. M.		• • • • • • • • • • • • • • • • •	#730

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CHANGE CONTROL NOTES

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REV 10 PCR NASA 84 **REV 11** PCR MIT 66 PEV 12 PCR 206 **REV 13** PCN 520 **REV 14** PCN 571 PCR 206 EDITORIAL

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EXIT

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CMC HIPPATE PROGRAM (P27)

LOGIC PEV 07 11/24/48

PURPOSE: (1) TO INSERT INFORMATION INTO THE CMC VIA THE DIGITAL UPLINK BY TRANSMISSION FROM THE GROUND OR VIA THE DSKY KEYROARD PY CREW MANUAL INPUT.

ASSUMPTIONS: (1) THE CMC MUST BE IN THE OPERATE CONDITION. THE IMU MAY BE IN STANDBY OR OPERATE CONDITION.

(2) CMC UPDATES APE OF FOUR CATEGORIES:

(A) PROVIDE AN UPDATE FOR CHC LIFTOFF TIME (V70).

IB) PROVIDE AN OCTAL INCREMENT FOR THE CMC CLOCK ONLY (V73).

(C) PROVIDE LOAD CAPABILITY FOR A BLOCK OF SEQUENTIAL FRASABLE LOCATIONS (1-18 INCLUSIVE LOCATIONS WHOSE ADDRESS IS SPECIFIED) (V71).

(D) PROVIDE LOAD CAPABILITY FOR 1-9 INCLUSIVE INDIVIDUALLY SPECIFIED ERASABLE LOCATIONS (V72).

(3) A COMPLETE DESCRIPTION OF THE CMC UPLINK FORMAT IS INCLUDED IN SECTION 2 OF R-577.

(4) UPDATE IS ALLOWED IN THE CSM WHEN THE CMC IS IN POO OR PO2, AND IF THE DSKY IS AVAILABLE.

(5) THE UPTEL ACCEPT/BLCCK SWITCH MUST BE IN ACCEPT FOR TELEMETRY UPDATE.

(6) THE PROGRAM IS MANUALLY SELECTED BY THE ASTRONAUT BY DSKY ENTRY OR BY THE GROUND BY UPLINK TRANSMISSION.

(7) THE AUTOMATIC MODE OF UPDATE IS PROGRAM SELECTION AND UPDATE VIA THE GROUND BY UPLINK TRANSMISSION. THE ONLY DIFFERENCE BETWEEN THIS AND MANUAL SELECTION BY THE ASTRONAUT IS THAT THE DSKY RESPONSES ARE KEYED IN BY THE ASTRONAUT RATHER THAN TRANSMITTED.

PROG CMC GROUND CPEW CHECKLIST TIME TOTAL CONT

NOTIFY CREW OF VERIFY THAT UPDATE INTENTION TO . WILL BE STAISFACTORY UPDATE. SPEC-...W.P.T. TIME, WORK IN TEY PARAMETERS . PROGRESS AND CMC ACTIVITY. TIME.

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P27/COLOSSUS P27/SUNDANCE P27/LUMTNARY

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		SELECT THE DE- SIRED TYPE DE UPDATC: V70-LICTOFC TIME INCREMENT V71-CONTIGUOUS	•				
		RETOR UPDATE V72-SCATTER UPDATE V73-DCTAL TIME INCREMENT	:				∉20
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	INTO MACHINE ADDRESS SPECIFIED IN P3 AND			#320
HOLD	• DISPLAY:	OF OCTAL		
SNAP	••• V21 NO2 • P1 BLANK	IDENTIFIER		
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TION COMPLETED .		
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TIME UPDATE		#510
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CLOCK TO OVER-		# 520
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		. DOWNLINK LIST,	P00 OR P02.	. #620
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CHANGE CONTROL MOTES

REV OF PCP MIT 66 REV OF PCP 206 PEV OF EDITORIAL THIS PAGE INTENTIONALLY LEFT BLANK