

CMC IDLING PROGRAM (P 00)

LOGIC REV 05 11/26/68

PURPOSE:

(1) TO PROVIDE A PROGRAM TO FULFILL THE FOLLOWING REQUIREMENTS:

(A) PROVIDE AN INDICATION TO THE CREW THAT THE CMC IS ENGAGED IN NO CONTROL OR COMPUTATIONAL OPERATIONS WHICH MIGHT REQUIRE CONSIDERATION FOR COORDINATION WITH OTHER CREW TASKS IN PROGRESS.

(B) TO MAINTAIN THE GNCS IN A CONDITION WHERE MANUAL ATTITUDE MANEUVERS CAN BE MADE BY THE CREW WITH MINIMAL CONCERN FOR THE GNCS (SFE ASSUMPTION 3).

(C) MAINTAIN THE CMC IN A CONDITION OF READINESS FOR ENTRY INTO OTHER PROGRAMS.

(2) TO UPDATE THE CSM AND LM STATE VECTORS EVERY FOUR TIME STEPS.

++

ASSUMPTIONS:

(1) THE IMU MAY OR MAY NOT BE ON. IF ON, THE IMU IS INERTIALLY STABILIZED BUT NOT NECESSARILY ALIGNED TO AN ORIENTATION WHICH IS KNOWN TO THE CMC.

++

EDIT

(2) IF NON-GNCS CONTROLLED ATTITUDE MANEUVERS ARE MADE BY THE CREW CARE MUST BE TAKEN TO AVOID IMU GIMBAL LOCK. THE IMU GIMBAL ANGLES MAY BE MONITORED BY OBSERVING THE ICDUS (V16 N20) OR BY MONITORING THE FPAI BALL.

(3) DURING THIS PROGRAM THE CMC ERASABLE STORAGE MAY BE INITIALIZED BY KEYING IN V36E (FRESH START). THIS WOULD BE DONE ONLY AT INITIAL CMC STARTUP OR WHEN THE CONTENT OF THE CMC ERASABLE STORAGE IS IN QUESTION. IF THIS ENTRY IS PERFORMED, THE CMC'S KNOWLEDGE OF THE PRESENT STATE VECTOR AND THE PRESENT IMU ORIENTATION (PEFSMMAT) IS INVALIDATED.

(4) THE PROGRAM IS MANUALLY SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PCR

507

++

+04 AND 05

++

EDIT

(5) THIS PROGRAM IS AUTOMATICALLY SELECTED BY V96E, WHICH MAY BE DONE DURING ANY PROGRAM. STATE VECTOR INTEGRATION IS PERMANENTLY INHIBITED FOLLOWING V96E. NORMAL INTEGRATION FUNCTIONS WILL RESUME AFTER SELECTION OF ANY PROGRAM OR EXTENDED VERB. P00 INTEGRATION WILL RESUME WHEN P00 IS RESELECTED. USAGE OF V96 CAN CAUSE INCORRECT W-MATRIX AND STATE VECTOR SYNCHRONIZATION.

PRG
CONT

CMC

GROUND

CREW

CHECKLIST

TIME

TOTAL
TIME• CMC PROG
• SELECTION

•

•

•

•

•

START CMC IDLING
PROGRAM 00
DISPLAY PROGRAM 00

•

•

• CREW PROG
• SELECTION

•

•

•

•

•

KEY IN CMC IDLING
PROGRAM (P00)
V37E 00E

•

•

•

•

•

•

#10

P00/COLOSSUS
P00/SUNDANCE
P00/LUMINARY

MONITOR DISK:
OBSERVE DISPLAY OF
PROGRAM 00

#20

IS QUIT FLAG SET?

.N .V

RESET
QUIT
FLAG

#30

AWAIT
ASTRO-
NAUT
ACTION

#40

EXTRAPOLATE PERMAN-
ENT STATE VECTORS
FORWARD TO PRESENT
TIME EVERY 4 TIME
STEPS. SEE SECTION
5. FIG 4.12-1

#50

IS THERE A REQUEST
FOR A NEW PROGRAM?

#60

POO/CLOSSUS
POO/SUNDANCE
POO/LUMINARY

```
.  
. .  
.  
.  
.  
.  
...  
.
```

EXIT R00
AND GO TO
PROGRAM
SELECTED
VIA R00

#79

#80

CHANGE CONTROL NOTES

REV 03
REV 04

DCR MIT 66
DCR 507

490

#100

THIS PAGE INTENTIONALLY LEFT BLANK

)

)

)

(2) ERASABLE LOCATIONS HAVE BEEN PROPERLY INITIALIZED.(AZIMUTH, +1; LATITUDE, +1; LAUNCHAZ, +1; IMU COMPENSATION PARAMETERS)

TOTAL
TIME

#10

#20

#30

++
+04
++

WAIT ABOUT 10
SECONDS

TURN ON "NO ATT"
LIGHT

OBSERVE "NO ATT"
LIGHT ON

#40

COMMAND COARSE ALIGN
IN ISS.
COARSE ALIGN TO
DESIRED PLATFORM
ORIENTATION

#50

REMOVE COARSE ALIGN
COMMAND (RELEASE
PLATFORM).

#60

TURN OFF "NO ATT"
LIGHT

OBSERVE "NO ATT"
LIGHT OFF

#70

TERMINATE PROGRAM 01
AND GO TO PRELAUNCH
OR SERVICE - GYRO
COMPASSING PROGRAM
(P02)

MONITOR DSKY:
OBSERVE TERMINATION
OF P01 AND DISPLAY
OF P02

EXIT P01

EXIT P01

#80

41

CHANGE CONTROL NOTES

REV 04 PCR 206

00110000

THIS PAGE INTENTIONALLY LEFT BLANK

11/27/68

PURPOSE: (1) TO PROVIDE THE PROPER STABLE MEMBER ORIENTATION FOR LAUNCH.

ASSUMPTIONS: (1) THIS PROGRAM MAY BE INTERRUPTED TO PERFORM THE PRELAUNCH OR SERVICE - OPTICAL VERIFICATION OF GYROCOMPASSING PROGRAM (PO3).

(2) V75 WILL BE KEYED IN AND DISPLAYED DURING THIS PROGRAM TO PERMIT CREW BACKUP OF THE LIFTOFF DISCRETE.

(3) THE PROGRAM IS AUTOMATICALLY SELECTED BY THE INITIALIZATION PROGRAM (POI).

(4) THIS PROGRAM HAS THE CAPABILITY (VIA V78E) TO CHANGE LAUNCH AZIMUTH OF THE STABLE MEMBER WHILE GYROCOMPASSING.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

- AGC
- PROG
- SELECTION

START PRELAUNCH OR
SERVICE-GYRO
COMPASSING PROGRAM
{P02}
DISPLAY PROGRAM 02

```

      . MONITOR DSKY:
.....  . OBSERVE DISPLAY OF
      . PROGRAM 02

```

#10

SET TIMER

#20

DO VERTICAL
ERECTION

#30

WAIT 0.5 SECONDS

#40

HAS TIMER
REACHED 640
SECS?

.N .Y

#50

DO GYRO COMPASS-
ING

#60

WAIT 0.5 SECONDS

DO I WISH TO CHANGE
LAUNCH AZIMUTH OF
STABLE MEMBER

#70

.Y .N

++
+04
+
+
+
+
+04
++
EDIT

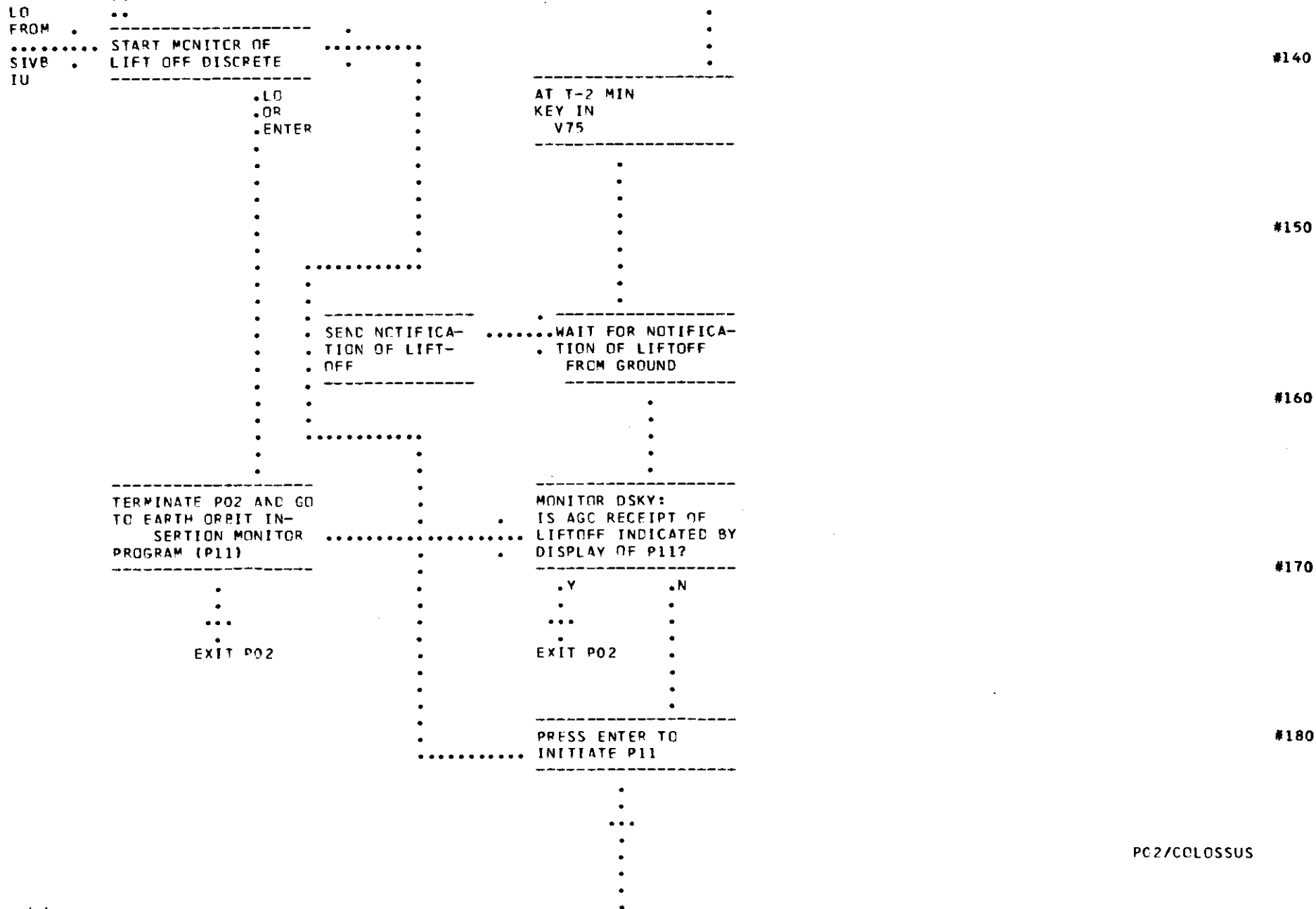
44

45

P02/COLOSSUS

HOLD	FLASH VERB-NOUN TO REQUEST PROCEED AND DISPLAY STORED LAUNCH AZIMUTH V06 N29 R1-XSM LAUNCH AZIMUTH R2-BLANK R3-BLANK	KEY IN V78E	#80
SNAP	XSM LAUNCH AZIMUTH MEASURED CLOCKWISE FROM TRUE NORTH IN DEGREES TO NEAREST .01 DEGREES	MONITOR DSKY: OBSERVE VERB-NOUN FLASH TO REQUEST PROCEED AND DISPLAY OF XSM LAUNCH AZIMUTH	#90
		AM I SATISFIED WITH XSM LAUNCH AZIMUTH?	#100
		.Y .N	
	WAIT FOR KEYBOARD ENTRY	KEY IN V21E AND LOAD NEW LAUNCH AZIMUTH	#110
	TERMINATE FLASH UPON RECEIPT OF PRO- CEED OR NEW DATA	KEY IN PROCEED	
	.P .NEW .R .DATA .O . .C . .E .E STORE .D NEW DATA		#120
			#130

P02/COLOSSUS



·
·
·
·
·
·
EXIT P02

P02/COLLOSSUS

CHANGE CONTROL NOTES

REV 04 EDITORIAL

THIS PAGE INTENTIONALLY LEFT BLANK

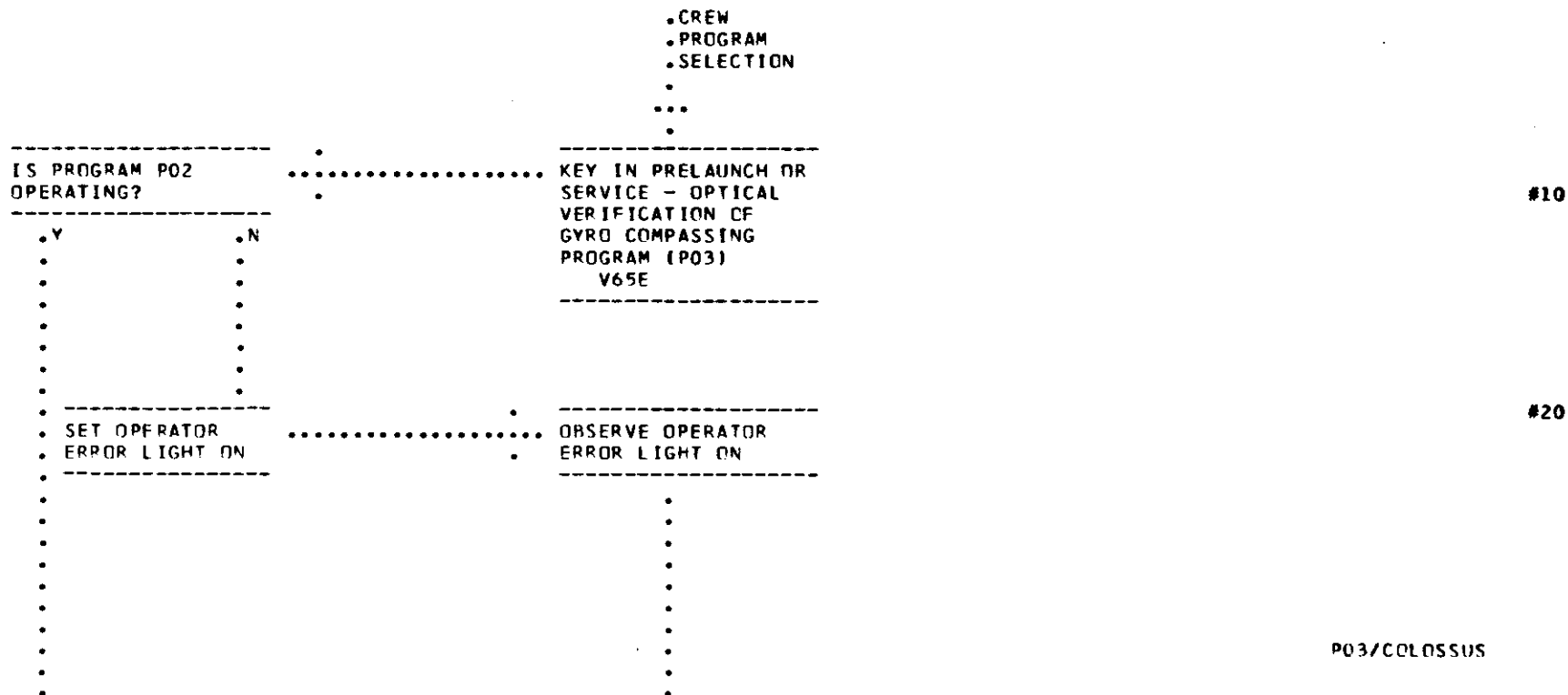
PRELAUNCH OR SERVICE - OPTICAL
VERIFICATION OF GYRO COMPASSING
(P03)

LOGIC REV 08 07/12/68

- PURPOSE: (1) TO PROVIDE AN OPTICAL CHECK FOR VERIFICATION OF ALIGNMENT OF THE STABLE MEMBER OF THE ISS DURING GYRO COMPASSING PRIOR TO LAUNCH.
- ASSUMPTIONS: (1) THE PROGRAM IS MANUALLY SELECTED BY DSKY ENTRY.
- (2) THE ASTRONAUT HAS ZEROED THE OPTICS JUST PRIOR TO PROGRAM (P03) SELECTION.
- (3) A MINIMUM OF 45 MINUTES BETWEEN V78E AND P03 (V65E) INSURES PROPER DAMPING OF TRANSIENTS.
- (4) IN ORDER TO PREMATURELY TERMINATE THIS PROGRAM AND RETURN TO P02 THE ASTRONAUT MAY KEY IN V34E ON ANY FLASHING DISPLAY

++
+OR
+OR
++

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------



HOLD .

 SNAP .

FLASH VERB-NOUN TO
 REQUEST PROCEED AND
 DISPLAY STORED
 TARGET 1 AZIMUTH AND
 ELEVATION:
 V06 N41
 R1-TARG AZ
 R2-TARG ELEV
 R3-TARG ICENT

TARG AZ-TARGET
 AZIMUTH-ANGLE CLOCK-
 WISE FROM TRUE NORTH
 TO THE TARGET. IN
 DEGREES TO NEAREST
 .01 DEGREE

TARG ELEV-TARGET
 ELEVATION-ANGLE FROM
 THE LOCAL HORIZONTAL
 (OF NAV BASE) TO THE
 TARGET. IN DEGREES
 TO NEAREST .001
 DEGREE

TARG IDENT-TARGET
 IDENTIFIER-IDENTIFI-
 ES AZIMUTH AND
 ELEVATION FOR TARGET
 1 OR 2

WAIT FOR KEYBOARD
 ENTRY

MONITOR DSKY:
 OBSERVE VERB-NOUN
 FLASH TO REQUEST
 PROCEED AND DISPLAY
 OF TARGET 1 AZIMUTH
 AND ELEVATION

AM I SATISFIED WITH
 THE AZIMUTH AND
 ELEVATION OF TARGET
 1?

.Y .N

KEY IN PROCEED

#90

#100

#110

#120

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA

KEY IN V21E,V22E, OP
V24E AND LOAD NEW
DATA

#130

.P .NEW
.R .DATA
.D .
.C .

.E
.E DISPLAY AND
.D STORE NEW
DATA

#140

SET TARG IDENT TO
00002 IN R3 OF NOUN
30

#150

++
+08
+
+08
++

PASTE V05N30 AND
THEN V06N41 (DO NOT
OVERWRITE R3)

#160

HOLD . FLASH VERB-NOUN TO
..... REQUEST PROCEED AND
SNAP . DISPLAY STORED
TARGET 2 AZIMUTH AND
ELEVATION:
V06 N41
R1-TARG AZ
R2-TARG ELEV
R3-TARG IDENT

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
PROCEED AND DISPLAY
OF TARGET 2 AZIMUTH
AND ELEVATION

#170

TARG AZ-TARGET
 AZIMUTH-ANGLE CLOCK-
 WISE FROM TRUE NORTH
 TO THE TARGET. IN
 DEGREES TO NEAREST
 .01 DEGREE

TARG ELEV-TARGET
 ELEVATION-ANGLE FROM
 THE LOCAL HORIZONTAL
 (OF NAV BASE) TO THE
 TARGET. IN DEGREES
 TO NEAREST .001
 DEGREE

TARG IDENT-TARGET
 IDENTIFIER-IDENTIFI-
 ES AZIMUTH AND
 ELEVATION FOR TARGET
 1 OR 2

WAIT FOR KEYBOARD
 ENTRY

TERMINATE FLASH UPON
 RECEIPT OF PROCEED
 OR NEW DATA

P. .NEW
 R. .DATA
 O. .
 C. .
 E. -----
 E. DISPLAY AND
 D. STORE NEW
 . DATA

AM I SATISFIED WITH
 THE AZIMUTH AND
 ELEVATION OF TARGET
 2?

.Y .N

KEY IN PROCEED

KEY IN V21E,V22E,
 OR V24E AND LOAD NEW
 DATA

#180

#190

#200

#210

#220

 COMPUTE LOS TO
 TARGET NUMBER ONE
 AND DRIVE OPTICS,
 IF ALLOWED.

#230

HOLD
 SNAP
 ++
 +07
 +
 +07
 ++

FLASH VERB TO
 REQUEST PLEASE MARK
 V51N BLANK
 R1-BLANK
 R2-BLANK
 R3-BLANK

 MONITOR DSKY:
 OBSERVE VERB
 FLASH TO REQUEST
 PLEASE MARK

#240

 SELECT CMC OPTICS
 MODE, OBSERVE SXT
 DRIVE (OPTIONAL).

#250

 SELECT MANUAL OPTICS
 MODE

#260

 WAIT FOR MARK
 TERMINATE FLASH UPON
 RECEIPT OF MARK

MARK

 WHEN SIGHTING ON
 TARGET 1 IS SATIS-
 FACTORY PRESS MARK
 BUTTON

#270

 STORE MARK DATA

#280

FLASH VERB-NOUN TO
REQUEST PLEASE PER-
FORM TERMINATE MARK
SEQUENCE
V50 N25
R1-00016

MONITOR DSKY:
OBSERVE FLASHING
VERB-NOUN TO REQUEST
PLEASE PERFORM TERM-
INATE MARK SEQUENCE

YES NO

#290

WAIT FOR KEYBOARD
ENTRY

REJECT

PRESS MARK
REJECT BUTTON

#300

TERMINATE FLASH UPON
RECEIPT OF
PROCEED OR REJECT

KEY IN PROCEED

REJECT PROCEED

#310

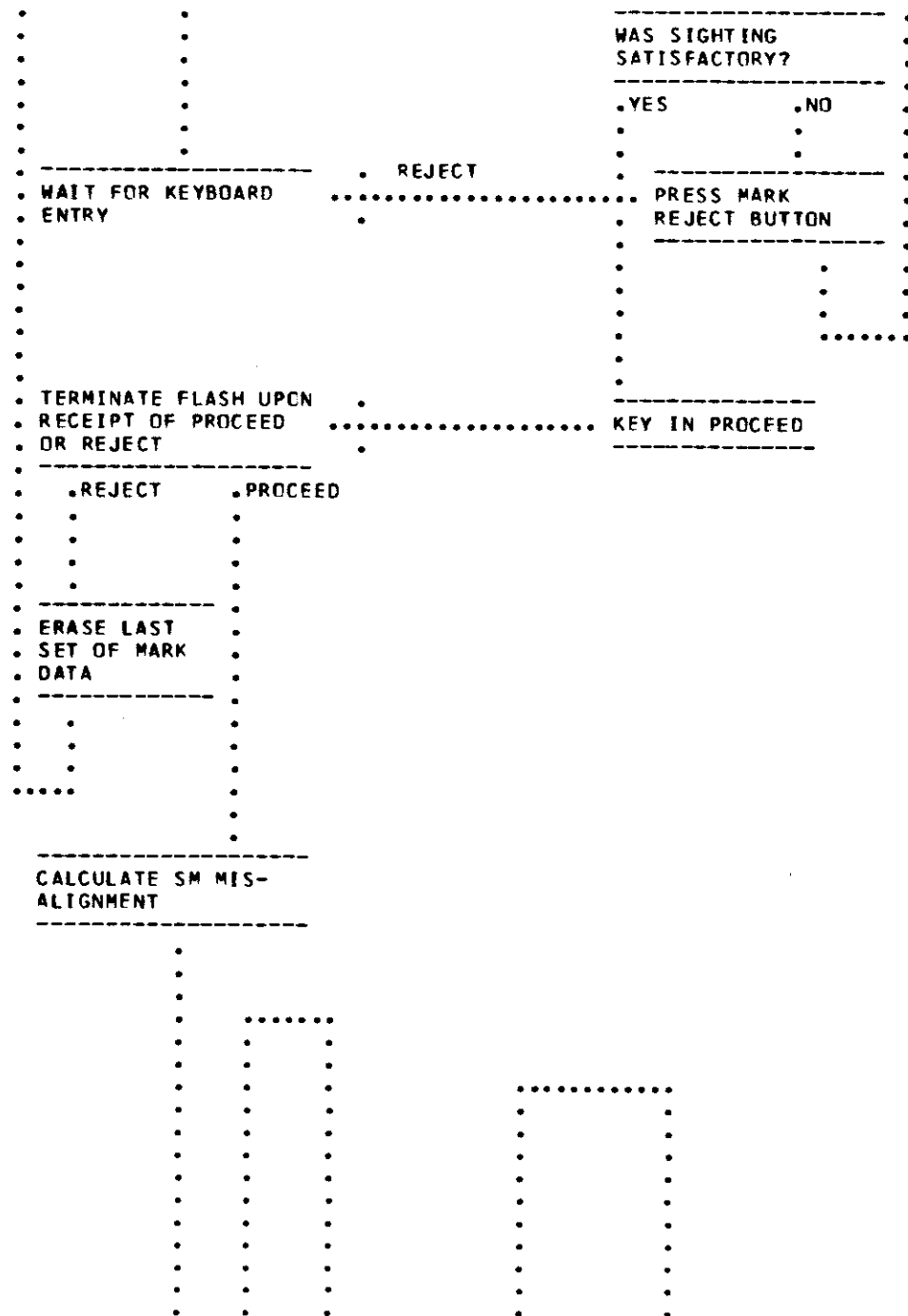
ERASE LAST
SET OF MARK
DATA

#320

COMPUTE LOS TO
TARGET NUMBER
TWO AND DRIVE
OPTICS IF
ALLOWED.

#330

56



#390

#400

#410

#420

#430

MONITOR DSKY:
OBSERVE FLASHING
VERB-NOUN AND
DISPLAY MISALIGNMENT
ANGLES

DELTA GYRO ANGLE-
ALIGNMENT ERROR IN
DEGREES TO NEAREST
.001 DEGREE

WAIT FOR KEYBOARD
ENTRY

CONFER WITH GROUND.
ARE GYRO MISALIGN-
MENT ANGLES ACCEPT-
ABLE?

• N	• Y
•	•
•	•

SHALL I CONTINUE WITH BAD DATA?

☐ NO ☐ YES

KEY IN TER-
MINATE
V34E

EXIT

#440

#450

#460

#470

#480

#490

HAVE I
ZEROED
THE X
AND Y
GYRO
ERROR?

N Y

#500

KEY IN V24E AND
LOAD R1 AND R2
WITH ZEROES.

#510

TERMINATE FLASH UPON
RECEIPT OF TERMINATE
OR PROCEED OR NEW
DATA

KEY IN PROCEED

#520

P	T	NEW
R	E	DATA
D	R	
C	M	
E	I	
E	N	
D	A	
	T	
	E	

#530

STORE NEW
DATA

#540

```

.-----
.      .
.      .
.      .
.      .
.     ...
.      .
. EXIT P03

```

-
-
-

TORQUE 2 GYRO PER
ERROR

•
•
•

TERMINATE P03 AND
REDISPLAY P02.
CONTINUE GYRO
COMPASSING.

```

      .
      .
      .
    . . .
      .
EXIT P03

```

#550

#560

#570

CHANGE CONTROL NOTES

LOGIC REV 06 PCR MIT 66
LOGIC REV 07 PCR 435
LOGIC REV 08 PCR 206
PCR 493

CMC POWER DOWN PROGRAM (P06)

LCGIC REV 08 07/08/68

PURPOSE: (1) TO TRANSFER THE CMC FROM THE OPERATE TO THE STANDBY CONDITION.

ASSUMPTIONS: (1) WHEN THIS PROGRAM IS TURNED ON THE ASTRONAUT MUST POWER DOWN THE CMC TO STANDBY.

(2) THE NORMAL CONDITION OF READINESS OF THE GNCS WHEN NOT IN USE IS STANDBY. ALL THE G/N CKT BKRS (PANEL 5) ARE CLOSED, THE IMU AND OPTICS G/N POWER SWITCHES (LEB PANEL 100) ARE OFF AND THE CMC STANDBY LIGHT (DSKY) IS ON. IN THIS CONDITION THE IMU IS IN STANDBY WITH ONLY HEATER POWER ON, OPTICS POWER IS OFF AND THE CMC IS IN STANDBY.

(3) A POSSIBLE CONDITION OF READINESS OF THE GNCS WHEN NOT COMPLETELY ON IS THE SAME AS STANDBY (2) ABOVE, EXCEPT THE CMC STANDBY LIGHT ON THE MAIN AND LEB DSKYS IS OFF. IN THIS CONFIGURATION THE CMC IS RUNNING FOR COMPUTATIONAL PURPOSES THAT DO NOT REQUIRE THE IMU OR OPTICS.

(4) IF THE COMPUTER POWER IS SWITCHED OFF IT WILL BE NECESSARY TO PERFORM A COMPUTER FRESH START (V36)E TO INITIALIZE THE ERASABLE STORAGE. THE CMC UPDATE PROGRAM (P27) WOULD HAVE TO BE DONE TO UPDATE THE STATE VECTOR AND COMPUTER CLOCK TIME.

(5) THE CMC IS CAPABLE OF MAINTAINING AN ACCURATE VALUE OF GROUND ELAPSED TIME (GET) FOR ONLY 23 HRS WHEN IN THE STANDBY MODE. IF THE CMC IS NOT BROUGHT OUT OF THE STANDBY CONDITION TO THE RUNNING CONDITION (SEE (3) ABOVE) AT LEAST ONCE WITHIN 23 HOURS THE CMC VALUE OF GET MUST BE UPDATED.

(6) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
			<ul style="list-style-type: none"> • CREW PROG • SELECTION • • • • 			
START CMC POWER DOWN PROGRAM (P06) DISPLAY PROGRAM 06	<ul style="list-style-type: none"> • • 	<ul style="list-style-type: none"> • • 	<ul style="list-style-type: none"> • • 	<ul style="list-style-type: none"> • KEY IN CMC POWER • DOWN PROGRAM (P06) • V37E 06E 		#10
		<ul style="list-style-type: none"> • • 	<ul style="list-style-type: none"> • • 	<ul style="list-style-type: none"> • MONITOR DSKY: • OBSERVE DISPLAY • OF PROGRAM 06 		
	<ul style="list-style-type: none"> • • • • 		<ul style="list-style-type: none"> • • • • 			#20
						P06/COLOSSUS P06/SUNDANCE P06/LUMINARY

RESET REFSMMAT FLAG

RESET RENDEZVOUS
FLAG

STORE CMC CLOCK TIME.

++
+08
++

HOLD.
.....
SNAP.

FLASH VERB-NOUN TO
REQUEST PLEASE PER-
FORM CMC POWER DOWN:
V50 N25
R1-00062
R2-BLANK
R3-BLANK

WAIT FOR KEYBOARD
ENTRY

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
PLEASE PERFORM CMC
POWER DOWN

PRESS PROCEED
BUTTON AND
HOLD DEPRESS-
ED UNTIL
STANDBY LIGHT
COMES ON.

#30

#40

#50

#60

#70

...

EXIT

EXIT
P06

PO6/COLOSSUS
PO6/SUNDANCE
PO6/LUMINARY

#80

#90

CHANGE CONTROL NOTES

LOGIC REV 07 PCR MIT 66
LOGIC REV 08 PCR NASA 206

THIS PAGE INTENTIONALLY LEFT BLANK

)

)

)

PURPOSE: (1) TO INDICATE TO THE ASTRONAUT THAT THE CMC HAS RECEIVED THE LIFTOFF DISCRETE.

(2) TO GENERATE AN ATTITUDE ERROR INDICATION ON THE FDI ERROR NEEDLES, SCALED FOR THE 50/15 SETTING; FROM LIFTOFF TO THE BEGINNING OF PITCHOVER/ROLLOUT THE ATTITUDE ERROR IS EQUAL TO THE DIFFERENCE BETWEEN THE CURRENT VEHICLE ATTITUDE AND THE ATTITUDE STORED AT LIFTOFF. DURING PITCHOVER/ROLLOUT THE ATTITUDE ERROR IS EQUAL TO THE DIFFERENCE BETWEEN THE CURRENT VEHICLE ATTITUDE AND THE CMC NOMINAL COMPUTATION OF VEHICLE ATTITUDE BASED ON THE STORED POLYNOMIALS IN PITCH AND ROLL.

(3) TO DISPLAY CMC COMPUTED TRAJECTORY PARAMETERS.

ASSUMPTIONS: (1) THE PROGRAM IS NORMALLY AUTOMATICALLY SELECTED BY THE GYRO COMPASSING PROGRAM (P02) WHEN THE CMC RECEIVES THE LIFTOFF DISCRETE FROM THE SIVB. IN THE BACKUP CASE IT WOULD HAVE BEEN SELECTED BY KEYING IN V75 ENTER AS NOTED EARLIER IN P02.

(2) THE ORBIT PARAMETER DISPLAY ROUTINE IS AVAILABLE BY KEYING IN V82E.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
	. CMC PROG . SELECTION					
	----- START ECI MONITOR PROGRAM (P11) DISPLAY PROGRAM 11 ----- MONITOR DSKY: . OBSERVE DISPLAY OF . PROGRAM 11 -----			#10
			
	----- SEND EFFECTIVE TIME OF LIFTOFF ON DOWNLINK -----		----- VERIFY AUTOMATIC START OF DIGITAL EVENT TIMER -----			
	. . .					#20
++ +06 ++	----- ZERO CMC CLOCK -----					
					

.
 .
 .

 UPDATE TEPHEM TO
 TIME OF LIFTOFF

#30

.
 .
 .

 CALL AVERAGE
 G INTEGRATION
 WITH DELTA V
 INTEGRATION

#40

.
 .
 .

 TERMINATE GYRO-
 COMPASSING

#50

.
 .
 .

 COMPUTE INITIAL
 STATE VECTOR

.
 .
 .

 COMPUTE REFSMMAT

#60

.
 .
 .

 SET REFSMMAT FLAG

.
 .
 .

 STORE LIFTOFF
 ATTITUDE

#70

CALL ROUTINE TO
LOAD ICDU DACS WITH
PITCH, ROLL, AND YAW
ATTITUDE ERRORS DE-
RIVED FROM PRESENT
ATTITUDE AND STORED
LIFTOFF ATTITUDE
UNTIL PRESENT TIME
EQUALS TE1 (STORED
IN ERASABLE MEMORY)
AT WHICH TIME THE
STORED LIFTOFF ATT-
ITUDE IS REPLACED BY
THE SOLUTION TO THE
STORED 6TH ORDER
BOOST POLYNOMIAL.

AT TIME TE1 + TE2
(TE2 IS STORED IN
ERASABLE MEMORY)
SHUT OFF BOOST POLY-
NOMIAL AND HOLD
ATTITUDE ERROR NEED-
LES CONSTANT AT
TERMINAL ERROR.

AT 163.84 SECS SHUT
OFF ROUTINE TO LOAD
ICDU DACS.

MONITOR:

(A) FDAI ATTITUDE
ERROR NEEDLES AS
INDICATION OF
CMC COMPUTATIONS
OF INSERTION.

BALL INDICATES
INITIAL VEHICLE
ROLLOUT AND THEN
GRADUAL PITCH-
OVER.

(B) DSKY:
R1-VI INCREASING
R2-H DOT FOLLOWS
NOMINAL HISTORY
R3-H INCREASING

#80

#90

#100

#110

#120

HOLD .
..... DISPLAY ON DSKY:
MON . V06 N62
R1 - VI
R2 - H DOT
R3 - H

VI-INERTIAL VELO-
CITY MAGNITUDE. IN
FPS TO NEAREST FPS

H DOT - RATE OF
CHANGE OF VEHICLE
ALTITUDE ABOVE
LAUNCH PAD RADIUS.
IN FPS TO
NEAREST FPS

SATURN
INN

TERMINATE P11 AND GO
TO PROGRAM SELECTED
VIA R00

KEY IN V37EXXE

EXIT P11

EXIT P11

#150

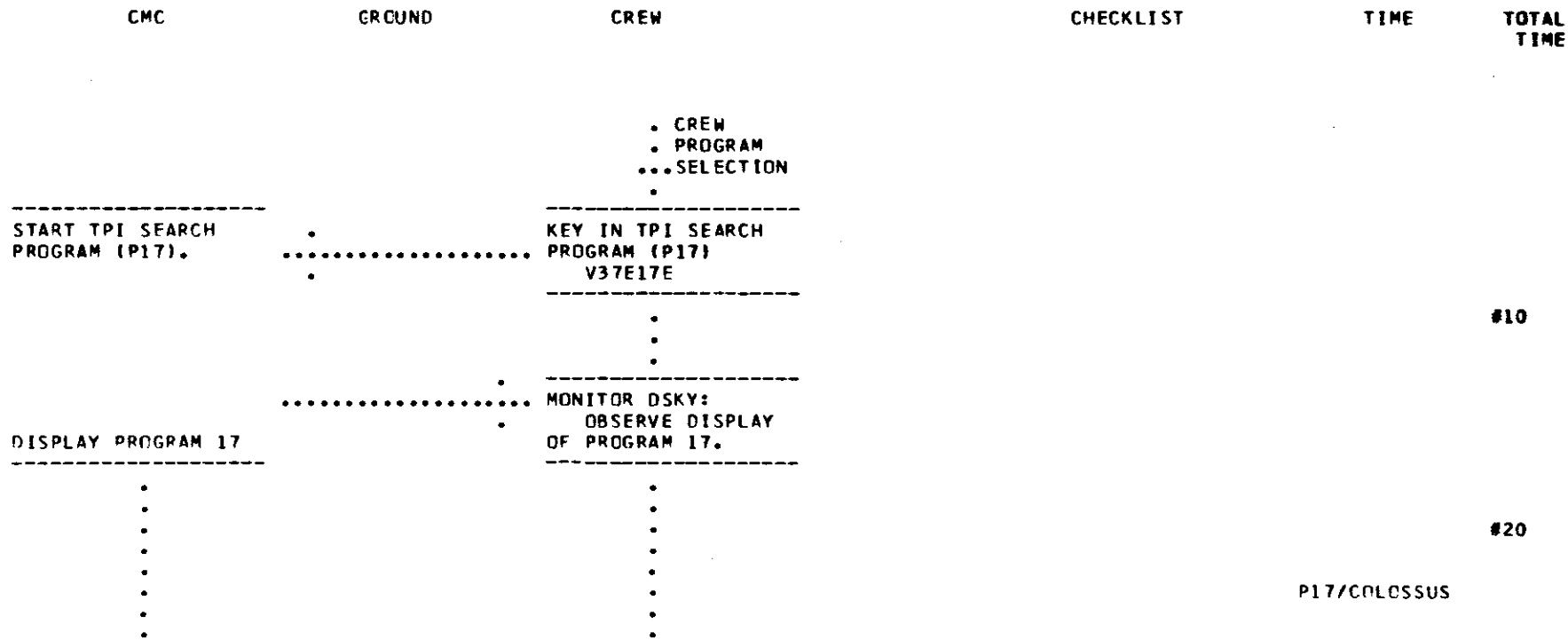
CHANGE CONTROL NOTES

LOGIC REV 5 PCR 3
LOGIC REV 6 PCR 206

TPI SEARCH PROGRAM (P17)

LOGIC REV 09 11/27/68

- PURPOSE:
- (1) TO ACCEPT A DESIRED TIME OF TRANSFER PHASE INITIATION (TIG(TPI)) AS A DSKY INPUT FROM THE ASTRONAUT, AND TO COMPUTE THEREFROM THE PARAMETERS ASSOCIATED WITH A MINIMUM ENERGY, SAFE PERIAPSIS TRANSFER MANEUVER AT TIG(TPI) AND THE RESULTANT RENDEZVOUS INTERCEPT.
 - (2) TO PROVIDE THE ASTRONAUT WITH THE OPTION OF DEFINING TO THE CMC THE INITIAL TRANSFER TRAJECTORY SEARCH SECTOR FOR CENTRAL ANGLES EITHER GREATER THAN OR LESS THAN 180 DEGREES FROM THE POSITION OF THE ACTIVE VEHICLE (CSM) AT TIG (TPI).
 - (3) TO DISPLAY TO THE ASTRONAUT THE PARAMETERS ASSOCIATED WITH THE TRANSFER (TPI AND INTERCEPT).
- ASSUMPTIONS:
- (1) IF P20 IS IN OPERATION WHILE THIS PROGRAM IS OPERATING THE ASTRONAUT MAY HOLD AT ANY FLASHING DISPLAY AND TURN ON THE RENDEZVOUS SIGHTING MARK ROUTINE (EITHER R21 OR R23) AND TAKE OPTICS MARKS AND/OR HE MAY ALLOW VHF RANGING MARKS TO ACCUMULATE. SEE P20 FOR DETAILED DESCRIPTION.
 - (2) THE OPERATION OF THIS PROGRAM UTILIZES THE ACTIVE VEHICLE FLAG WHICH DESIGNATES THE VEHICLE WHICH IS DOING THE RENDEZVOUS THRUSTING MANEUVERS TO THE PROGRAM WHICH CALCULATES THE MANEUVER PARAMETERS. SET AT THE START OF EACH RENDEZVOUS PRE-THRUSTING PROGRAM.
 - (3) TO EXECUTE THE TPI MANEUVER SELECT THE TRANSFER PHASE INITIATION (TPI) PROGRAM (P34).
 - (4) THIS PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.



SET TRACK FLAG
(SEE P20)

SET UPDATE FLAG
(SEE P20)

SET ACTIVE VEHICLE
FLAG TO CSM.

"A"

"A"

HOLD .
.....
SNAP .

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY TIG(TPI):
VO6N37
R1-TIG(TPI)-HRS
R2-TIG(TPI)-MINS
R3-TIG(TPI)-SECS

TIG(TPI)-TIME OF TPI
IGNITION (GET). IN
HRS, MINS, SECS TO
NEAREST .01 SEC.

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST RE-
SPONSE AND DISPLAY
OF TIG(TPI).

AM I SATISFIED WITH
THESE VALUES?

.Y

.N

RECORD THIS
VALUE.

#30

#40

#50

#60

#70

WAIT FOR KEYBOARD
ENTRY

KEY IN PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA

KEY IN V25E AND
LOAD THE DESIR-
ED TIG(TPI)

.PROCEED .NEW
.DATA
STORE DATA

RESET UPDATE FLAG

COMPUTE VEHICLE
POSITIONS AT
TIG(TPI) AND SELECT
NOMINAL SEARCH OP-
TION AS FOLLOWS:
IF ACTIVE VEHICLE IS
ABOVE PASSIVE VEH-
ICLE (NEGATIVE DELTA
ALTITUDE) SELECT
OPTION 00001.
IF ACTIVE VEHICLE IS
BELOW PASSIVE VEHICLE
(POSITIVE DELTA ALT)
SELECT OPTION 00002

SET UPDATE FLAG

#80

#90

#100

#110

#120

HOLD	FLASH VERB-NOUN TO	MONITOR DSKY:	
.....	REQUEST RESPONSE AND	OBSERVE VERB-NOUN	#130
SNAP	DISPLAY TPI PARA-	FLASH TO REQUEST	
	METERS:	RESPONSE AND DISPLAY	
	V06N72	OF TPI PARAMETERS.	
	R1-DELT ANG(TPI)		
	R2-DELT ALT(TPI)		
	R3-SEARCH OPTION		
	DELT ANG(TPI)-CENT-		
	RAL ANGLE AROUND THE		
	EARTH OR MOON AT TIG		#140
	(TPI) BETWEEN THE		
	ACTIVE AND PASSIVE		
	VEHICLES. POLARITY		
	INDICATES ACTIVE VE-		
	HICLE IS BEHIND (-)		
	OR AHEAD OF (+)		
	PASSIVE VEHICLE. IN		
	DEGREES TO NEAREST		
	.01 DEGREE.		
	DELT ALT(TPI)-THE		#150
	MAGNITUDE OF THE		
	ALTITUDE DIFFERENCE		
	BETWEEN THE POSITION		
	VECTORS OF THE ACT-		
	IVE AND PASSIVE VE-		
	HICLES AT TIG(TPI).		
	IN NAUTICAL MILES TO		
	THE NEAREST .1 NM.		
	POLARITY INDICATES		#160
	THE PASSIVE VEHICLE		
	IS ABOVE (+) OR BE-		
	LOW (-) THE ACTIVE		
	VEHICLE.		
	SEARCH OPTION-AN OP-		
	TION CODE TO CONTRL		
	CMC SEARCH FOR TERM-		
	INAL PHASE		
	DEFINITION:		#170
	00001-CMC SEARCHES		
	TO DEFINE A TERMINAL		
	PHASE WHERE THE REN-		
	DEZVOUS INTERCEPT		
	OCCURS AT A CENTRAL		
	ANGLE LESS THAN 180		
	DEGREES FROM THE		
	ACTIVE VEHICLE PCS-		
	ITION AT TIG(TPI).		
	OR		
	00002-SAME AS		#180
	00001 EXCEPT CENTRAL		

ANGLE IS GREATER
THAN 180 DEGREES.

WAIT FOR KEYBOARD
ENTRY.

DO I WISH TO
CHANGE TIG(TPI)?

.N .Y

KEY IN RECY-
CLE
V32F

GO TO
"A"
ABOVE

DO I WISH TO
CHANGE THE SEARCH
OPTION DISPLAYED?

.N .Y

RECORD THESE
VALUES

KEY IN PROCEED

#190

#200

#210

#220

#230

THIS PAGE INTENTIONALLY LEFT BLANK

)

)

)

#290

.NO .ALARM . .	.ALARM . . .
SET UP- DATE FLAG (SEE P20)	SET UP- DATE FLAG (SEE P20)

#300

POSS
HOLD .
.....
SNAP .

```

.
.
.
.
.-----
. FLASH VERB-NOUN
. TO REQUEST RE-
. SPONSE AND DIS-
. PLAY PROGRAM
. ALARM CODE.
.     V05N09
.     R1-
.     R2-
.     R3-
.

```

• EXPECTED ALARM
• CODE AT THIS
• TIME IS 00124

• •

• •

• •

- WAIT FOR KEY-
- BOARD ENTRY

- TERMINATE FLASH
- UPON RECEIPT OF
- RECYCLE.

. R
 . E
 . C
 . Y
 . C
 . L
 . E
 .
 .
 .
 .
 .

MONITOR DSKY:
DOFS PROGRAM
ALARM INDICATE THAT
CMC CAN FIND NO SOL-
UTION.

.N .Y

- TO ADJUST TIG
- (TPI) AND/OR THE
- SEARCH OPTION
- KEY IN RECYCLE
- V32E

GO TO
"A"
ABOVE

4310

#320

#330

#340

SNAP

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY CALCULATED
TERMINAL PHASE
PARAMETERS:

V06N58
R1-PER ALT
R2-DELTA V(TPI)
R3-DELTA V(TPF)

PER ALT-ALTITUDE OF
PERIGEE ABOVE LAUNCH
PAD RADIUS (EARTH
ORBIT) OR ALTITUDE
OF PERILUNE ABOVE
THE MOST RECENTLY
DEFINED LANDING SITE
RADIUS (LUNAR ORBIT)
AFTER TPI MANEUVER
IN NAUTICAL MILES TO
NEAREST .1 NM.

DELTA V(TPI)-REQUIRED IMPULSIVE DELTA V TO ACCOMPLISH TPI MANEUVER AT TIG (TPI). IN FPS TO NEAREST .1 FPS.

DELTA V(TPF)-REQUIRED IMPULSIVE DELTA V TO ACCOMPLISH TPF MANEUVER AT TIME OF INTERCEPT. POLARITY INDICATES ACTIVE VEHICLE ASCENDING (+) OR DESCENDING (-) TO INTERCEPT. IN FPS TO NEAREST .1 FPS.

WAIT FOR KEYBOARD
ENTRY.

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF CALCULATED TER-
MINAL PHASE
PARAMETERS.

DO I WISH TO ADJUST
TIG(TPI) AND/CR THE
SEARCH OPTION?

• N • Y
• •
• •
• •
• •
• •

RECORD THESE
VALUES.

KEY IN PROCEED

#390

CENTANG-THE ORBITAL
CENTRAL ANGLE OF THE
PASSIVE VEHICLE
DURING TRANSFER FROM
TIG(TPI) TO TIME
OF INTERCEPT. IN
DEGREES TO THE NEAR-
EST .01 DEGREES.

DO I WISH TO ADJUST
TIG(TPI) AND/OR THE
SEARCH OPTION?

.N .Y
.
.

#450

RECORD THESE
VALUES

.
.
.

#460

WAIT FOR KEYBOARD
ENTRY.

KEY IN PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCEED,
OR RECYCLE.

KEY IN RECYCLE
V32E

#470

.P .R
.R .E
.O .C
.C .Y
.E .C
.E .L
.D .E

GO TO
"A"
ABOVE

#480

GO TO
"A"
ABOVE

DO ROUTINE R00

DO ROUTINE R00

#490

EXIT

EXIT

CHANGE CONTROL NOTES

LOGIC REV 08 PCR MIT 66
LOGIC REV 09 EDITORIAL

THIS PAGE INTENTIONALLY LEFT BLANK

PURPOSE:

- (1) TO CONTROL THE CSM ATTITUDE AND OPTICS TO ACQUIRE THE LM IN THE SXT FIELD AND TO POINT THE CSM TRANSPONDER AT THE LM OR TO CONTROL THE CSM ATTITUDE TO ACQUIRE THE LM ALONG THE +X AXIS BASED ON THE OPTION CODE CHOSEN.
- (2) TO UPDATE EITHER THE LM OR CSM STATE VECTOR (AS SPECIFIED BY THE ASTRONAUT BY DSKY ENTRY) ON THE BASIS OF OPTICAL TRACKING DATA AND/OR VHF RANGE DATA.

ASSUMPTIONS:

- (1) THE IMU MUST BE ON AND ALIGNED IN ORDER TO PERFORM THIS PROGRAM.
- (2) THE GNCS IS IN CONTROL OF THE VEHICLE IN THE AUTO MODE IN THE NOMINAL CASE. IF THE ASTRONAUT TAKES OVER CONTROL OF THE VEHICLE WITH RHC THE CSM WILL REMAIN AT THE ATTITUDE IT IS DRIVEN TO. REGARDLESS OF MODE SELECTION THE GNCS WILL CALCULATE THE PREFERRED TRACKING ATTITUDE AND THE +X-AXIS TRACKING ATTITUDE.
- (3) ROUTINE R03 HAS BEEN PERFORMED PRIOR TO SELECTION OF THIS PROGRAM, IN ORDER FOR THE GNCS TO PERFORM THE AUTOMATIC ATTITUDE MANEUVERS THE ASTRONAUT SHOULD KEY IN V46F AT SOME TIME PRIOR TO THE FIRST MANEUVER.
- (4) THE LM IS MAINTAINING A PREFERRED TRACKING ATTITUDE TO CORRECTLY ORIENT THE OPTICAL BEACON.
- (5) THE OPERATION OF THE PROGRAM INCLUDES THE FOLLOWING FLAGS:

RENDEZVOUS FLAG- CONTROLS THE PERMANENT TERMINATION OF THE TOTAL RENDEZVOUS NAVIGATION PROCESS. THIS PROGRAM WILL ONLY RUN OR RESUME RUNNING WHEN THIS FLAG IS SET. SET BY P20 SELECTION. RESET BY SELECTION OF CMC IDLING PROGRAM(P001), CMC POWER DOWN PROGRAM(P06), THE ORBITAL NAVIGATION PROGRAM (P22), THE Cislunar NAVIGATION PROGRAM (P23), OR BY V56F. THE KEYING IN OF V56F WILL IMMEDIATELY TERMINATE P20 UNLESS A NAVIGATION MEASUREMENT IS BEING PROCESSED IN WHICH CASE IT WILL HOLD UNTIL COMPLETION OF THE INCORPORATION AND THEN TERMINATE P20.

TRACK FLAG- CONTROLS THE TEMPORARY TERMINATION OF THE TOTAL RENDEZVOUS NAVIGATION PROCESS. RESET OF THIS FLAG INTERRUPTS THE AUTOMATIC ATTITUDE MANEUVER/OPTICS POINTING PROCESS, AS WELL AS THE STATE VECTOR UPDATE PROCESS. THE SET OF THIS FLAG ENABLES ALL THESE PROCESSES. SET BY P17, 20, 30, 31, 34, 35, 38, 39, 74, 75, 76, 77, 78, 79 SELECTION. RESET BY AN V37EXX

UPDATE FLAG- CONTROLS THE TEMPORARY TERMINATION OF THE STATE VECTOR UPDATE PROCESS ONLY. SET BY P17, 20, 30, 31, 34, 35, 38, 39, 74, 75, 77, 78, 79 SELECTION. RESET BY ANY V37EXX AND IT IS ALSO RESET AND SET DURING THE PRETHRUST COMPUTATIONS TO PROTECT ERASABLE MEMORY.

PREFERRED ATTITUDE FLAG-DEFINES WHICH ATTITUDE THE SPACECRAFT IS TO BE ALIGNED TO BY R61. SET BY TURN ON OF P20, OR V76F. RESET BY V77F.

STICK FLAG - RESET BY EXECUTION OF A PROGRAM CHANGE VIA R00 AND BY V58F. SET BY TAKING RHC OUT OF DETENT WHEN THE SC CONTROL SWITCH IS CMC AND WHEN THE THO IS NOT CLOCKWISE. CPU RATE DRIVE IS NOT PERFORMED IF THE STICK FLAG IS SET.

STATE VECTOR FLAG - DEFINES WHICH STATE VECTOR WILL BE UPDATED BY SIGHTING MARKS AND VHF RANGING. SET TO LM BY P20 TURN ON AND V70F. SET TO CSM BY V81F.

1A

1A

PCR
606
++
+22
++

VHF RANGE FLAG - SET BY V87E, RESET BY V88E, AND BY ANY EXECUTION OF A PROGRAM CHANGE VIA R00. ALLOWS AUTOMATIC VHF RANGE DATA TO BE USED BY THE RENDEZVOUS DATA PROCESSING ROUTINE (P22).

(6) THIS PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY. IT MAY BE TERMINATED BY THE SELECTION OF THE CMC IDLING PROGRAM (P00), CMC POWER DOWN PROGRAM (P06), THE ORBITAL NAVIGATION PROGRAM (P22), THE Cislunar NAVIGATION PROGRAM (P23) OR BY V56F. P00 SELECTION WILL TERMINATE P20 AND ANY OTHER PROGRAM IN PROCESS AND ESTABLISH P00. ALL OTHER PROGRAMS SELECTED WILL CHANGE THE PROGRAM NUMBER DISPLAYED ON THE DSKY BUT WILL NOT TERMINATE P20. THIS PROGRAM IS DESIGNED TO OPERATE AUTOMATICALLY AND SIMULTANEOUSLY WITH ANOTHER PROGRAM WITHOUT REQUIRING USE OF THE DSKY UNLESS NON-NOMINAL CIRCUMSTANCES REQUIRE CMC COMMUNICATION WITH THE ASTRONAUT. IF V56F IS KEYED INTO THE DSKY AND P20 IS THE ONLY PROGRAM RUNNING, ROUTINE R00 WILL BE INITIATED.

(7) W MATRIX INITIALIZATION FOR RENDEZVOUS MAY BE ENABLED (RESET RENDEWFLAG) IN ANY OF THE FOLLOWING WAYS:

- (A) KEYING IN OF VERR 93E
- (B) COMPUTER FRESH START (KEYING IN OF VERR 36E)
- (C) STATE VECTOR UPDATE FROM THE GROUND
- (D) STATE VECTOR CALCULATION MADE DURING ENTRANCE TO P22, OR P23.
- (E) CHANGES TO INITIALIZATION PARAMETERS IN V67

(8) THERE IS A RENDEZVOUS OPTICS MARK COUNTER USED IN THE CMC TO COUNT THE NUMBER OF OPTICS MARKS USED TO CHANGE EITHER STATE VECTOR AND THERE IS A RENDEZVOUS VHF RANGING MARK COUNTER USED IN THE CMC TO COUNT THE NUMBER OF VHF RANGING MARKS USED TO CHANGE EITHER STATE VECTOR.

THESE COUNTERS ARE ZEROED BY SEVERAL DISTINCT EVENTS, THEY ARE:

- (A) KEYING IN OF V37E20E (INITIATION OF (P20)).
- (B) COMPLETION OF THE TARGET DELTA V PROGRAM (P76).
- (C) SELECTION OF A NEW PROGRAM FROM A PROGRAM WHICH HAD TURNED ON AVERAGE G (SEE R00 LOGIC).
- (D) INITIALIZATION OF THE W MATRIX FOR RENDEZVOUS (FOR ANY REASON, SEE ASSUMPTION 7).
- (E) SELECTION OF RETURN TO EARTH PROGRAM (P37)

(9) SUMMARY OF EXTENDED VERRS ASSOCIATED WITH THE PROGRAM:

- V44E - SET SURFACE FLAG-CAUSES TRACKING OF THE LANDING SITE BASED ON THE STORED LANDING SITE
- V45E - RESET SURFACE FLAG-CAUSES TRACKING OF THE LM BASED ON THE LM STATE VECTOR
- V54E - ON R23 - ALLOWS BACK-UP MARKING ON THE LM.
- V56F - RESET THE RENDEZVOUS FLAG. CAUSES TERMINATION OF P20.
- V57E - ON R21 - ALLOWS OPTICS MARKING ON THE LM.
- V58E - RESET STICK FLAG - ALLOW AUTO MANEUVERS
- V67E - W-MATRIX RMS ERROR DISPLAY
- V76E - SET PREFERRED ATTITUDE FLAG - DRIVE TO PREFERRED ATTITUDE
- V77E - RESET PREFERRED ATTITUDE FLAG - DRIVE TO +X-AXIS ATTITUDE
- V80E - SET STATE VECTOR FLAG TO LM. DATA WILL UPDATE LM STATE VECTOR
- V81E - SET STATE VECTOR FLAG TO CSM. DATA WILL UPDATE CSM STATE VECTOR
- V87E - SET VHF RANGE FLAG - ALLOWS R22 TO ACCEPT RANGE DATA.
- V88E - RESET VHF RANGE FLAG - STOPS ACCEPTANCE OF RANGE DATA.
- V93E - RESET RENDEWFLAG - CAUSE INITIALIZATION OF W MATRIX FOR RENDEZVOUS AT NEXT DATA INCORPORATION

INCOMPLETE SUMMARY OF ADDITIONAL DATA OR JOBS AVAILABLE WHILE THIS PROGRAM IS OPERATING:

P17 - AVAILABLE BY KEYING IN V37E17E
 P30 - AVAILABLE BY KEYING IN V37E30E
 P31 - AVAILABLE BY KEYING IN V37E31E
 P34 - AVAILABLE BY KEYING IN V37E34E
 P35 - AVAILABLE BY KEYING IN V37E35E
 P38 - AVAILABLE BY KEYING IN V37E38E
 P39 - AVAILABLE BY KEYING IN V37E39E
 P74 - AVAILABLE BY KEYING IN V37E74E
 P75 - AVAILABLE BY KEYING IN V37E75E
 P76 - AVAILABLE BY KEYING IN V37E76E
 P77 - AVAILABLE BY KEYING IN V37E77E
 P78 - AVAILABLE BY KEYING IN V37E78E
 P79 - AVAILABLE BY KEYING IN V37E79E
 R03 - AVAILABLE BY KEYING IN V48E
 R30 - AVAILABLE BY KEYING IN V82E
 R31 - AVAILABLE BY KEYING IN V83E
 R34 - AVAILABLE BY KEYING IN V85E
 N09 - AVAILABLE BY KEYING IN V15N09E
 N20 - AVAILABLE BY KEYING V16N20E
 N22 - AVAILABLE BY KEYING V16N22E
 N36 - AVAILABLE BY KEYING V16N36E
 N91 - AVAILABLE BY KEYING V16N91E
 N92 - AVAILABLE BY KEYING V16N92E
 N95 - AVAILABLE BY KEYING V16N95E
 N96 - AVAILABLE BY KEYING V16N96E
 V46E- AVAILABLE BY KEYING IN V46E
 V60 - AVAILABLE BY KEYING V60E
 V61 - AVAILABLE BY KEYING V61E
 V62 - AVAILABLE BY KEYING V62E
 V63 - AVAILABLE BY KEYING V63E
 V96E- AVAILABLE BY KEYING V96E

 PROG
CONT

CMC

GROUND

CREW

CHECKLIST

TIME

TOTAL
TIME
 .CREW
 .PROGRAM
 .SELECTION
 .
 .
 .

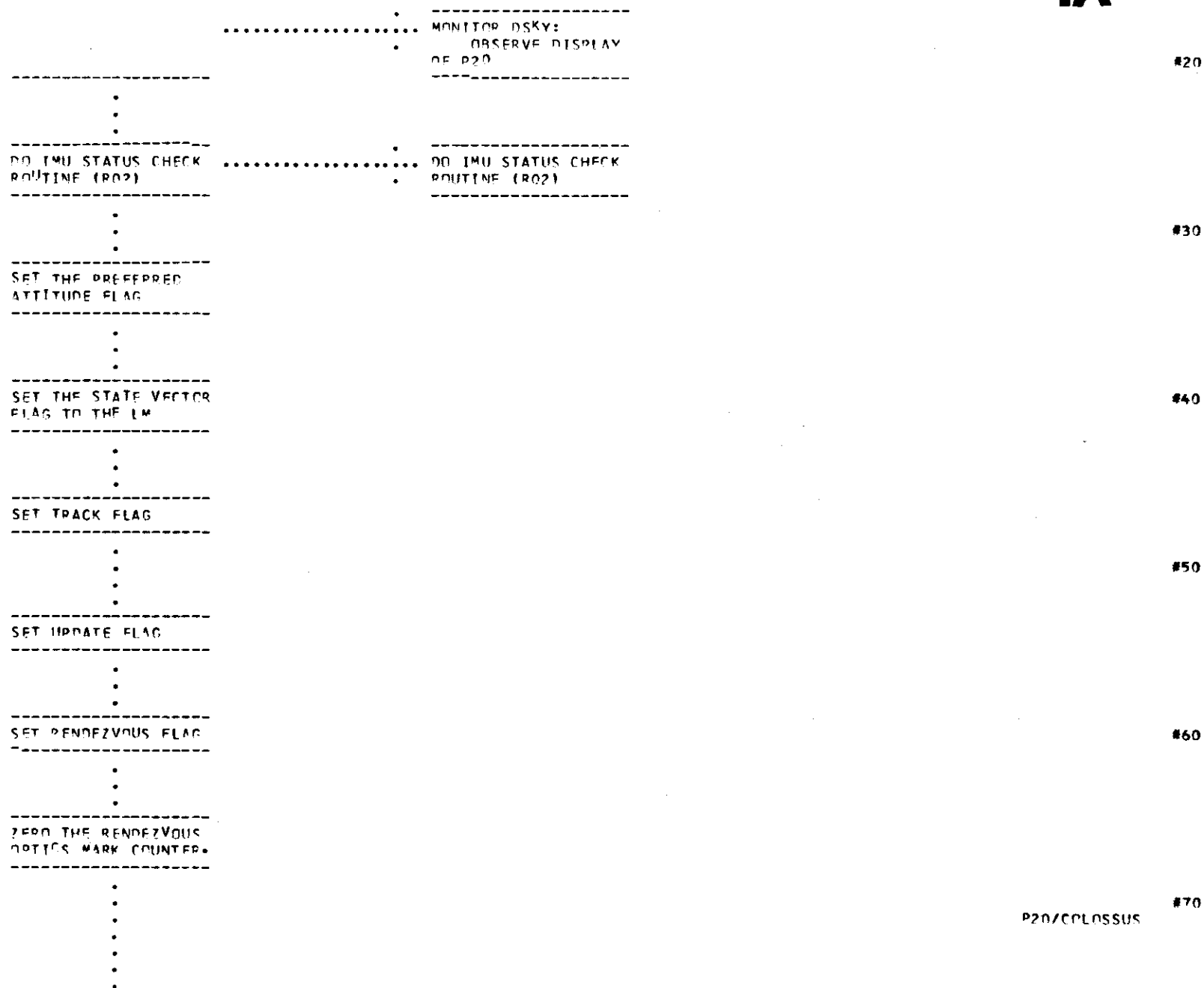
 START RENDEZVOUS
 NAVIGATION PROGRAM
 (P20).
 DISPLAY P20

 .

 KEY IN RENDEZVOUS
 NAVIGATION PROGRAM
 (P20)
 V37E20E

#10

1A

1A

EXTRAPOLATE PERMANENT STATE VECTORS (LM AND CSM) TO THE PRESENT TIME USING PRECISION INTEGRATION.

IS TRACK FLAG SET?

IS THE REFSMMAT FLAG
SET?

#120

1A

1A

CALL THE RENDEZVOUS
DATA PROCESSING
ROUTINE (R22)

THE RENDEZVOUS DATA
PROCESSING ROUTINE
IS NOW AUTOMATIC

#130

SET R61 COUNTER
EQUAL TO ZERO.

#140

SET LM TARGET FLAG
FOR USE BY AUTO
OPTICS POSITIONING
ROUTINE (R52).

#150

DO TRACKING ATTITUDE
ROUTINE (R61) WHICH
WILL CALL ATTITUDE
MANEUVER ROUTINE
(R60)

DO TRACKING ATTITUDE
ROUTINE (R61) WHICH
WILL CALL ATTITUDE
MANEUVER ROUTINE
(R60)

#160

++
+22
++
PCR
606

++
+22
++
PCR
606

1A

GO TO CMC
IDLING PROGRAM
(P00) VIA R00
(NOTE: P00
WILL TERMINATE
THE RENDEZVOUS,
TRACK, AND UP-
DATE FLAGS AND
ALSO THE AUTO
OPTICS POSI-
TIONING ROU-
TINE (R52))
AND THE RENDEZ-
VOUS DATA PRO-
CESSING ROUT-
INE (R22)

OBSERVE
DISPLAY OF P00.

EXIT P20

#230

#240

TERMINATE
THE REN-
DEZVOUS,
TRACK,
AND, UP-
DATE
FLAGS AND
THE AUTO
OPTICS
POSITION-
ING ROU-
TINE
(R52)
AND THE
RENDEZ-
VOUS DATA
PROCESS-
ING ROUT-
INE (R22)

#250

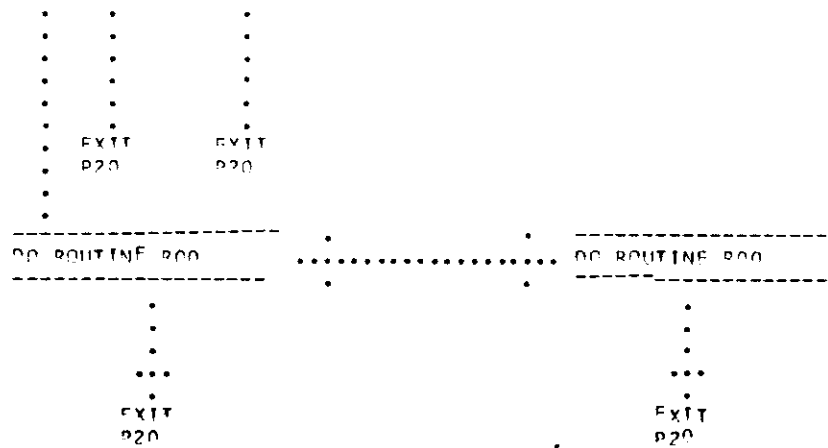
#260

IS P20 THE
ONLY PRO-
GRAM RUN-
NING?

#270

Y N

13



P20/DO ROUTINE

#290

#290

CHANGE CONTROL NOTES

LOGIC REV 18	PCR NASA 147
LOGIC REV 19	PCR MIT 66
LOGIC REV 20	PCR 468
	PCR 206
	PCR 465
LOGIC REV 21	PCR 679
LOGIC REV 22	PCR 605

THIS PAGE INTENTIONALLY LEFT BLANK

PURPOSE: (1) TO PROVIDE THE ASTRONAUT DETAILS OF HIS GROUND TRACK WITHOUT THE NEED FOR GROUND COMMUNICATION.

ASSUMPTIONS: (1) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY CSKY ENTRY.

(2) THIS PROGRAM MAY BE SELECTED WHILE THE CSM IS IN EITHER EARTH OR LUNAR ORBIT TO DEFINE THE GROUND TRACK OF EITHER THE LM OR CSM.

(3) THIS PROGRAM ASSUMES THE VEHICLE WHOSE GROUND TRACK PARAMETERS ARE CALCULATED TO REMAIN IN FREE FALL FROM THE PRESENT TIME UNTIL T LAT LONG.

PROG CONT	CNC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

START GROUND TRACK
DETERMINATION
PROGRAM (P21).
DISPLAY PROGRAM 21

KEY IN GROUND TRACK
DETERMINATION
PROGRAM (P21)
V37E21E

- CREW
- PROGRAM
- SELECTION

•

•

MONITOR DSKY:
OBSERVE DISPLAY OF
PROGRAM 21

#10

#20

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY OPTION CODE
FOR ASSUMED VEHICLE
(LM OR CSM)

V04 N06
R1 00002
R2 0000X
R3 BLANK

R1 IS THE OPTION
CODE FOR ASSUMED
VEHICLE.

R2 IS THE ASSUMED
OPTION:

00001 - THIS
VEHICLE
00002 - OTHER
VEHICLE

WAIT FOR KEYBOARD
ENTRY

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA.

.P .NEW
.R .DATA
.Q .
.C
.E STORE
.E NEW
.D DATA

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF OPTION CODE FOR
ASSUMED VEHICLE
(LM OR CSM).

IS THE ASSUMED
OPTION CORRECT?

.Y .N
. .
. .
. .
. .
. .
. .
. .

KEY IN
PROCEED

KEY IN V22E
AND LOAD THE
DESIRED OPTION
INTO R2

#30

#40

#50

#60

#70

#80

HOLD
 SNAP
 FLASH VERB-NOUN TO
 REQUEST RESPONSE AND
 DISPLAY T LAT LONG:
 V06 N34
 R1-T LAT LONG-HRS
 R2-T LAT LONG-MIN
 R3-T LAT LONG-SECS

T LAT LONG - TIME
 (GET) AT WHICH LAT
 AND LONG OF VEHICLE
 POSITION IS DESIRED
 IN HRS, MINS, SECS
 TO NEAREST .01 SEC.

WAIT FOR KEYBOARD
 ENTRY

TERMINATE FLASH UPON
 RECEIPT OF NEW DATA
 OR PROCEED

.NEW .P
 .DATA .R
 .C
 .C
 STORE .E
 NEW .E
 DATA .C

MONITOR DSKY:
 OBSERVE VERB-NOUN
 FLASH TO REQUEST
 RESPONSE AND DISPLAY
 OF T LAT LONG. AM I
 SATISFIED WITH THIS
 TIME?

.N .Y

KEY IN
 PROCEED

KEY IN V25E AND LOAD
 DESIRED T LAT LONG.

#90

#100

#110

#120

#130

CALCULATE LATITUDE,
LONGITUDE AND ALTI-
TITUDE OF VEHICLE AT
T LAT LONG

#140

HOLD .. FLASH VERB-NOUN TO
..... REQUEST RESPONSE AND
SNAP ... DISPLAY LATITUDE,
LONGITUDE AND
ALTITUDE:
V06 N43
R1-LAT
R2-LONG
R3-ALT

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF LATITUDE, LONGI-
TITUDE AND ALTITUDE

#150

LAT-LATITUDE OF
VEHICLE. + IS NORTH.
IN DEGREES TO NEAR-
EST .01 DEGREE.

LONG-LONGITUDE OF
VEHICLE. + IS EAST.
IN DEGREES TO NEAR-
EST .01 DEGREE.

#160

ALT-ALTITUDE OF
VEHICLE ABOVE THE
LAUNCH PAD RADIUS
(EARTH ORBIT) OR
THE LUNAR RADIUS AT
THE MOST RECENTLY
DEFINED (SEE SECTION
5 OF R577)
LANDING SITE (LUNAR
ORBIT). IN NAUTICAL
MILES TO NEAREST .1
NM.

#170

#180

DO I WISH TO OBTAIN
NEW PARAMETERS FOR
A TIME 10 MIN.
LATER THAN THAT
WHICH APPLIES TO
PRESENT DISPLAY?

.N .Y

WAIT FOR KEYBOARD
ENTRY

WHEN FINISHED
WITH DISPLAY
KEY IN
PROCEED

#190

TERMINATE FLASH UPON
RECEIPT OF RECYCLE
OR PROCEED

WHEN FINISHED
WITH DISPLAY KEY
IN RECYCLE
V32E

#200

.R .P
.E .R
.C .O
.Y .C
.C .E
.L .E
.E .D

INCREMENT T-LAT-
LONG BY 10 MIN.

#210

DO ROUTINE R00

DO ROUTINE R00

#220

EXIT

EXIT

#230

LOGIC REV 6 PCR 206

- PURPOSE:**
- (1) TO LOCATE AND TRACK A LANDMARK SUITABLE FOR NAVIGATION PURPOSES.
 - (2) TO OBTAIN SIGHTING MARKS ON THE CHOSEN LANDMARK.
 - (3) TO CALCULATE THE ORBITAL PARAMETER CHANGES GENERATED BY THE LANDMARK SIGHTING MARKS.
 - (4) TO DISPLAY THE ORBITAL PARAMETER CHANGES GENERATED BY THE FIRST SIGHTING MARK ON A LANDMARK, FOR DECISION BY THE NAVIGATOR/GROUND AS TO THE VALIDITY OF THE LANDMARK AND NAVIGATION PROCESS PRIOR TO INCORPORATION OF STATE VECTOR CHANGES AS A RESULT OF THE SIGHTING MARKS.
 - (5) TO PROVIDE UPDATED COORDINATES OF THE KNOWN LANDMARKS.
 - (6) TO PROVIDE COORDINATES OF UNKNOWN LANDMARKS.
 - (7) TO TRACK A PRELOADED LANDING SITE.
 - (8) TO PROVIDE COORDINATES OF A NEW LANDING SITE (TREATED AS AN UNKNOWN LANDMARK)
 - (9) TO PROVIDE COORDINATES OF AN OFFSET LANDING SITE.
 - (10) TO POINT THE OPTICS ALONG AN ADVANCED ORBIT GROUND TRACK FOR THE PURPOSE OF TRACKING AND MAPPING A NEW LANDING SITE.
- ASSUMPTIONS:**
- (1) THERE ARE 25 KNOWN LUNAR LANDMARKS STORED IN FIXED MEMORY IN THE CMC AND ONE LANDMARK MAY BE STORED IN ERASABLE MEMORY IN THE CMC. THE LANDMARK STORED IN ERASABLE MEMORY IS REFERRED TO AS THE LANDING SITE AND IS DESIGNATED BY LANDMARK CODE NUMBER 01, THIS CODE IS AVAILABLE FOR LUNAR LANDMARK STORAGE ONLY.
 - (2) THERE ARE TWO TYPES OF LANDMARK TRACKING METHODS:
 - (A) "KNOWN" LANDMARK TRACKING - THE TRACKING OF AN EARTH LANDMARK MADE KNOWN TO THE CMC BY LATITUDE, LONGITUDE OVER 2, AND ALTITUDE AND THE TRACKING OF A LUNAR LANDMARK MADE KNOWN TO THE CMC BY ITS LANDMARK CODE NUMBER OR BY LATITUDE, LONGITUDE OVER 2, AND ALTITUDE.
 - (B) "(UNKNOWN)" LANDMARK TRACKING - THE TRACKING OF A LANDMARK OR SURFACE FEATURE IDENTIFIED TO THE CMC AS AN UNKNOWN LANDMARK, ONE WHOSE COORDINATES ARE NOT KNOWN.
 - (3) THERE ARE TWO TYPES OF LANDING SITE MAPPING METHODS, IN EITHER CASE THE LANDING SITE COORDINATES MAY BE STORED IN CMC MEMORY (REFER TO ASSUMPTION 1), THEY ARE:
 - (A) LANDING SITE DESIGNATION - TRACK AND MARK ON AN UNKNOWN LANDMARK. STORE THE RESULTING COORDINATES IN LANDMARK CODE 01. IF MAPPING ONLY IS DESIRED, I.E. NO STATE VECTOR CALCULATION OR CORRECTIONS, THE ASTRONAUT NEED TAKE ONLY ONE MARK.

++
+10
++
PCN
552

(B) LANDING SITE OFFSET - WHILE TRACKING AND MARKING ON A PRIMARY LANDMARK (KNOWN OR UNKNOWN), POINT THE OPTICS SLOS AT THE CHOSEN LANDING SITE AND MARK IT ONCE, (AT LEAST ONE MARK ON THE PRIMARY LANDMARK MUST HAVE BEEN MADE PRIOR TO THIS), THEN CONTINUE MARKING ON THE PRIMARY LANDMARK. STORE THE RESULTING COORDINATES OF THE OFFSET LANDING SITE IN LANDMARK CODE 01. THE ASTRONAUT HAS TWO WAYS OF DEFINING TO THE CMC WHICH MARK WAS MADE ON THE OFFSET LANDING SITE. THEY ARE:

1. KEY IN V52E AFTER MARKING ON THE OFFSET LANDING SITE. THIS WILL SET THE INDEX OF OFFSET DESIGNATOR EQUAL TO THE VALUE OF THE MARK COUNTER.
2. SIMPLY MARK ON THE OFFSET LANDING SITE BUT MAKE A MENTAL NOTE OF WHICH MARK IN THE SEQUENCE IT WAS AND THEN SET THE INDEX OF OFFSET DESIGNATOR TO THAT VALUE WHEN IT IS DISPLAYED FOLLOWING THE SIGHTING MARK ROUTINE.

(4) ACQUISITION OF A LANDMARK MAY BE AIDED BY THE CMC BY USE OF THE AUTOMATIC OPTICS POSITIONING ROUTINE (R52). CARE SHOULD BE EXERCISED WHEN AN UNKNOWN LANDMARK IS CHOSEN TO KEEP THE OPTICS OUT OF THE CMC MODE TO AVOID POSSIBLE PROGRAM ALARMS.

(5) ACQUISITION OF A PRELCADED LANDING SITE MAY BE AIDED BY KEYING LANDMARK CODE 01 INTO THE V05N70 DISPLAY FOR USE BY THE AUTOMATIC OPTICS POSITIONING ROUTINE (R52). TO IMPROVE THESE COORDINATES REFER TO ASSUMPTION 3.

(6) DURING LUNAR ORBIT WHILE IN THE CMC IDLING PROGRAM (P00) THE LUNAR LANDMARK SELECTION ROUTINE (R35) IS AVAILABLE TO AID THE CREW IN THE SELECTION OF APPROPRIATE LANDMARKS PRIOR TO THE SELECTION OF THIS PROGRAM.

(7) THE GROUND TRACK DETERMINATION PROGRAM (P21) IS AVAILABLE TO AID THE CREW IN CHOOSING APPROPRIATE LANDMARKS PRIOR TO SELECTION OF THIS PROGRAM.

(8) THE GROUND TRACK DETERMINATION PROGRAM (P21) IS AVAILABLE TO THE CREW FOLLOWING THIS PROGRAM TO PROVIDE UPDATED GROUND TRACK INFORMATION.

(9) POSSIBLE ATTITUDE CONTROL METHODS MIGHT BE AS FOLLOWS (IN ALL CASES CARE MUST BE TAKEN TO MONITOR POSSIBLE IMPENDING IMU GIMBAL LOCK).

(A) MANUAL CONTROL BY THE PILOT OR NAVIGATOR WITH THE ROTATIONAL HAND CONTROLLER.

(B) MANUAL RATE CONTROL BY THE NAVIGATOR WITH THE MINIMUM IMPULSE CONTROL IN THE GNC FREE MODE.

(10) THE PROGRAM MAY BE PERFORMED WITH SIVB ATTACHED IF THE LAUNCH VEHICLE GUIDANCE SWITCH IS PLACED IN THE CMC POSITION THEREBY PERMITTING SIVB ATTITUDE CONTROL WITH THE ROTATIONAL HAND CONTROLLER. GNC A/P CONTROL IS REQUIRED IN THIS CASE.

(11) THE IMU MUST BE ON AND ALIGNED IN ORDER TO COMPLETE THIS PROGRAM.

(12) SELECTION OF THIS PROGRAM WILL TERMINATE THE RENDEZVOUS NAVIGATION PROGRAM (P20).

(13) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

			. CREW PROG. . SELECTION . . .			
-----			-----			
START ORBITAL NAVIGATION PROGRAM (P22) DISPLAY PROGRAM 22	KEY IN ORBITAL NAVIGATION PROGRAM (P22) V37E22E			#10
	.		-----			
			.			
			.			
			.			
		MONITOR DSKY: OBSERVE DISPLAY OF PROGRAM 22			
-----			-----			
.						
.						
.						
-----			-----			
DO IMU STATUS CHECK ROUTINE (R02)	DO IMU STATUS CHECK ROUTINE (R02)			
	.		-----			
	.					
	.					
	.					
-----			-----			
RESET THE RENDEZVOUS FLAG						#30
-----			-----			
.						
.						
.						
-----			-----			
COMPUTE ANGLE BE- TWEEN Y AND V X R. -SM - -						#40
THIS ANGLE IS THE MAGNITUDE OF THE MAXIMUM POSSIBLE MIDDLE GIMBAL ANGLE ASSUMING THE SC X AXIS IS KEPT IN THE ORBIT PLANE.						
-----			-----			
.						#50
.						
.						
.						
.						
.						
.						

SELECT IMU
REALIGN (P52).
WHEN COMPLETE,
RESELECT THIS
PROGRAM.

EXIT P22

#110

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW PROGRAM

KEY IN PROCEED

.P .NEW
.R .PRCG
.O
.C
.E
.E EXIT
.D P22

CONTROL ATTITUDE
FOR LANDMARK
ACQUISITION

#120

"A"
SEE BELOW

#130

SET THE INDEX OF
OFFSET DESIGNATOR
EQUAL TO ZERO.

#140

RESET MARK DATA
VALID BIT FOR INFOR-
MATION TO DOWNLINK.

#150

"A"
SEE BELOW

IS THE SPACECRAFT
IN LUNAR ORBIT?

.N .Y
.
.
.
GO TO
"B"
BELOW

IS THE SPACECRAFT IN
LUNAR ORBIT?

.Y .N
.
.
.
GO TO
"B"
BELOW

#160

#170

LUNAR ORBIT

#180

POSS
HOLD
.....
SNAP

FLASH VERB NOUN TO
REQUEST RESPONSE AND
DISPLAY LMK CODE:
V05N70
R1-BLANK
R2 ABCDE
R3-BLANK

A=1 IF KNOWN LMK
A=2 IF UNKNOWN LMK.
B= INDEX OF OFFSET
DESIGNATOR
C= NOT USED IN THIS
PROGRAM
DE-LMK ID NO N

++
+11
++
PCN
594

NOTE: THE PROBABLE
OPTIONS ARE:
A=1. DE=00-
LANDMARK
KNOWN BUT
NOT STORED
IN CMC
MEMORY
A=1. DE>00-
LANDMARK
KNOWN AND
STORED

MONITOR DSKY:
OBSERVE VERB NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
LMK CODE.
TO HAVE THE AUTOMAT-
IC OPTICS POSITION-
ING ROUTINE (R52)
POINT ALONG THE
GROUND TRACK OF AN
ADVANCED ORBIT AND
60 DEGREES AHEAD OF
THE SPACECRAFT
CHANGE THE LANDMARK
CODE TO PX (WHERE P
>3 AND X DEFINES HOW
MANY ORBITS AHEAD
ARE DESIRED). THIS
CODE IS USED ONLY BY
THE AUTOMATIC OPTICS
POSITIONING ROUTINE
(R52) AT THIS TIME.

#190

#200

#210

++
+11
++
EDIT

A=2. DE NOT
MEANINGFUL:
UNKNOWN
LANDMARK.

DO I WISH TO NAVI-
GATE ON AN UNKNOWN
LANDMARK?

.Y .N

DO I DESIRE THE
USE OF THE AUTO-
MATIC OPTICS
POSITIONING
ROUTINE (R52)
TO AID IN ACQUI-
SITION?

.N .Y

DO I WISH TO
NAVIGATE ON
THIS LAND-
MARK?

.Y .N

SET OP-
TICS
MODE
SWITCH
TO MAN-
UAL.

WAIT FOR KEYBOARD
ENTRY

KEY IN
V22E AND
LOAD NEW
LMK CODE

#220

#230

#240

#250

#260

IS DE > 0?
(STORED LANDMARK).

.N .Y

"B"

POSS
HOLD .
.....
SNAP .

FLASH VERB-NOUN
TO REQUEST RESP-
ONSE AND DISPLAY
LMK COORDINATES:
V06 N89
R1-LAT
R2-LONG/2
R3-ALT

R1-LAT IS LATI-
TUDE OF LMK IN
DEG TO NEAREST
.001 DEGREE, +
IS NORTH

R2-LONG/2 IS
LONGITUDE OF
LMK DIVIDED BY
2. IN DEGREES
TO NEAREST .001
DEGREE, + IS
EAST

R3-ALT IS ALTI-
TUDE OF LMK
ABOVE THE FISCH-
ER ELLIPSOID
IN EARTH ORBIT
ABOVE THE MEAN
LUNAR RADIUS FOR
LUNAR ORBIT
IN NAUTICAL
MILES TO THE
NEAREST .01 NM.

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
LMK COORDINATES FOR
USE ONLY BY THE AUT-
OMATIC OPTICS POSIT-
IONING ROUTINE (R52)
AT THIS TIME.

DO I WISH TO NAVI-
GATE ON AN UNKNOWN
LANDMARK?

Y. .N

DO I DESIRE THE
USE OF THE AUTO-
MATIC OPTICS
POSITIONING ROU-
TINE (R52) TO AID
IN ACQUISITION?

.N .Y

SET OPTICS
MODE SWITCH
TO MANUAL.

#320

#330

#340

#350

#360

#370

WAIT FOR KEY-
BOARD ENTRY

AM I SATIS-
FIED WITH
THESE VALUES?

.Y .N

KEY IN PROCEED

KEY IN V25E AND
LOAD LMK PARAMETERS:

TERMINATE FLASH
UPON RECEIPT
OF PROCEED OR
NEW DATA

.N	.P
.E	.R
.W	.O
.D	.C
.A	.E
.T	.E
.A	.D

STORE
NEW
DATA

SET SIGHTING FLAG TO
LMK FOR USE BY
SIGHTING MARK AND
AUTO OPTICS POSITION-
ING ROUTINES.

#380

#390

#400

#410

#420

ZERO.

#480

MAKE SURE B RE-
MAINS UNCHANGED
IF A OR DE ARE
CHANGED.

#490

IS THE DATA IN R2
CORRECT FOR THIS
SIGHTING?
NOTE: IN EARTH ORBIT
DE MUST BE SET EQUAL
TO 00 BECAUSE LAND-
MARK CODE 01 IS
AVAILABLE FOR LUNAR
LANDMARKS ONLY.

#500

.Y .N

WAIT FOR KEYBOARD
ENTRY

KEY IN V22E AND
LOAD CORRECT
DATA.

#510

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA

KEY IN PROCEED

#520

.N .P
.E .R
.W .C
.D .C
.A .F
.T .F
.A .D

#530

109

P22/COLOSSUS

#540

#550

#560

#570

#580

P22/COLOSSUS

POSS
HOLD
.....
SNAP

FLASH VERB-NOUN
TO REQUEST RESP-
ONSE AND DISPLAY
LMK COORDINATES:
V06 N89

R1-LAT
R2=LONG/2
R3-ALT

R1=LAT IS LATI-
TUDE OF LMK IN
DEG TO NEAREST
.001 DEGREE,
+ IS NORTH

R2=LONG/2 IS
LONGITUDE OF LMK
DIVIDED BY 2.
IN DEGREES TO
NEAREST .001
DEGREE + IS EAST

R3=ALT IS ALTI-
TUDE OF LMK
ABOVE THE FIS-
CHER ELLIPSOID
FOR EARTH AND
MEAN LUNAR RADI-
US FOR MOON. IN
NAUTICAL MILES
TO THE NEAREST
.01 NM.

WAIT FOR KEY-
BOARD ENTRY

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
LMK COORDINATES

AM I SATISFIED WITH
THESE VALUES?

Y. N

KEY IN PROCEED

#590

#600

#610

#620

#630

III

P22/COLOSSUS

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR NEW
DATA

.N	.P
.E	.R
.W	.C
	.C
.D	.E
.A	.E
.T	.D
.A	

STORE NEW
DATA

SET THE DATA VALID
BIT FOR INFORMATION
TO DOWNLINK.

SET K = 0

SET J = 0

KEY IN V25E AND
LOAD LMK PAPAMETERS:

#640

#650

#660

#670

#680

++
+11
++
PCN
580

P22/COLOSSUS

SET J = J+1

IS J GREATER THAN
THE MARK COUNTER?

• N • Y

GET JTH MARK DATA
FROM CMC STORAGE

IS J = INDEX OF OFF-
SET DESIGNATOR?

•Y •N

STORE MARK
IN OFFSET
DESIGNATOR
LOCATION

IS THIS THE
FIRST MARK?

•Y •N

++
+11

112

+ 113

P22/COLOSSUS

+11 IS THE KNOWN FLAG
++ RESET?

#740

EDIT .Y .N

COMPUTE AND
STORE LMK
LOCATION FROM
FIRST MARK
DATA

#750

++
+11
++
EDIT

COMPUTE ORBIT PARA-
METERS AND LANDMARK
COORDINATE CHANGES
AS A RESULT OF THIS
SIGHTING MARK

#760

IS K = 0?

#770

.Y .N

POSS HOLD SNAP
FLASH VERB NOUN
TO REQUEST RES-
PONSE AND DIS-
PLAY ORBIT PARA-
METER CHANGES:

V06N49
R1-DELTA R
R2-DELTA V
R3-BLANK

DELTA R-MAGNI-
TUDE OF THE
DIFFERENCE BE-
TWEEN THE POSI-
TION VECTOR
BEFORE AND AFTER
INCORPORATION OF
THE LANDMARK

MONITOR DSKY: OB-
SERVE VERB NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF ORBIT PARAMETER
CHANGES

#780

ARE ORBIT PARAMETER
CHANGES ACCEPTABLE
FOR INSERTION INTO
CMC CALCULATION OF
POSITION AND VELO-
CITY?

#790

.Y .N

P22/COLOSSUS

SIGHTING DATA IN
N.M. TO NEAREST
0.1 N.M.

DELTA V-MAGNI-
TUDE OF THE
DIFFERENCE BE-
TWEEN THE VELO-
CITY VECTOR BE-
FORE AND AFTER
INCORPORATION OF
THE LANDMARK
SIGHTING DATA IN
FPS TO THE NEAR-
EST .1 FPS.

WAIT FOR KEY-
BOARD ENTRY

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR RE-
CYCLE.

.R	.P
.E	.R
.C	.D
.Y	.C
.C	.E
.L	.E
.E	.D

GO TO
"A"
ABOVE

SET K = 1

KEY IN PROCEED

KEY IN RECYCLE
V32E

GO TO
"A"
ABOVE

#800

#810

#820

#830

#840

UPDATE CMC STATE
VECTOR AND COMPUTE
REVISED LANDMARK
COORDINATES

#850

IS THERE DATA IN THE
OFFSET DESIGNATOR
LOCATION?

#B60

• N • Y

- MAP THE OFFSET
- LANDING SITE

#870

DID I DESIGNATE AN
OFFSET LANDING SITE
FROM A PRIMARY LAND-
MARK BY USE OF V52E
FOLLOWING A MARK?

4880

.YES **.NO**

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY POTENTIAL
LANDING SITE COORDI-
NATES

MONITOR DSKY:
OBSERVE VERB-
NOUN FLASH TO
REQUEST RES-
PONSE AND DIS-
PLAY OF THE
LANDMARK COOR-
DINATES ON
WHICH I WAS
MARKING.

#890

POS S
HOLD .
.....
SNAP .

V06 N89
R1-LAT
R2-LONG/2
R3-ALT

R1=LAT IS LATITUDE
OF LMK IN DEG TO
NEAREST .001 DEGREE
+ IS NORTH

R2=LONG/2 IS LONGI-
TUDE OF LMK DIVIDED
BY 2. IN DEGREES TO
NEAREST .001 DEGREE
+ IS EAST

R3=ALT IS ALTITUDE
OF LMK ABOVE THE
FISCHER ELLIPSOID
FOR EARTH AND MEAN
LUNAR RADIUS FOR
MOON. IN NAUTICAL
MILES TO THE NEAREST
.01 NM.

.
.
.

WAIT FOR KEY-
BOARD ENTRY

MONITOR DSKY:
OBSERVE VERB-
NOUN FLASH TO
..... REQUEST RESPONSE
AND DISPLAY OF
THE OFFSET LAND-
ING SITE COORDI-
NATES.

#910

#920

#930

SHALL THESE COORDIN-
ATES BE STORED IN
CMC MEMORY AND BE
DEFINED AS THE LAND-
ING SITE (LANDMARK
CODE 01)?

#940

NOTE1: IF THESE CO-
ORDINATES ARE STOR-
ED, THE PREVIOUSLY
STORED COORDINATES
OF THE LANDING SITE
WILL BE ERASED. IF
THEY ARE NOT STORED
THE PREVIOUSLY
STORED LANDING SITE
COORDINATES WILL BE
SAVED IN THE LAND-
MARK CODE 01
LOCATION.

#950

NOTE2: THE CAPABIL-
ITY OF STORING THE
LANDING SITE IS ONLY
AVAILABLE FOR LUNAR
LANDMARKS. FOR THIS
REASON THE ASTRONAUT

SHOULD ALWAYS KEY
EITHER V32E OR V34E
ON THIS DISPLAY
WHEN IN EARTH ORBIT.

#960

.YES .NO

.
.

KEY IN PROCEED

.
.
.

KEY IN RECYCLE
V32E

.
.
.
...

GO TO
"A"
ABOVE

GO TO
"A"
ABOVE

#970

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR
RECYCLE

.R .P
.E .R
.C .C
.Y .C
.C .E
.L .E
.E .D
.
.
...
.
GO TO
"A"
ABOVE

#980

STORE THESE COORDIN-
ATES AS UPDATED
LANDING SITE
PARAMETERS.

.
.
.
...
.
GO TO
"A"
ABOVE

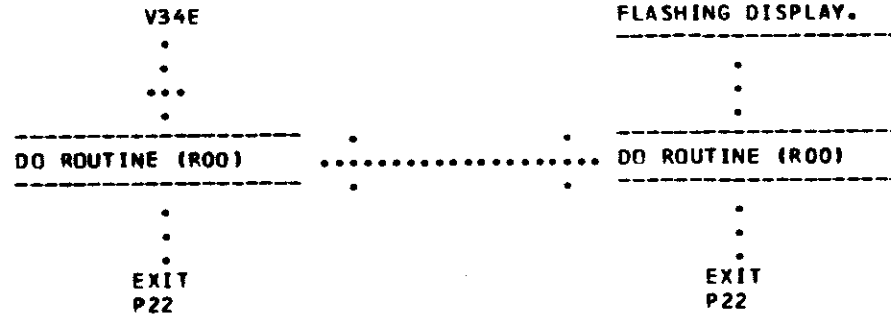
#990

#1000

V34E

.
.
.
.

#1010



#1020

#1030

CHANGE CONTROL NOTES

LOGIC REV 07 PCR MIT 66
 LOGIC REV 08 PCR MIT 116
 PCR MIT 83
 LOGIC REV 09 PCR 206
 LOGIC REV 10 PCN 552
 LOGIC REV 11 PCN 580
 PCN 594

CISLUNAR MIDCOURSE NAVIGATION PROGRAM (P23)

LCGIC REV 14 11/27/68

PURPOSE: (1) TO DO MIDCOURSE NAVIGATION BY INCORPORATION OF STAR/EARTH AND STAR/MOON OPTICAL MEASUREMENTS.

ASSUMPTIONS: (1) THIS PROGRAM DOES NOT REQUIRE THAT THE IMU BE ON.

(2) IF THE IMU IS NOT ALIGNED THE ASTRONAUT MUST ACQUIRE THE STAR/LMK OR STAR/HOR MANUALLY.

(3) (A) IF THE IMU IS ALIGNED THE ASTRONAUT MAY ACQUIRE THE LMK/HOR AUTOMATICALLY.

(B) IF THE IMU IS ALIGNED THE ASTRONAUT MAY ACQUIRE THE STAR AUTOMATICALLY.

(C) IF THE IMU IS ON THE ASTRONAUT MUST TAKE APPROPRIATE PRECAUTIONS TO PREVENT POSSIBLE IMU GIMBAL LOCK.

(4) PRIOR TO EACH MARK THE PROGRAM WILL CALL FOR AN OPTICS CALIBRATION WHICH MAY BE DONE OR BYPASSED DEPENDENT UPON THE STABILITY HISTORY OF THE CALIBRATION (SEE PURPOSES AND ASSUMPTIONS OF R57).

(5) TO PERFORM THE MARK THE ASTRONAUT SHOULD FINALLY SELECT MINIMUM IMPULSE CONTROL (EITHER GNCS OR SCS) AND THE OPTICS SHOULD BE IN MANUAL IN ORDER TO MAINTAIN THE FIX.

(6) THE OPTICS SHOULD BE ON FOR 15 MINUTES PRIOR TO MARKING.

(7) THE CMC DOES NOT CHECK FOR MOON/EARTH OCCULTATION OR SUN BRIGHTNESS IN THIS PROGRAM

(8) THIS PROGRAM IS DESIGNED FOR ONE MAN OPERATION WITHIN THE CONSTRAINTS OF MODE SWITCHING WHILE IN THE LEB.

++ (9) THE SIGHTING IS ON THE BODY FOR WHICH THE STATE VECTOR IS DEFINED.

+14 (10) A STAR SERIAL 0 WILL GIVE PARITY FAILURE, AND LANDMARK SERIAL 01 WILL NOT WORK.

++ (11) THE PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

EDIT

PROG CONT	CMC	GROUND	CREW	CHECKLIST	TIME	TOTAL TIME
--------------	-----	--------	------	-----------	------	---------------

.CREW PROG.
.SELECTION

...

.

.

.

.

.

.

.

.

.

.

P23/COLOSSUS

 START CISELUNAR MID-
 COURSE NAVIGATION
 PROGRAM (P23)
 DISPLAY P23

 KEY IN CISELUNAR MID-
 COURSE NAVIGATION
 PROGRAM (P23)
 V37E 23E

#10

 MONITOR DSKY:
 OBSERVE DISPLAY OF
 PROGRAM 23

#20

 SET MARK INDEX TO 1
 FOR USE BY THE
 SIGHTING MARK
 ROUTINE R53

#30

 RESET RENDEZVOUS FLAG

 SET TARGET FLAG TO
 STAR FOR USE BY THE
 SIGHTING MARK ROU-
 TINE (R53) AND BY
 THE AUTO OPTICS
 POSITIONING ROUTINE
 (R52)

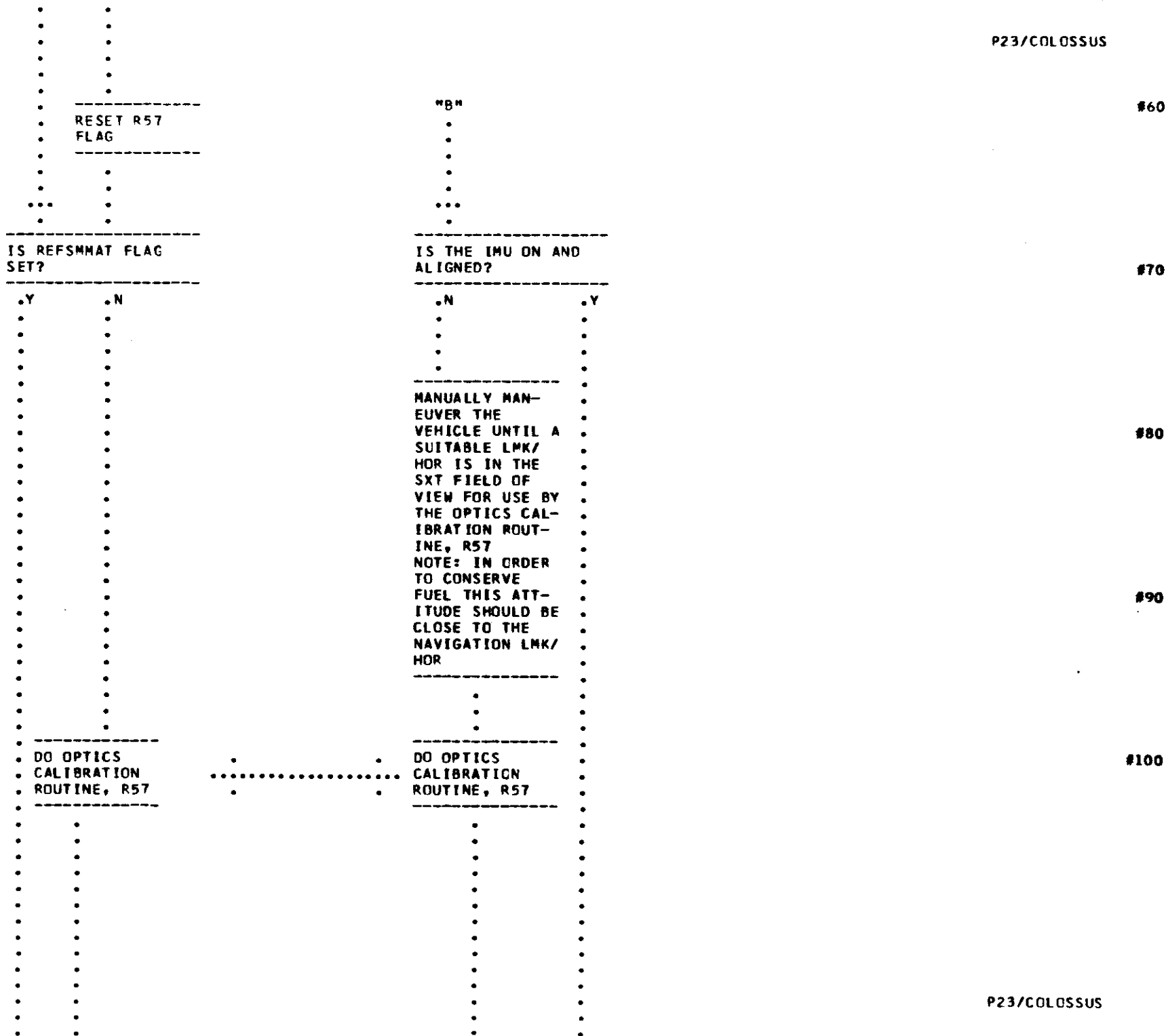
#40

 RESET V94 FLAG

#50

"B"
 .
 .
 .
 .
 .
 .
 .
 .
 .

++
 +13
 ++
 EDIT
 PCR
 206



DO SIGHTING
MARK ROUTINE,
(R53).

GO TO
"A"
BELOW

DO THE SIGHTING
MARK ROUTINE,
R53.
MANUALLY MAN-
EUVER THE
VEHICLE UNTIL A
SUITABLE LMK/
HOR IS IN THE
SXT FIELD OF
VIEW. THEN,
BY USE OF MIN-
IMUM IMPULSE
CONTROL (GNCS
OR SCS) AC-
QUIRE THE LMK
AND BY USE OF
OPTICS CONTROL-
LER SUPERIMPOSE
THE NAVIGATION
STAR PER CHECK-
LIST.

GO TO
"A"
BELOW

#110

#120

#130

#140

#150

POSS
HOLD .
.....
SNAP .

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY MEASUREMENT
IDENTIFICATION
V05 N70
R1-000DE (STAR ID)
R2-ABCOE (LMK ID)
R3-00CDO (HOR ID)

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF MEASUREMENT
IDENTIFICATION

STAR IDENTIFICATION-
R1=000XX FOR ID
NUM. OF STAR
USED

LANDMARK IDENTIFI-
CATION
R2=ABCDE WHERE

A AND B-NOT
USED

C=1 FOR EARTH
LMK

C=2 FOR MOON
LMK

DE-LANDMARK ID
NUMBER

R2=00000 FOR
HOR. MEAS.

HORIZON IDENTIFI-
CATION
R3=00CDO WHERE

C=1 FOR EARTH
HOR.

C=2 FOR MOON
HOR.

D=1 FOR NEAR
HORIZON

D=2 FOR FAR
HORIZON

R3=00000 FOR
LMK MEAS.

DO I INTEND TO DO
THE SIGHTING
MANUALLY?
THIS DATA WILL BE
USED AT THIS TIME
FOR THE AUTOMATIC
TRACKING MANEUVER
AND BY THE AUTO-
MATIC OPTICS
POSITIONING ROUTINE
(R52) ONLY. IF IT IS
INTENDED TO DO MAN-
UAL ACQUISITION THIS
DISPLAY NEED NOT BE
REVIEWED. IT WILL BE
REDISPLAYED PRIOR TO
STATE VECTOR
CALCULATION.

.Y .N

SET OPTICS MODE
SWITCH TO MANUAL

AM I SATISFIED
WITH THIS DATA?

.Y .N

RECORD MEAS-
UREMENT IDEN-
TIFICATION
DATA IN
FLIGHT DATA
BOOK IF
DESIRED

#160

#170

#180

#190

#200

#210

WAIT FOR KEYBOARD
ENTRY

KEY IN
PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCEED
OR NEW DATA

KEY IN V25E AND
LOAD NEW DATA

#220

.P .NEW
.R .DATA
.O .
.C
.E STORE NEW
.E DATA
.D

#230

WAS R3=0
(LMK MEASUREMENT?)

#240

.N(HOR) .Y(LMK)
.Y
ARE R2(DE) NON-
ZERO (LMK IDENT-
IFIED)

#250

.Y .N

#260

POSS
HOLD .
.....
SNAP .

FLASH VERB-NCUN
TO REQUEST RE-
SPONSE AND DIS-
PLAY LMK DATA
V06 N89

R1: LAT
R2: LONG/2
R3: ALT

LAT:-LATITUDE
OF LMK IN DEG
TO NEAREST .001
DEG

+ IS NORTH

LONG/2-LONGI-
TUDE OF LMK DI-
VIDED BY 2 IN
DEG TO NEAREST
.001 DEG

+ IS EAST

ALT-ALT OF LMK
IN NAUTICAL
MILES TO NEAR-
EST .01 N.M.
(FOR EARTH
ABOVE FISCHER
ELLIPSOID) (FOR
MOON ABOVE MEAN
LUNAR RADIUS)

WAIT FOR KEY-
BOARD ENTRY

MONITOR DSKY:
OBSERVE VERB-
NOUN FLASH TO
REQUEST RESPONSE
AND DISPLAY OF
LMK DATA IF A
NON-IDENTIFIED
LANDMARK WAS
CHOSEN

AM I SATISFIED
WITH THIS DATA?

.Y .N

KEY IN
PROCEED

#270

#280

#290

#300

#310

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR NEW
DATA

.P .NEW
.R .DATA
.O
.C
.E STORE NEW
.E DATA
.D

KEY IN
V25E
AND
LOAD
NEW
DATA

#320

#330

POSS
HOLD . FLASH VERB-NOUN TO
..... REQUEST PLEASE
SNAP . PERFORM GNCS AUTO-
MATIC MANEUVER
V50 N25

R1-00202
R2-BLANK
R3-BLANK

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH REQUESTING
PLEASE PERFORM AUTO-
MATIC MANEUVER

#340

#350

DO I WISH TO HAVE
THE GNCS COMPUTE THE
SPACECRAFT ATTITUDE
REQUIRED TO POINT
THE LLOS AT THE
CHOSEN LMK/HOR AND
PERFORM THE ATTITUDE
MANEUVER ROUTINE
(R60)?
THIS ATTITUDE WILL
BE COMPUTED (VEC-
POINT) TO POINT THE
LLOS AXIS AT THE
LMK/HOR BUT WILL NOT
CONSTRAIN THE ORIEN-
TATION ABOUT THAT
VECTOR (THE ORIEN-
TATION ABOUT THAT
VECTOR COULD RESULT

#360

.....

• N	• Y
•	•
•	•

KEY IN
ENTER

• •

• •

• •

•E	•P
•N	•R
•T	•O
•E	•C
•R	•F

- P
- R
- Q
- C
- E
- E
- D

३५४

- INTEGRATE CSM
- STATE VECTOR TO
- THE PRESENT
- TIME.

MANUALLY
MANEUVER THE
VEHICLE
UNTIL A
SUITABLE LMK
/HOR IS IN
THE SXT
FIELD OF
VIEW FOR USE
BY THE
OPTICS CAL-
IBRATION
ROUTINE, R57
NOTE: IN
ORDER TO
CONSERVE
FUEL THIS
ATTITUDE
SHOULD BE
CLOSE TO THE
NAVIGATION
LMK/HOR

#400

++
+13
+
+
+
+
+
+13
++
520+
EDIT

#420

#430

DO THE ATTITUDE
MANEUVER ROUTINE
(R60)

DO THE
ATTITUDE
MANEUVER
ROUTINE (R60)

#440

IS R57 FLAG SET?

.N .Y

DO R57 OPTICS
CALIBRATION
ROUTINE

DO R57 OPTICS CALI-
BRATION ROUTINE
(WHEN LMK ACQUIRED)

#450

SET V94 FLAG

SET R57 FLAG

#460

DO THE AUTOMATIC
OPTICS POSITIONING
ROUTINE (R52) (CALLS
R53)

DO R52 AUTOMATIC
OPTICS POSITIONING
ROUTINE. THE OPTICS
POINTING PROCESS IS
NOW AUTOMATIC AND
WILL POINT THE SXT
STAR LOS AT THE
CHOSEN STAR UNTIL
THE SIGHTING MARK
ROUTINE (R53) IS

#470

.NORMAL .PREMATURE
.EXIT .EXIT VIA
.V04E

DO V94 LOGIC (AS

++
+14
++
PCN
571

++
+13
++
++
++
++

179 +

EDIT
PCR
206

P23/COLOSSUS

#480

#490

#500

#510

#520

DEFINED IN EXTEN-
DED VERB SECTION
OF THIS DOCUMENT)

GO TO
"C"
ABOVE

RESET V94 FLAG

RESET R57 FLAG

"A"

CALLED AUTOMATICALLY
BY SWITCHING OPTICS
MODE TO MANUAL.

IF THE ASTRONAUT DE-
SIRES TO HAVE THE
CMC AID IN REACQUIR-
ING THE LANDMARK/
HORIZON HE MAY KEY
IN V94E UNTIL HE HAS
COMPLETED THE SIGHT-
ING MARK ROUTINE
(R53).

PRIOR TO ACTUAL MARK
THE ASTRONAUT SHOULD
SELECT A MINIMUM
CONTROL MODE (EITHER
GNCS OR SCS) IN
ORDER TO MAINTAIN
THE LMK/HOR AND STAR
FIXED IN THE SXT
FIELD OF VIEW.

"A"

MONITOR DSKY:
OBSERVE VERB-NOUN
FLASH TO REQUEST
RESPONSE AND DISPLAY
OF MEASUREMENT
IDENTIFICATION

FLASH VERB-NOUN TO
REQUEST RESPONSE AND
DISPLAY MEASUREMENT
IDENTIFICATION

V05 N71
R1-000DE (STAR ID)
R2-ARCCE (LMK ID)
R3-00CDO (HOR ID)

HOLD .
.....
SNAP .

P23/COLOSSUS

STAR IDENTIFICATION-
R1=000XX FOR ID
NUM. OF STAR
USED

#530

LANDMARK IDENTIFI-
CATION
R2=ABCDE WHERE

IS THIS DATA CORRECT
FOR THE MARK I MADE?

.Y .N

A AND B-NOT
USED

#540

C=1 FOR EARTH
LMK

C=2 FOR MOON
LMK

DE-LANDMARK ID
NUMBER

R2=00000 FOR
HOR. MEAS.

#550

HORIZON IDENTIFI-
CATION
R3=00CDO WHERE

C=1 FOR EARTH
HOR.

C=2 FOR MOON
HOR.

D=1 FOR NEAR
HORIZON

D=2 FOR FAR
HORIZON

R3=00000 FOR
LMK MEAS.

#560

RECORD MEASURE-
MENT IDENTIFIC-
ATION DATA IN
FLIGHT DATA
BOOK IF DESIRED

#570

WAIT FOR KEYBOARD
ENTRY

KEY IN
PROCEED

TERMINATE FLASH UPON
RECEIPT OF PROCFED
OR NEW DATA

KEY IN V25E AND
LOAD NEW DATA

#580

.P .NEW
.R .DATA
.D .
.C
.E STORE NEW
.E DATA
.D

#590

WAS R3=0
(LMK MEASUREMENT?)

#600

.N(HOR) .Y(LMK)

ARE R2(DE) NON-
ZERO (LMK IDENT-
IFIED)

#610

.Y .N

#620

#630

POSS
HOLD .
.....
SNAP .

FLASH VERB-NOUN
TO REQUEST RE-
SPONSE AND DIS-
PLAY LMK DATA
V06 N89

R1: LAT
R2: LONG/2
R3: ALT

LAT:-LATITUDE
OF LMK IN DEG
TO NEAREST .001
DEG

+ IS NORTH

LONG/2-LONGI-
TUE OF LMK DI-
VIDED BY 2 IN
DEG TO NEAREST
.001 DEG

+ IS EAST

ALT-ALT OF LMK
IN NAUTICAL
MILES TO NEAR-
EST .01 N.M.
(FOR EARTH
ABOVE FISCHER
ELLIPSCID) (FOR
MOON ABOVE MEAN
LUNAR RADIUS)

WAIT FOR KEY-
BOARD ENTRY

TERMINATE FLASH
UPON RECEIPT OF
PROCEED OR NEW
DATA

.P .NEW
.R .DATA
.C .
.C -----
.E STORE NEW
.E DATA

MONITOR DSKY:
OBSERVE VERB-
NOUN FLASH TO
REQUEST RESPONSE
AND DISPLAY OF
LMK DATA IF A
NON-IDENTIFIED
LANDMARK WAS
CHOSEN

ARE THESE THE
COORDINATES OF
THE LANDMARK I
MARKED ON?

.Y .N

KEY IN
PROCEED

KEY IN
V25E
AND
LOAD
NEW
DATA

#640

#650

#660

#670

#680

ARE ORBITAL PARAMETER CHANGES ACCEPTABLE FOR INSERTION INTO CMC CALCULATIONS OF POSITION AND VELOCITY?

#750

WAIT FOR KEYBOARD
ENTRY

KEY IN PROCEED

#760

TERMINATE FLASH ON
RECEIPT OF PROCEED.
OR RECYCLE

KEY IN
RECYCLE
V32E

#770

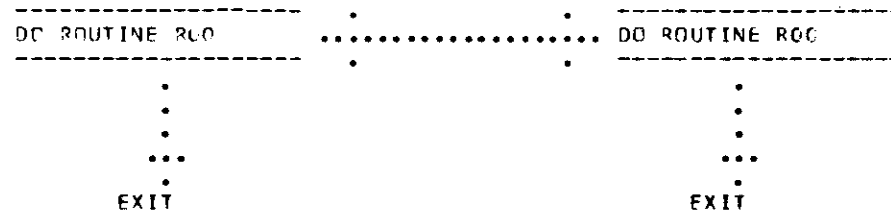
.P	.R
.R	.E
.O	.C
.C	.Y
.E	.C
.E	.L
.D	.E
.	.
.	.
.	.
.	.
.	GC TO
.	"B"
.	ABOVE
.	
.	

GO TO
"B"
ABOVE

#780

UPDATE CMC
STATE VECTOR

135



P23/COLOSSUS

#790

#800

CHANGE CONTROL NOTES

REV 10 PCR NASA 84
REV 11 PCR MIT 66
REV 12 PCR 206
REV 13 PCN 520
REV 14 PCN 571
PCR 206 EDITORIAL

THIS PAGE INTENTIONALLY LEFT BLANK

)

)

)

PURPOSE: (1) TO INSERT INFORMATION INTO THE CMC VIA THE DIGITAL UPLINK BY TRANSMISSION FROM THE GROUND OR VIA THE DSKY KEYBOARD BY CREW MANUAL INPUT.

ASSUMPTIONS: (1) THE OMC MUST BE IN THE OPERATE CONDITION. THE IMU MAY BE IN STANDBY OR OPERATE CONDITION.

(2) CMC UPDATES ARE OF FOUR CATEGORIES:

(A) PROVIDE AN UPDATE FOR CMC LIFTOFF TIME (V70).

(8) PROVIDE AN OCTAL INCREMENT FOR THE CMC CLOCK ONLY (V73).

(C) PROVIDE LOAD CAPABILITY FOR A BLOCK OF SEQUENTIAL FRASABLE LOCATIONS (1-18 INCLUSIVE LOCATIONS WHOSE ADDRESS IS SPECIFIED) (V71).

(D) PROVIDE LOAD CAPABILITY FOR 1-9 INCLUSIVE INDIVIDUALLY SPECIFIED ERASABLE LOCATIONS (V72).

(3) A COMPLETE DESCRIPTION OF THE CMC UPLINK FORMAT IS INCLUDED IN SECTION 2 OF R-577.

(4) UPDATE IS ALLOWED IN THE CSM WHEN THE CMC IS IN P00 OR P02, AND IF THE DSKY IS AVAILABLE.

(5) THE UPTCL ACCEPT/BLOCK SWITCH MUST BE IN ACCEPT FOR TELEMETRY UPDATE.

(6) THE PROGRAM IS MANUALLY SELECTED BY THE ASTRONAUT BY DSKY ENTRY OR BY THE GROUND BY UPLINK TRANSMISSION.

(7) THE AUTOMATIC MODE OF UPDATE IS PROGRAM SELECTION AND UPDATE VIA THE GROUND BY UPLINK TRANSMISSION. THE ONLY DIFFERENCE BETWEEN THIS AND MANUAL SELECTION BY THE ASTRONAUT IS THAT THE DSKY RESPONSES ARE KEYED IN BY THE ASTRONAUT RATHER THAN TRANSMITTED.

PROG
CNT

CMC

GROUND

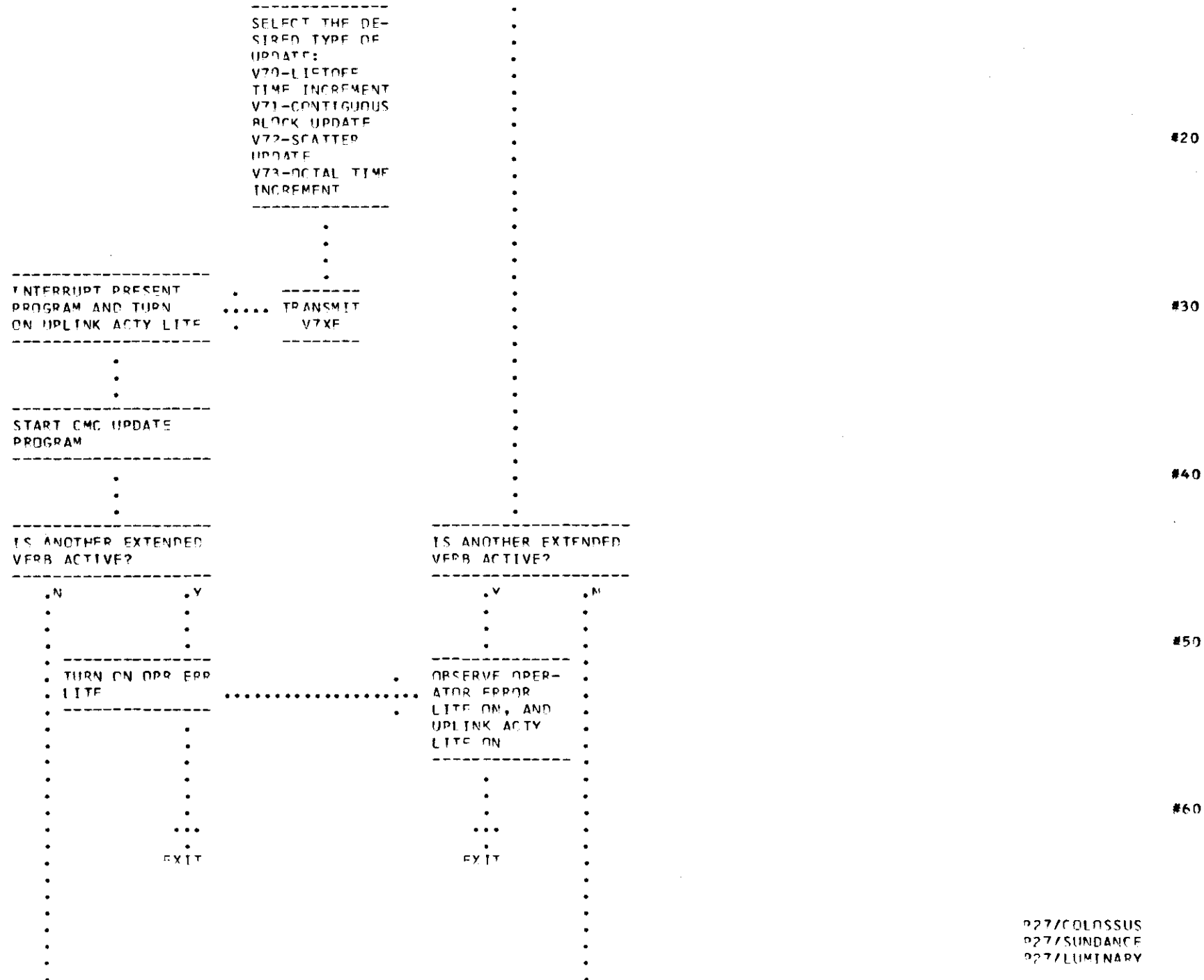
CPFW

CHECKLIST

TIME

TOTAL
TIME

NOTIFY CREW OF INTENTION TO UPDATE. SPECIFY PARAMETERS AND COORDINATE TIME. VERIFY THAT UPDATE WILL BE SATISFACTORY W.R.T. TIME, WORK IN PROGRESS AND CMC ACTIVITY.



IS PROGRAM 000 OR
0022

.Y .N

TURN ON OPR ERR
LITE. TURN OFF
UPLINK ACTY
LITE

EXIT

SELECT UPDATE PRO-
GRAM (P27) DOWNLIST
FOR DOWNLINK TRANS-
MISSION

DISPLAY PROGRAM 27

IS THIS A TIME IN-
CREMENT UPDATE
(V70 OR V73)?

.N .Y

"N"
FROM
BELOW

SET COMPNUMB
EQUAL TO 2

MONITOR DSKY:
IS PROGRAM 000 OR
0022

.N .Y

OBSEVE OPR
ERR LITE ON
AND UPLINK
ACTY LITE OFF.

EXIT

MONITOR DSKY:
OBSEVE DISPLAY
OF PROGRAM 27

GO TO
"N"
BELOW

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

#70

#80

#90

#100

#110

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

```

FLASH VERB/NOUN TO
REQUEST LOAD OF
INDEX IN MACHINE
ADDRESS SPECIFIED IN
R3 AND DISPLAY:
POSS HOLD . V21 N01 .....
..... R1-BLANK
SNAP . R2-UNCHANGED
R3-AAAAA

```

```

AAAAA-MEMORY LOCA-
TION IN WHICH THE
INDEX VALUE WILL BE
LOADED. THE INDEX
VALUE REPRESENTS THE
TOTAL NUMBER OF
NUMERIC VALUES TO BE
LOADED, INCLUDING
THE INDEX VALUE IT-
SELF. MINIMUM INDEX
IS 3; MAXIMUM IS 20.

```

```

WAIT FOR KEYBOARD
ENTRY

```

```

MONITOR DOWN-
LINK:
OBSERVE VERB/
NOUN FLASH AND
DISPLAY OF
AAAAA

```

```

DO I WISH TO
TERMINATE?

```

```

.Y .N

```

```

TRANSMIT
TERMINATE
V24F

```

```

GO TO "A"
BELOW

```

```

TRANSMIT
INDEX
VALUE

```

#120

#130

#140

#150

#160

TERMINATE FLASH UPON
RECEIPT OF TERMINATE
OR INDEX

.T	.I
.F	.N
.R	.D
.M	.E
.I	.X
.N	.
.A	.
.T	.
.E	.
.	.
.	.
.	.

GO TO "A" BELOW	----- DISPLAY INDEX VALUE IN R1 AS IT IS LOADED -----
--------------------	---

.

++
+07
+07
++
EDIT

IS INDEX LESS THAN
3 OR GREATER THAN
20?

.Y	.N
.	.
.	.
.	.
.	.

GO TO "B"
ABOVE

STORE INDEX
IN COMPNUM

.....
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.
.	.	.

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

#170

#180

#190

#200

#210

#220

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

CALCULATE ADDRESS
FOR STORAGE OF NEXT
DATA LOAD

"CM"
FROM
BELOW

#230

FLASH VERB/NOUN TO
REQUEST LOAD OF DATA
INTO CALCULATED
ADDRESS SPECIFIED IN...
R3 AND DISPLAY: MONITOR DOWN-
 LINK:
 OBSERVE VERB/
 NOUN FLASH TO
 REQUEST DATA
 LOAD

#240

HOLD .. V21 NO1
..... R1 BLANK
SNAP .. R2 UNCHANGED
 R3 AAAAA

DO I WISH TO
TERMINATE?

#250

WAIT FOR KEYBOARD
ENTRY

.Y .N

TRANSMIT
TERMINATE
V34F

#260

GO TO "A"
BELOW

#270

TRANSMIT
DATA

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

TERMINATE FLASH
UPON RECEIPT OF
TERMINATE OP DATA

.T .D
.E .A
.R .T
.M .A
.I .
.N .

.A
.T DISPLAY DATA
.E IN R1 AS IT
. IS LOADED

GO TO "AW"
BELOW

IS THIS THE LAST
DATA LOAD? (UPLINK
COMPONENT COUNTER =
COMPNUMB)

.N .Y

"DW"
FROM
BELOW

FLASH VERB/NOUN TO
REQUEST LOAD OF
OCTAL IDENTIFIER
INTO MACHINE ADDRESS...
SPECIFIED IN R3 AND
DISPLAY:
V21 N02
R1 BLANK
R2 UNCHANGED
R3 AAAAA

MONITOR DOWN-
LINK:
. OBSERVE VERB/
NOUN FLASH TO
. REQUEST LOAD
OF OCTAL
IDENTIFIER

#280

#290

#300

#310

#320

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

OCTAL IDENTIFIER -
IDENTIFIES WHICH OF
THE TRANSMITTED
WORDS WILL BE
CORRECTED.

#330

.
.

WAIT FOR KEYBOARD
ENTRY

DO I WISH TO
TERMINATE?

.Y .N

#340

DO I WISH
TO COR-
RECT ANY
DATA?

.Y .N

#350

.
.....
.

TRANSMIT
TERMIN-
ATE V34E

GO TO
"A"
BELOW

#360

.
.....
.

TRANSMIT
OCTAL
IDENTIF-
IER

#370

.
.....
.

TRANSMIT
PROCEED

#380

IS THIS AN OCTAL
TIME INCREMENT
(V73)?

#440

.Y .N
.
.
.
.
.

WOULD THE INCRE-
MENT CAUSE THE
CMC CLOCK TO
OVERFLOW?

#450

.Y .N
.
.

TURN ON
OPERATOR
ERROR LITE

MONITOR DSKY:
OBSERVE OPER-
ATOR LITE

#460

GO TO "A"
BELOW

INCREMENT
CMC CLOCK

GO TO "A"
BELOW

#470

GO TO "A"
BELOW

#480

IS ORBITAL INTEGRA-
TION IN PROGRESS?

.Y .N
.
.
.
.
.
.

- STALL
- INTEGRATION
- UNTIL DATA
- TRANSFER

IS THIS A LIFTOFF
TIME UPDATE
(V70)?

WOULD UPDATE
CAUSE THE CMC
CLOCK TO OVER-
FLOW?

TURN ON OP-
ERATOR ERR-
OR LITE

GO TO "A"
BELOW

MONITOR DSKY:
OBSERVE OPERA-
TOR ERROR LITE

GO TO "A"
BELOW

P27/COLOSSUS
P27/SUNDANCE
P27/LUMINARY

#490

#500

#510

4520

#530

#540

P27/CLOSSUS
P27/SUNDANCE
P27/LUMINEM

MONITOR DSKY:
OBSERVE OPERA-
TOR EPPROP LITE

#590

TRANSFER DATA TO
SPECIFIED BLOCK
(V71) OR SPECIFIED
ADDRESSES (V72)

HAW
 FROM
 ABOVE

WALL
FROM
ABOVE

WAW
FROM
ABOVE

TURN OFF UPTL ACT-
IVITY LITE

```

....MONITOR DOWN-
. LINK:
  OBSERVE UPTL
  ACTIVITY LITE
  OUT, REVERSION
  TO ORIGINAL
. DOWNLINK LIST,
....TERMINATION OF
. P27, AND RE-
  TURN TO P00 OR
  P02.

```

MONITOR DSKY:
SUCCESSFUL UPDATE IS
INDICATED BY UPTL
ACTIVITY LITE OUT,
OPERATOR ERROR LITE
OUT, AND RETURN TO
P00 OR P02.

CHANGE DOWNLINK LIST
TO ORIGINAL

TERMINATE P27 AND GO
TO PROGRAM WHICH WAS
INTERRUPTED (P00 OR
P02)

EXIT P27

CHANGE CONTROL NOTES

REV 05 PCP MIT 66
REV 06 PCP 206
REV 07 EDITORIAL

THIS PAGE INTENTIONALLY LEFT BLANK

)

)

)