Fecent auden by HSC, MT, Bell Comm, and MM have defined the relation characteristics ot the present MTF-Raytheon Commuter and
 action being proposed by Headquarters now is to develop the lt Computer so there it can be used for the Spacecraft also. In evaluathe its action further, some additional information needs to be developed and come decisions or ground rules established. These are disctassed as follows:
a. To what extent will Spacecraft modification be acceptable in acommodrang the now computer at bone future the? The mooned int Computer when not in the present space allocated in the wacecraft. The Mri Computer dits, but must have more space tor mares store are This is mot easier than locating en other active back bot. bo we wait to redesign the Spacecraft now to make more row in one place?
b. In planing for a possible future change in Computers, wis. it. be parlssmle to write all net nroprame, or should it bo a mavis rus that no program changes are required. the logical fokine of the IE A manixe is different to the extent that complete re-orogrming would be required. ito preserve ATT ;rograming with Th a mold require a compete, new Computer design.
c. In resolving $A S C-f S B$ interface problems, how different win the two versions of the TH R Computer be permitted to grow? The problems mentioned above can be solved by building two different Th t Comport, but this is not a are backup" iron a vasa standpoint. On the other hand, agreeing on instruction code, word length, intersupt sequence, bic, would be a monomial task and require a new Computer cosign. The effort so far with Gel Com, is honing ow ont Damper man away from the eutoting grogram almost sompletely. The degree to which the logic is identical also affects the domed to which program can be ideation, Can the mechanical packagha be datrerent in the two applications, or must this be negotiated also?
d. That mev the relative administrative, management and technical responsibilities or MSC and MSFC in this matter? tho pays for what? How do we decide?


In thinking over how we might resolve these problens, I have formed the following opinions:
3. We should proceed with the wrort to mate the TR combitibie with the Spacecrait needs with the understanding that it shouid be pocsible to use min at some future date, but not necessarily convenient.
b. Plan no change to existing Spacecraft to accomodate the TMR Computer. The Tht computer would be designed to fit in the existing hole plus a spill-over box that would be mounted in some other location with a minimm of re-design at the time the Trit was installed.
c. Plan to re-program when TR is used in the Spacecraft. This is a big eifort, but the only practical solution.
d. It shouid be a requirenent that only one basic Tif Computer be developed. The basic Computer should fit the hole in the Cormand Module and this next should be common with Saturn. The "spili-over box: cain be mique for the Spacecraft application, and other boxes can be added for the Saturn application. The IRy version should be derived from some simple (smaller) modification of the basic Computer.

With this set of ruies, the major cost to MSC should consist of the re-programing efiort, The ThR desime effort could proceed alnost. indoondentiy. We wouldn't plan to do aray more until we really fot serious about changing over to ThR. If, however, we plen to do more than this, or deviate in some manner from these rules, the amount of coordination required between MSG and liSFg could inerease rapidly and could keep an additional four or five peonle busy iull timo, guita easily. This doesn't sound like much oi e problem, but the existing personiel problem is already grim. We should have one rull-time man assigned to coordinato design data, in any event. If you have any firm opinions on direction in these matters, $\rightarrow$ appreciate discussing the matter with you before the nex

Original' ' David 7.
David W. Manager: Guidanc

